# IEEE Guide for Design of Substation Rigid-Bus Structures 

Sponsor

Substations Committee
of the
IEEE Power Engineering Society

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#### Abstract

Rigid-bus structures for outdoor and indoor, air-insulated, and alternating-current substations are covered. Portions of this guide are also applicable to strain-bus structures or direct-current substations, or both. Ampacity, radio influence, vibration, and forces due to gravity, wind, fault current, and thermal expansion are considered. Design criteria for conductor and insulator strength calculations are included.


Keywords: ampacity, bus support, mounting structure, rigid-bus structures, strain-bus structure

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#### Abstract

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## Introduction

(This introduction is not part of IEEE Std 605-1998, IEEE Guide for Design of Substation Rigid-Bus Structures.)
Substation rigid-bus structures are being applied by electrical utilities and other industries. Such structures usually reduce substation heights and have greater ampacity and lower corona than single-conductor strainbus structures.

This guide presents an integrated design approach with methods for calculating the forces to which rigid-bus structures are subjected.

The data in this guide are taken from empirical and theoretical sources that can form the basis of a good design by a knowledgeable design engineer. The guide is not intended as a rigid procedural design guide for the inexperienced. Some of the empirical methods presented in this guide are based on experience.

Future work is required to give the design engineer a better understanding of certain portions of the design process, such as the flexibility of support structures and insulator overload factors.

The Working Group D3 responsible for the preparation of this guide had the following membership at the time this guide was submitted for approval:

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# IEEE Guide for Design of Substation Rigid-Bus Structures 

## 1. Overview

### 1.1 Scope

The information in this guide is applicable to rigid-bus structures for outdoor and indoor, air-insulated, and alternating-current substations. Portions of this guide are also applicable to strain-bus structures or directcurrent substations, or both. Ampacity, radio influence, vibration, and forces due to gravity, wind, fault current, and thermal expansion are considered. Design criteria for conductor and insulator strength calculations are included. This guide does not consider
a) The electrical criteria for the selection of insulators
b) The seismic forces to which the substation may be subjected
c) The design of mounting structures

### 1.2 Purpose

Substation rigid-bus structure design involves electrical, mechanical, and structural considerations. It is the purpose of this guide to integrate these considerations into one document.

Special consideration is given to fault current-force calculations. Factors considered include the decrement of the fault current, the flexibility of supports, and the natural frequency of the bus. These factors are mentioned in ANSI C37.32-1996, but are not taken into consideration in the equations presented in that standard.

## 2. References

This guide shall be used in conjunction with the following publications. If the following publications are superseded by an approved revision, the revision shall apply.

ANSI C29.1-1988 (R1996), American National Standard Test Methods for Electrical Power Insulators. ${ }^{1}$

[^1]ANSI C29.9-1983 (R1996), American National Standard for Wet-Process Porcelain Insulators (Apparatus, Post Type).

ANSI C37.32-1996, American National Standard for High-Voltage Air Disconnect Switches Interrupter Switches, Fault Initiating Switches, Grounding Switches, Bus supports and Accessories Control Voltage Ranges - Schedule of Preferred Ratings, Construction Guidelines and Specifications.

ASCE 7-95, Minimum Design Loads for Buildings and Other Structures. ${ }^{2}$
ASTM B188-96, Standard Specification for Seamless Copper Bus Pipe and Tube. ${ }^{3}$
ASTM B241/B241M-96, Standard Specification for Aluminum and Aluminum-Alloy Seamless Pipe and Seamless Extruded Tube.

IEEE Std C2-1997, National Electrical Safety Code. ${ }^{4}$
IEEE Std C37.30-1997, IEEE Standard Requirements for High-Voltage Air Switches.
IEEE Std 100-1996, IEEE Standard Dictionary of Electrical and Electronics Terms.
IEEE Std 693-1997, IEEE Recommended Practice for Seismic Design of Substations.
NEMA CC 1-1993, Electric Power Connectors for Substations. ${ }^{5}$

NEMA 107-1988 (R1993), Methods of Measurement of Radio-Influence Voltage (RIV) of High-Voltage Apparatus.

NFPA 70-1996, National Electrical Code. ${ }^{6}$

## 3. Definitions

The following definitions apply specifically to the subject matter of this guide:
3.1 bus structure: An assembly of bus conductors, with associated connection joints and insulating supports.
3.2 bus support: An insulating support for a bus.

NOTE-A bus support includes one or more insulator units with fittings for fastening to the mounting structure and for receiving the bus.
3.3 mounting structure: A structure for mounting an insulating support.
3.4 rigid-bus structure: A bus structure comprised of rigid conductors supported by rigid insulators.
3.5 strain-bus structure: A bus structure comprised of flexible conductors supported by strain insulators.

[^2]
## 4. The design problem

The design problem considered in this guide is the selection of rigid-bus structure components and their arrangements. For a safe, reliable, and economic design, the components and their arrangements should be optimized to satisfy the design conditions.

The design conditions will establish minimum electrical and structural performance. These conditions are dependent upon the characteristics of the power system involved and the location of the substation. The design conditions specify the following:
a) Ampacity requirements
b) Maximum anticipated fault current
c) Maximum operating voltage
d) Maximum anticipated wind speeds
e) Maximum expected icing conditions combined with wind
f) Altitude of the substation site
g) Basic substation layout

The selection of conditions acting simultaneously on the bus structure (that is, fault current, extreme wind, combined wind and ice, or a combination of these) involves probability, and some risk is involved in their selection. The design engineer should consider the risks to life, property, and system operation when the design conditions are selected.

Design conditions should also be specified for the electrical performance of insulators. If the substation is located in an area of possible seismic activity, additional design conditions should be established. IEEE Std 693-1997 and the seismic zone maps in ASCE 7-95 may be used to establish these seismic design conditions.

The actual design can begin after the design conditions are firmly established. Because of the various busstructure components available to the designer and their various possible physical arrangement, the design becomes an iterative process. This iterative process is interrelated by conductor ampacity, suppression of radio influence, elimination of conductor vibrations, and structural integrity (see Figure 1).

It should be noted that a guide is presently being developed by the ASCE that will address the structural aspects of rigid-bus design. When approved, this ASCE guide may be used to verify the structural aspects of this guide.


NOTE-This diagram assumes that maximum span length is not limited by aeolian vibration.
Figure 1-Design process for horizontal rigid bus

## 5. Ampacity

The ampacity requirement of the bus conductor is usually determined by either the electrical system requirements or the ampacity of the connected equipment. Conductor ampacity is limited by the conductor's maximum operating temperature. Excessive conductor temperatures may anneal the conductor, thereby reducing its strength, or may damage connected equipment by the transfer of heat. Excessive temperatures may also cause rapid oxidation of the copper conductor.

### 5.1 Heat balance

The temperature of a conductor depends upon the balance of heat input and output. For balance, the heat input due to $I^{2} R$ and solar radiation equals the heat output due to convection, radiation, and conduction. The heat balance may be expressed as

$$
\begin{equation*}
I^{2} R F+q_{\mathrm{s}}=q_{\mathrm{c}}+q_{\mathrm{r}}+q_{\mathrm{cond}} \tag{1}
\end{equation*}
$$

Solving the current for a given conductor temperature rise is

$$
\begin{equation*}
I=\sqrt{\frac{q_{\mathrm{c}}+q_{\mathrm{r}}+q_{\mathrm{cond}}-q_{\mathrm{s}}}{R F}} \tag{2}
\end{equation*}
$$

where
$I=$ current for the allowable temperature rise, $A$
$R=$ direct-current resistance at the operating temperature, $\Omega / \mathrm{m}[\Omega / \mathrm{ft}]$
$F=$ skin-effect coefficient
$q_{\mathrm{s}}=$ solar heat gain, W/m [W/ft]
$q_{\mathrm{c}}=$ convective heat loss, W/m [W/ft*]
$q_{\mathrm{r}}=$ radiation heat loss, W/m [W/ft]
$q_{\text {cond }}=$ conductive heat loss, W/m [W/ft*]

* Values for convective or conductive heat gains on the right side of Equation (2) are entered as negative numbers.


### 5.1.1 Effective resistance, RF

A conductor's effective resistance at a given temperature and frequency is the direct-current resistance $R$ modified by the skin-effect coefficient $F$. These values may be obtained from published data.

### 5.1.2 Solar heat gain, $q_{s}$

The amount of solar heat gained is a function of
a) The total solar and sky radiation
b) The coefficient of solar absorption for the conductor's surface
c) The projected area of the conductor
d) The altitude of the conductor above sea level
e) The orientation of the conductor with respect to the sun's rays

### 5.1.3 Convective heat loss, $q_{c}$

A bus conductor loses heat through natural or forced convection.

### 5.1.3.1 Natural convective heat loss

Natural convective heat loss is a function of
a) The temperature difference between the conductor surface and the ambient air temperature
b) The orientation of the conductor's surface
c) The width of the conductor's surface
d) The conductor's surface area

### 5.1.3.2 Forced convective heat loss

Forced convective heat loss is a function of
a) The temperature difference between the conductor's surface and the ambient air temperature
b) The length of flow path over the conductor
c) The wind speed
d) The conductor's surface area

### 5.1.4 Radiation heat loss, $q_{r}$

A conductor loses heat through the emission of radiated heat. The heat lost is a function of
a) The difference in the absolute temperature of the conductor and surrounding bodies
b) The emissivity of the conductor's surface
c) The conductor's surface area

### 5.1.5 Conductive heat loss, $q_{\text {cond }}$

Conduction is a minor method of heat transfer since the contact surface is usually very small. Conduction may cause an increase in the temperature of the equipment attached to the bus conductor. Conductive heat loss is usually neglected in bus-ampacity calculations.

### 5.2 Conductor temperature limits

### 5.2.1 Continuous (see IEEE Std C37.30-1997)

Aluminum alloy and copper conductors may be operated continuously at $90^{\circ} \mathrm{C}$ without appreciable loss of strength. They may also be operated at $100{ }^{\circ} \mathrm{C}$ under emergency conditions with some annealing. Copper may, however, suffer excessive oxidation if operated at or above $80^{\circ} \mathrm{C}$. Conductors should not be operated at temperatures high enough to damage the connected equipment.

### 5.2.2 Fault conditions (see ANSI C37.32-1996)

A conductor's temperature will rise rapidly under fault conditions. This is due to the inability of the conductor to dissipate the heat as rapidly as it is generated. Annealing of conductor alloys may occur rapidly at these elevated temperatures. The maximum fault current that can be allowed for copper and aluminum alloy conductors may be calculated using Equations (3) and (4). In general, the final temperature of the conductor is limited to the maximum temperature considered for thermal expansion (see Clause 13).

For aluminum conductors [ $40 \%$ to $65 \%$ International Annealed Copper Standard (IACS) conductivity],

$$
\begin{equation*}
I=C \times 10^{6} A \sqrt{\frac{1}{t} \log _{10} \frac{T_{\mathrm{f}}-20+(15150 / G)}{T_{\mathrm{i}}-20+(15150 / G)}} \tag{3}
\end{equation*}
$$

where

```
\(C=92.9\) for Metric units [. 144 for English units]
\(I=\) maximum allowable root-mean-square (rms) value of fault current, \(A\)
\(A=\) conductor cross-sectional area, \(\mathrm{mm}^{2}\left[\mathrm{in}^{2}\right]\)
\(G=\) conductivity in percent International Annealed Copper Standard (IACS)
\(t=\) duration of fault, s
\(T_{\mathrm{f}}=\) allowable final conductor temperature, \({ }^{\circ} \mathrm{C}\)
\(T_{\mathrm{i}}=\) conductor temperature at fault initiation, \({ }^{\circ} \mathrm{C}\)
```

And for copper conductors [ $95 \%$ to $100 \%$ International Annealed Copper Standard (IACS) conductivity],

$$
\begin{equation*}
I=C \times 10^{6} A \sqrt{\frac{1}{t} \log _{10} \frac{T_{\mathrm{f}}-20+(25400 / G)}{T_{\mathrm{i}}-20+(25400 / G)}} \tag{4}
\end{equation*}
$$

$C=142$ for Metric units [0.22 for English units]
All other variables have been defined previously.

### 5.2.3 Attached equipment

Since heat generated in the bus conductor may be conducted to attached equipment, allowable conductor temperatures may be governed by the temperature limitations of attached equipment. Equipment temperature limitations should be obtained from the applicable specification or the manufacturer. High-voltage air switches and bus supports are described in IEEE Std C37.30-1997.

### 5.3 Ampacity tables

The ampacities for most aluminum-alloy and copper bus-conductor shapes are included in Annex B. These ampacities were calculated using the methods outlined in Annex C, which neglect conductive heat loss.

## 6. Corona and radio influence

Corona develops when the voltage gradient at the surface of a conductor exceeds the dielectric strength of the air surrounding the conductor and ionizes the air molecules. Radio influence (RI) is caused by corona. In practice, corona has not been a factor in rigid-bus design at 115 kV and below. However, the rigid-bus designer should be aware that radio influence can be produced at any voltage by arcing due to poor bonding between bus conductors and associated hardware.

The proximity and largeness of the equipment within a substation create multiple low-impedance paths to ground for radio-frequency current. The Radio Noise Subcommittee of the IEEE Transmission and Distribution Committee states that actual radio influence will be less than that calculated because of this effect [B15]. ${ }^{7}$

The designer's problem is to select a bus conductor and specify bus hardware that is corona free during fairweather conditions at the operating voltage, altitude, and temperature. It should be noted that corona may exist under wet or contaminated conditions.

[^3]
### 6.1 Conductor selection

For corona-free operation, the maximum surface voltage gradient of the bus-conductor $\mathrm{E}_{\mathrm{m}}$ should be less than the allowable surface voltage gradient $\mathrm{E}_{\mathrm{o}}$.

Four basic factors determine the maximum surface voltage gradient of a smooth bus-conductor $\mathrm{E}_{\mathrm{m}}$. They are

1) Conductor diameter or shape
2) Distance from ground
3) Phase spacing
4) Applied voltage

Circular bus shapes will generally give the best performance. A smooth surface condition is important if operating near the allowable surface voltage gradient.

Formulae are provided in Annex D for calculating the maximum surface voltage gradient for a smooth, circular bus-conductor $\mathrm{E}_{\mathrm{m}}$. The calculation should be $110 \%$ of the nominal line-to-ground voltage to provide for an operating margin.

The allowable surface voltage gradient for equal radio-influence generation $\mathrm{E}_{\mathrm{O}}$ for smooth, circular bus conductors is a function of bus diameter, barometric pressure, and operating temperature. Annex D gives a method for determining the allowable surface voltage gradient.

### 6.2 Hardware specifications

Bus fittings and hardware for use in rigid-bus structures should be specified as being free of corona under fair-weather conditions at the intended operating voltage, altitude, and temperature.

It should be noted that the testing methods referred to in 6.2.1 do not require the control of air temperature and air pressure during testing. The specifier should refer to Annex D to determine the difference between the allowable voltage gradients under expected operating conditions and possible laboratory conditions. If the difference is significant, the designer may specify that the testing voltage be increased according to the methods of Annex D to compensate for the test pressure and temperature.

### 6.2.1 Testing methods

Bus fittings and hardware should be tested by the manufacturer in a laboratory under simulated field configuration. All bus fittings and hardware should be tested while attached to a section of the bus conductor for which they are to be used.

### 6.2.1.1 Visual corona

The visual corona extinction voltage should be tested according to NEMA CC 1-1993.

### 6.2.1.2 Radio-influence voltage (RIV) level

The radio-influence voltage (RIV) level should be tested according to NEMA 107-1988.

### 6.2.2 Acceptance criteria

The following performance should be specified for fittings and hardware under fair-weather conditions.

### 6.2.2.1 Visual corona

The extinction voltage for visual corona should be at least $110 \%$ of nominal operating voltage or at least $110 \%$ of the testing voltage adjusted to compensate for pressure and temperature.

### 6.2.2.2 Radio-influence voltage (RIV)

The specified radio-influence voltage (RIV) limits for various bus system components should match those given in the following standards:
a) For fittings and connectors, see NEMA CC 1-1993.
b) For insulators and hardware assemblies, see ANSI C29.9-1983.

## 7. Conductor vibration

A span of rigid conductor has its own natural frequency of vibration. If the conductor is displaced from its equilibrium position and released, it will begin to vibrate at this natural frequency. The magnitude of the oscillations will decay due to damping. If, however, the conductor is subjected to a periodic force whose frequency is near the natural frequency of the span, the bus may continue to vibrate and the amplitude will increase.

This vibration may cause damage to the bus conductor by fatigue or by excessive fiber stress.

### 7.1 Natural frequency

The natural frequency of a conductor span is dependent upon the manner in which the ends are supported and upon the conductor's length, mass, and stiffness. The natural frequency of a conductor span can be calculated using Equation (5).

$$
\begin{equation*}
f_{\mathrm{b}}=\frac{\pi K^{2}}{C L^{2}} \sqrt{\frac{E J}{m}} \tag{5}
\end{equation*}
$$

where

```
\(C=20\) for Metric units [24 for English units]
\(f_{\mathrm{b}}=\) natural frequency of conductor span, Hz
\(L=\) span length, \(\mathrm{m}[\mathrm{ft}]\)
\(E=\) modulus of elasticity, \(\mathrm{MPa}\left[\mathrm{lbf} / \mathrm{in}^{2}\right]\)
\(J=\) moment of inertia of cross-sectional area, \(\mathrm{cm}^{4}\left[\mathrm{in}^{4}\right]\)
\(m=\) mass per unit length, \(\mathrm{kg} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]\)
\(K=1.00\) for two pinned ends (dimensionless)
\(K=1.22\) for one pinned end and one fixed end (dimensionless)
\(K=1.51\) for two fixed ends (dimensionless)
```

End conditions can range between fixed and pinned. A fixed end is not free to rotate (moment resisting), whereas a pinned end is free to rotate (not moment resisting). Because of structure flexibility and connection friction, the end conditions are not truly fixed or pinned. However, the end conditions are generally closer to fixed than to pinned.

### 7.2 Driving functions

Either alternating current or wind may induce vibrations in a bus conductor with frequencies near the natural frequency of the bus conductor.

### 7.2.1 Current-induced vibrations

Currents flowing through parallel conductors create magnetic fields that interact and exert forces on the parallel conductors. This driving force oscillates at twice the power frequency.

If the calculated natural frequency of a bus span is found to be greater than half the current-force frequency (that is, greater than the power frequency), the bus spans' calculated natural frequency should be changed or a dynamic analysis should be made to determine stresses involved.

### 7.2.2 Wind-induced vibration

When a laminar (constant, nonturbulent) wind flows across a conductor, aeolian vibration may occur. This vibration may cause bus-conductor fatigue. Laminar flow does not usually occur at high wind speeds because of the ground effects created by terrain, trees, buildings, local thermal conditions, etc. Experience has shown that wind with speeds up to $15 \mathrm{mi} / \mathrm{h}$ can have laminar flow.

The maximum frequency of the aeolian force for circular conductors may be calculated from Equation (6), which is based on the Strovhal formula.

$$
\begin{equation*}
f_{\mathrm{a}}=\frac{C V}{d} \tag{6}
\end{equation*}
$$

where
$C=5.15$ for Metric units [3.26 for English units]
$f_{\mathrm{a}}=$ maximum aeolian force frequency, Hz
$V=$ maximum wind speed for laminar flow, $\mathrm{km} / \mathrm{h}$ [mi/h]
$d=$ conductor diameter in, $[\mathrm{cm}]$

Formulae for calculating aeolian force frequency for bus cross-sectional shapes other than circular are not available.

If twice the calculated natural frequency of the bus span is greater than the aeolian force frequency, then the bus span length should be changed or the bus should be damped.

### 7.3 Damping

Bus spans may be damped to reduce aeolian vibration. For tubular bus conductors, damping may be accomplished by installing stranded bare cable inside the bus conductor to dissipate vibrational energy. The cable should be of the same material as the bus conductor to prevent corrosion, and the weight of the cable should be from $10 \%$ to $33 \%$ of the bus-conductor weight, although some designers have found that from $3 \%$ to $5 \%$ of the bus-conductor weight is adequate. In some locations, the audible noise generated by stranded cable dampers may be unacceptable.

Commercially available vibration dampers may be used for both tubular and nontubular conductors. Commercial vibration dampers should be sized and placed according to the manufacturer's recommendations.

## 8. Conductor gravitational forces

Gravitational forces determine the vertical deflection of bus conductors and are a component of the total force, which the conductor must withstand. Gravitational forces consist of the weights of the conductor, damping material, ice, and concentrated masses.

### 8.1 Conductor

Conductor weight should be obtained from applicable specifications or from the manufacturer.

### 8.2 Damping material

The weight of the material used to damp vibration should be included in computing gravitational forces. If commercial dampers are used, these should be considered as concentrated masses.

### 8.3 Ice

The minimum radial ice thickness used for design should be determined from IEEE Std C2-1997. See Figure 2 or [B3].


Figure 2-General loading map showing territorial division of the United States with respect to loading of overhead lines

Consideration should be given to special local conditions where greater ice thicknesses may occur, such as near a cooling-tower installation. The ice weight on a circular conductor is given as

$$
\begin{equation*}
F_{\mathrm{I}}=C \pi W_{\mathrm{I}} r_{\mathrm{I}}\left(d+r_{\mathrm{I}}\right) \tag{7}
\end{equation*}
$$

where
$C=0.0001$ for Metric units [12 for English units]
$F_{\text {I }}=$ ice unit weight, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]$
$W_{\mathrm{I}}=$ ice weight $=7.18, \mathrm{~N} / \mathrm{m}^{3}\left[0.0330, \mathrm{lbf} / \mathrm{in}^{3}\right]$
$r_{\mathrm{I}}=$ radial ice thickness, cm [in]
$d=$ outside conductor diameter, cm [in]

Equation (7) may be simplified to

$$
\begin{equation*}
F_{\mathrm{I}}=C r_{\mathrm{I}}\left(d+r_{\mathrm{I}}\right) \tag{8}
\end{equation*}
$$

where

$$
C=2.26 \times 10^{-3} \text { for Metric units [1.24 for English units] }
$$

Similar equations may be derived for other conductor shapes.

### 8.4 Concentrated masses

Gravitational forces due to concentrated masses (vibration dampers, equipment attachments, cross conductors, etc.) should be determined and included in the summation of gravitational forces.

## 9. Conductor wind forces

The bus structure should be capable of withstanding the mechanical forces due to expected winds. The maximum force due to wind may occur either during extreme wind conditions with no ice or high wind conditions with ice. In general, the maximum wind speed with ice is less than the extreme wind speed.

The annual extreme fastest-mile wind speeds for design without ice should be determined from ASCE 7-95. See Figure 3 or [B3]. The choice of the 50- or 100-year recurrence map depends upon the degree of hazard to life or property. Local or state codes should be followed if their wind-force requirements exceed those determined by reference to ASCE 7-95.

The fastest-mile wind speed with ice should be determined from the ice/wind history at the substation site. In general, the wind speed that occurs after icing conditions is lower than the annual extreme fastest-mile wind speed.


Table 1 displays the drag coefficients for structural shapes in relation to profile and wind direction.

Table 1-Drag coefficients for structural shapes


Factors that will affect wind forces are the speed and gust of the wind, radial ice thickness, and the shape, diameter, height, and exposure of the conductors.

The unit wind force for bus is given as

$$
\begin{equation*}
F_{\mathrm{W}}=C C_{\mathrm{D}} K_{\mathrm{Z}} G_{\mathrm{F}} V^{2} I\left(d+2 r_{\mathrm{I}}\right) \tag{9}
\end{equation*}
$$

where

$$
\begin{aligned}
& C=6.13 \times 10^{-3} \text { for Metric unit }\left[2.132 \times 10^{-4} \text { for English units }\right] \\
& F_{\mathrm{W}}=\text { wind unit force on bus, } \mathrm{N} / \mathrm{mt}[\mathrm{lbf} / \mathrm{f}] \\
& d=\text { outside conductor diameter, } \mathrm{cm}[\mathrm{in}] \\
& r_{\mathrm{I}}=\text { radial ice thickness, } \mathrm{cm}[\mathrm{in}]
\end{aligned}
$$

```
\(C_{\mathrm{D}}=\) drag coefficient, (see 9.1)
\(K_{\mathrm{Z}}=\) height and exposure factor, (see 9.2)
\(G_{\mathrm{F}}=\) gust factor, (see 9.3)
\(V=\) wind speed at \(9.1 \mathrm{~m}(30 \mathrm{ft})\) above ground, \(\mathrm{km} / \mathrm{h}[\mathrm{mi} / \mathrm{h}]\)
\(I=\) importance factor (see 9.4)
```


### 9.1 Drag coefficient, $C_{D}$

The wind force exerted on a conductor varies with the shape of the conductor. This variation is reflected in the drag coefficient $C_{\mathrm{D}}$. The drag coefficient for smooth tubular conductors is 1.0 . Coefficients for other shapes are given in Table 1.

### 9.2 Height and exposure factor, $\mathrm{K}_{\mathrm{Z}}$

In the height zone from $0 \mathrm{~m}(0 \mathrm{ft})$ to $9.1 \mathrm{~m}(30 \mathrm{ft})$ and for exposure category $\mathrm{A}, \mathrm{B}, \mathrm{C}$, the height and exposure factor $K_{\mathrm{z}}=1.0$, and the wind speed at $9.1 \mathrm{~m}(30 \mathrm{ft})$ should be used. For exposure category, $K_{\mathrm{z}}=1.16$. ASCE 7-95 has a detailed definition of each of these exposure categories. Summarized definitions are as follows:

- Exposure A: Large city centers with at least $50 \%$ of the buildings having a height in excess of 21.3 m (70 ft).
- Exposure B: Urban and suburban areas, wooded areas, or other terrain with numerous closely spaced obstructions having the size of single-family dwellings or larger.
- Exposure C: Open terrain with scattered obstructions having heights generally less than $9.1 \mathrm{~m}(30 \mathrm{ft})$. This category includes flat open country and grassland.
- Exposure D: Flat, unobstructed areas exposed to wind flowing over open water for a distance of at least $1.61 \mathrm{~km}(1 \mathrm{mi})$.


### 9.3 Gust factors, $\mathbf{G}_{\mathrm{F}}$

A gust factor $G_{\mathrm{F}}$ of 0.8 shall be used for exposure A and B , and 0.85 shall be used for exposure C and D .

### 9.4 Importance factor, I

The importance factor $I$ for electric substations shall be 1.15 as classified by ASCE 7-95.

## 10. Conductor fault current forces

The magnetic fields produced by fault current cause forces on the bus conductors. The bus conductors and bus supports must be strong enough to withstand these forces.

The force imparted to the bus structure by fault current is dependent on conductor spacing, magnitude of fault current, type of short circuit, and degree of short-circuit asymmetry. Other factors to be considered are support flexibility, and corner and end effects.

### 10.1 Classical equation

The classical equation for the force between parallel, infinitely long conductors in a flat configuration due to an asymmetrical short-circuit current is as follows:

For Metric units:

$$
\begin{equation*}
F_{\mathrm{SC}}=\frac{5.4 \Gamma\left(2 \sqrt{2} I_{\mathrm{SC}}\right)^{2}}{10^{7}(D)}=\frac{43.2 \Gamma I_{\mathrm{SC}}^{2}}{10^{7}(D)} \tag{10a}
\end{equation*}
$$

For English units:

$$
\begin{equation*}
F_{\mathrm{SC}}=\frac{2 \Gamma\left(2 \sqrt{2} I_{\mathrm{SC}}\right)^{2}}{10^{4}(D)}=\frac{1.6 \Gamma I_{\mathrm{SC}}^{2}}{10^{4}(D)} \tag{10b}
\end{equation*}
$$

where

```
\(F_{\mathrm{sc}}=\) Fault current unit force, \(\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]\)
\(I_{\mathrm{sc}}=\) symmetrical rms fault current, \(A\)
\(D=\) conductor spacing center-to-center, cm [in]
\(\Gamma=\) constant based on type of fault and conductor location (see Table 2)
```

Equation (10) assumes that the fault is initiated to produce the maximum current offset. The magnitudes of the fault current $I_{S C}$ for each type of fault (three-phase, phase-to-phase, etc.) are not equal to each other and will depend upon the electrical system parameters.

Unless data on the present and future available fault currents are known, it is suggested that the interrupting capability of the substation interrupting equipment (circuit breakers, circuit switchers, etc.) be considered as the maximum $I_{\mathrm{SC}}$.

Table 2-Constant $\Gamma$ for calculating short-circuit current forces

| Type of short circuit | Configuration | Force on conductor | $\Gamma$ |
| :---: | :---: | :---: | :---: |
| Phase-to-Phase |  | A or B | 1.00 |
| Three-Phase |  | B | 0.866 |
| Three-Phase |  | A or C | 0.808 |

### 10.2 Decrement factor

Due to the presence of system impedance, there is a decrement of the asymmetrical wave in the first halfcycle of the fault. Therefore, it is practical to assume a lower value of peak fault current. Using a value of 1.6 as the assumed current offset, Equation (10) becomes

$$
\begin{equation*}
F_{\mathrm{SC}}=\frac{C \Gamma\left(D_{f} \sqrt{2 I_{\mathrm{SC}}}\right)^{2}}{(D)} \tag{11}
\end{equation*}
$$

where
$C=0.2 \times 1010^{-4}$ for Metric units [5.4 $\times 10^{-7}$ for English units]
$F_{\mathrm{sc}}=$ short-circuit current unit force, $\mathrm{N} / \mathrm{mt}[\mathrm{lbf} / \mathrm{f}]$
$I_{\mathrm{sc}}=$ symmetrical short-circuit current, $A, \mathrm{rms}$
$D=$ conductor spacing center-to-center, cm [in]
$\Gamma=$ constant based on type of short circuit and conductor location (see Table 2)
$D_{\mathrm{f}}=$ decrement factor is given by the following equation:

$$
\begin{equation*}
D_{\mathrm{f}}=\sqrt{1+\frac{T_{\mathrm{a}}}{t_{\mathrm{f}}}\left(1-\exp ^{-\frac{2 t_{\mathrm{f}}}{T_{\mathrm{a}}}}\right)} \tag{11a}
\end{equation*}
$$

where

$$
\begin{aligned}
& T_{\mathrm{a}}=\frac{X}{R} \frac{1}{2 \pi f}(X=\text { System Reactance, } R=\text { System Resistance, and } f=60 \mathrm{~Hz}) \\
& t_{\mathrm{f}}=\text { fault current duration in seconds }
\end{aligned}
$$

If a system's maximum current offset is less than the assumed value of 1.6 , the force will be further reduced.

Equation (11) gives the maximum force in the first half-cycle of the fault. The actual force present when maximum conductor span deflection occurs is usually less because
a) Most conductor spans will not reach maximum deflection until after the first quarter-cycle, and
b) Additional current decrement occurs as the fault continues.

The combination of these two factors results in lower maximum deflection than the deflection caused by a steady-state force equal to the maximum force in the first quarter-cycle.

Tests have shown that conductor spans with natural frequencies of $1 / 10$ of the power frequency or less, and in a system with an $\mathrm{X} / \mathrm{R}$ ratio of 13 or less, will have fault current forces of less than one half the calculated first quarter-cycle force when the conductor span reaches full deflection.

In practice, a static force equal to the first quarter-cycle force is generally used to calculate rigid-bus structure deflections and stresses. This practice has given a margin of safety to the rigid-bus structure design for fault current forces.

### 10.3 Mounting-structure flexibility

Because of their flexibility, the bus and mounting structures are capable of absorbing energy during a fault. Thus, depending on the type of mounting structures and their heights, the effective fault current forces can be further reduced by using Equation (12).

$$
\begin{equation*}
F_{\mathrm{SC}}=K_{\mathrm{f}} \frac{C \Gamma\left(D_{\mathrm{f}} \sqrt{2 I_{S C}}\right)^{2}}{(D)} \tag{12}
\end{equation*}
$$

$K_{\mathrm{f}}=$ mounting-structure flexibility factor

Values of $K_{\mathrm{f}}$, as suggested by Working Group D3 for single-phase mounting structures, are given in Figure 4. $K_{\mathrm{f}}$ is usually assumed to be unity for three-phase mounting structures.

All other variables have been defined previously.

There have been fault current tests conducted on specific combinations of rigid-bus structures with mounting structures that indicate lower values of $K_{\mathrm{f}}$ than those shown in Figure 4. Where the structures are similar to those tested, the lower values of $K_{\mathrm{f}}$ may apply. Future work is expected to produce methods for determining values of $K_{\mathrm{f}}$ for specific mounting structures.


$$
\begin{aligned}
& A=\text { Lattice and tubular aluminum } \\
& B=\text { Tubular and wide-flange steel, and wood pole } \\
& C=\text { Lattice steel } \\
& D=\text { Solid concrete }
\end{aligned}
$$

Figure $4-K_{f}$ for various types of single-phase mounting structures

### 10.4 Corner and end effects

The values for the short-circuit current force calculated by Equations (10), (11), and (12) are for parallel and infinitely long conductors. The results for short bus lengths will be conservative because of end effects. The equations cannot be used for special cases, such as corners and nonparallel conductors. Annex E provides methods for determining the forces for special bus configurations.

## 11. Conductor strength considerations

Any span of a bus conductor must have enough stiffness and strength to withstand the expected forces of gravity, wind, and short circuits, and maintain its mechanical and electrical integrity. The span should also not sag excessively under normal conditions.

This clause only includes equations for single-level, single-span bus conductors supported at both ends, and for continuous bus conductors supported at equal spans without concentrated loads. Annex F of this guide covers analysis for other forms. The simple static method given in Annex F can be used for analyzing distributed loads and concentrated loads on continuous bus conductors supported at equal or unequal spans.

This method is particularly valuable for analyzing a two-level bus arrangement, where one bus at the lower level supports the other bus at upper level using an A-frame form. In this case, the forces acting on the upper bus are transmitted to the lower bus as concentrated loads at each base of the A-frame. Such loading can impose severe stress on the bus conductors and the supporting insulators. A full static analysis could be performed to determine the stresses at various points using the method described in Annex F or other methods obtained from structural design handbooks.

### 11.1 Vertical deflection

### 11.1.1 Vertical deflection limits

The allowable vertical deflection of a bus conductor is usually limited by appearance. Commonly used limits are based either on the ratio of conductor deflection to span length ( $1: 300$ to $1: 150$ ), or the vertical dimension of the conductor ( $0.5 \%$ to $1 \%$ times the vertical dimension). Vertical deflection depends upon the total gravitational force. In practice, since appearance is usually not considered during icing conditions, the ice weight is usually not considered for vertical deflection. However, if the vertical deflection during icing conditions is important, then ice weight should be considered.

### 11.1.2 Total gravitational force

The total gravitational force on a conductor is the sum of the weights of the conductor, ice, damping material, and any concentrated loads. Without concentrated loads,

$$
\begin{equation*}
F_{\mathrm{G}}=F_{\mathrm{c}}+F_{\mathrm{I}}+F_{\mathrm{D}} \tag{13}
\end{equation*}
$$

where

$$
\begin{aligned}
& F_{\mathrm{G}}=\text { total bus unit weight, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}] \\
& F_{\mathrm{c}}=\text { conductor unit weight, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}] \\
& F_{\mathrm{I}}=\text { ice unit weight, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}] \\
& F_{\mathrm{D}}=\text { clamping material unit weight, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]
\end{aligned}
$$

If the bus span is subjected to concentrated loads, the force distribution on that span should be analyzed more thoroughly.

### 11.1.3 Allowable span length for vertical deflection

The maximum allowable bus span length may be calculated with a given vertical deflection limit, end conditions, and total vertical force distribution.

The deflection may be based on either the vertical conductor dimension or a fraction of the span length.

End conditions for a single span range between fixed and pinned - A fixed end is not free to rotate (moment resisting), whereas a pinned end is free to rotate (not moment resisting). In reality, because of supporting structure flexibility and connection friction, the end conditions are not truly fixed or pinned.

If the end conditions of the single bus span are unknown, then Equation (14) for two pinned ends should be used. For a continuous bus, end conditions are assumed to be pinned and mid-supports are fixed.

### 11.1.3.1 Single-span bus, two pinned ends

For a single span with two pinned ends, the allowable span length based on vertical deflection may be calculated by one of the equations given in Table 3:

Table 3-Modulus of elasticity E for common conductor alloys

| Bus-Conductor Alloy | $E$ |  |
| :--- | :---: | :---: |
|  | $\mathbf{k P a}$ | $\mathbf{l b f} / \mathbf{i n}^{2}$ |
| Aluminum 6061-T6 | $6.895 \times 10^{7}$ | $10 \times 10^{6}$ |
| Aluminum 6063-T6 | $6.895 \times 10^{7}$ | $10 \times 10^{6}$ |
| Aluminum 6101-T61 | $6.895 \times 10^{7}$ | $10 \times 10^{6}$ |
| Copper | $11.03 \times 10^{7}$ | $16 \times 10^{6}$ |

$$
\begin{equation*}
L_{\mathrm{D}}=C \sqrt[4]{\frac{384(E)(J)\left(Y_{\mathrm{A}}\right)}{5 F_{\mathrm{G}}}} \text { or } L_{\mathrm{D}}=C\left[\frac{384(E)(J)\left(Y_{\mathrm{A}}\right)}{5 F_{\mathrm{G}}}\right]^{\frac{1}{4}} \tag{14}
\end{equation*}
$$

where

$$
\begin{aligned}
& C=1.78 \text { for Metric units }[1.86 \text { for English units }] \\
& L_{\mathrm{D}}=\text { allowable span length, } \mathrm{cm}[\mathrm{in}] \\
& Y_{\mathrm{A}}=\text { allowable deflection, } \mathrm{cm}[\mathrm{in}] \\
& E=\text { modulus of elasticity, } \mathrm{kPa}\left[\mathrm{lbf} / \mathrm{in}^{2}\right](\text { see Table } 3) \\
& \left.J=\text { cross-sectional moment of inertia, } \mathrm{cm}^{4}\left[\mathrm{in}^{4}\right] \text { (see }[\mathrm{B} 1], \text { chapter } 13\right) \\
& F_{\mathrm{G}}=\text { total bus unit weight, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]
\end{aligned}
$$

or

$$
\begin{equation*}
L_{\mathrm{D}}=C \sqrt[3]{\frac{384(E)(J)\left(Y_{\mathrm{B}}\right)}{5 F_{\mathrm{G}}}} \text { or } L_{\mathrm{D}}=C\left[\frac{384(E)(J)\left(Y_{\mathrm{B}}\right)}{5 F_{\mathrm{G}}}\right]^{\frac{1}{3}} \tag{15}
\end{equation*}
$$

where
$Y_{\mathrm{B}}=$ allowable deflection as a fraction of span length

All other variables have been defined previously.

### 11.1.3.2 Single-span bus, two fixed ends

For a span with two fixed ends, the allowable span length based on vertical deflection may be calculated by Equations (16) or (17).

$$
\begin{equation*}
L_{\mathrm{D}}=C \sqrt[4]{\frac{384(E)(J)\left(Y_{\mathrm{A}}\right)}{F_{\mathrm{G}}}} \text { or } L_{\mathrm{D}}=C\left[\frac{384(E)(J)\left(Y_{\mathrm{A}}\right)}{F_{\mathrm{G}}}\right]^{\frac{1}{4}} \tag{16}
\end{equation*}
$$

or

$$
\begin{equation*}
L_{\mathrm{D}}=C \sqrt[3]{\frac{384(E)(J)\left(Y_{\mathrm{B}}\right)}{F_{\mathrm{G}}}} \text { or } L_{\mathrm{D}}=C\left[\frac{384(E)(J)\left(Y_{\mathrm{B}}\right)}{F_{\mathrm{G}}}\right]^{\frac{1}{3}} \tag{17}
\end{equation*}
$$

where

$$
\begin{aligned}
& C=1.78 \text { for Metric units }[1.86 \text { for English units }] \\
& L_{\mathrm{D}}=\text { allowable span length, } \mathrm{cm}[\mathrm{in}] \\
& Y_{\mathrm{A}}=\text { allowable deflection, } \mathrm{cm}[\mathrm{in}] \\
& Y_{\mathrm{B}}=\text { allowable deflection as a fraction of span length } \\
& \left.E=\text { modulus of elasticity, } \mathrm{kPa}\left[\mathrm{lbf} / \mathrm{in}^{2}\right] \text { (see Table } 3\right) \\
& J=\text { cross-sectional moment of inertia, } \mathrm{cm}^{4}\left[\mathrm{in}^{4}\right] \text { (see [B1], chapter 13) } \\
& F_{\mathrm{G}}=\text { total bus unit weight, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]
\end{aligned}
$$

### 11.1.3.3 Single-span bus, one pinned end, one fixed end

For a single span with one pinned end and one fixed end, Equations (18) and (19) may be used to calculate the maximum allowable span length based on vertical deflection.

$$
\begin{equation*}
L_{\mathrm{D}}=C \sqrt[4]{\frac{185(E)(J)\left(Y_{\mathrm{A}}\right)}{F_{\mathrm{G}}}} \text { or } L_{\mathrm{D}}=C\left[\frac{185(E)(J)\left(Y_{\mathrm{A}}\right)}{F_{\mathrm{G}}}\right]^{\frac{1}{4}} \tag{18}
\end{equation*}
$$

or

$$
\begin{equation*}
L_{\mathrm{D}}=C \sqrt[3]{\frac{185(E)(J)\left(Y_{\mathrm{B}}\right)}{F_{\mathrm{G}}}} \text { or } L_{\mathrm{D}}=C\left[\frac{185(E)(J)\left(Y_{\mathrm{B}}\right)}{F_{\mathrm{G}}}\right]^{\frac{1}{3}} \tag{19}
\end{equation*}
$$

where

$$
C=1.78 \text { for Metric units [1.86 for English units] }
$$

All other variables have been defined previously.

### 11.1.3.4 Continuous bus

For a continuous bus, Equations (20) or (21) may be used to calculate the maximum allowable span length based on vertical deflection.

$$
\begin{align*}
& L_{\mathrm{D}}=C \sqrt[4]{\frac{185(E)(J)\left(Y_{\mathrm{A}}\right)}{F_{\mathrm{G}}}} \text { or } L_{\mathrm{D}}=C\left[\frac{185(E)(J)\left(Y_{\mathrm{A}}\right)}{F_{\mathrm{G}}}\right]^{\frac{1}{4}}  \tag{20}\\
& L_{\mathrm{D}}=C \sqrt[3]{\frac{185(E)(J)\left(Y_{\mathrm{B}}\right)}{F_{\mathrm{G}}}} \text { or } L_{\mathrm{D}}=C\left[\frac{185(E)(J)\left(Y_{\mathrm{B}}\right)}{F_{\mathrm{G}}}\right]^{\frac{1}{3}} \tag{21}
\end{align*}
$$

where

$$
C=1.78 \text { for Metric units [1.86 for English units] }
$$

All other variables have been defined previously.
NOTE-The above equations are for two-span buses. For continuous bus of more than two spans, the maximum deflection occurs in the end spans and is slightly less than that of the two-span bus. The allowable span will be slightly longer.

### 11.2 Conductor fiber stress

In some cases, span lengths may be limited by the fiber stress of the bus-conductor material. The elastic limit and minimum yield stresses for common conductor materials are tabulated in Table 4. In practice, when wind and gravitational forces are combined, the elastic limit stress is commonly used as the maximum allowable stress. When wind, gravitational, and fault current forces $F_{\mathrm{SC}}$ are combined, the minimum yield stress is commonly used as the maximum allowable stress, since $F_{\text {SC }}$ is conservative.

### 11.2.1 Effects of welding

Where welded fittings are used for bus, the allowable stress for the bus should be reduced to allow for annealing due to welding. Tests have shown that the reduction in allowable stress is approximately $50 \%$ for aluminum. The reduction in allowable stress for copper is dependent on the welding method (brazing, exothermic, etc.) and should be discussed with the manufacturers. Locating the weld in a region of moderate stress is a usual method of offsetting the effect of weld annealing. Where welded splices are used with a tubular bus, the reduction in allowable stress may not be required if a reinforcing insert is incorporated.

### 11.2.2 Summation of conductor forces

The maximum bending stresses a conductor withstands are a function of the total vectorial force on the conductor. The total force on a conductor in a horizontal configuration is

$$
\begin{equation*}
F_{\mathrm{T}}=\sqrt{\left[\left(F_{\mathrm{w}}+F_{\mathrm{SC}}\right)^{2}+\left(F_{\mathrm{G}}\right)^{2}\right]} \tag{22}
\end{equation*}
$$

where
$F_{\mathrm{T}}=$ total unit force, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]$
$F_{\mathrm{w}}=$ wind unit force, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]$
$F_{\mathrm{SC}}=$ fault unit force, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]$
$F_{\mathrm{G}}=$ total bus unit weight, $\mathrm{N} / \mathrm{m}[\mathrm{lb} / \mathrm{ft}]$

Table 4-Allowable stress for common conductor materials

| Bus-Conductor Material | Stress (lbf/in²) |  | Stress (kPa) |  |
| :--- | :---: | :---: | :---: | :---: |
|  | Elastic <br> Limit | Minimum <br> Yield | Elastic <br> Limit | Minimum <br> Yield |
| Aluminum alloy-6063-T6 or 6101-T6 | 20500 | $25000^{\mathrm{a}}$ | 141348 | $172375^{\mathrm{a}}$ |
| Aluminum alloy-6061-T6 | 29500 | $35000^{\mathrm{a}}$ | 203403 | $241325^{\mathrm{a}}$ |
| Aluminum alloy-6061-T61 | 11000 | $15000^{\mathrm{a}}$ | 75845 | $103452^{\mathrm{a}}$ |
| Copper No. 110 hard drawn | - | $24000^{\mathrm{b}}$ | - | $275800^{\mathrm{b}}$ |

${ }^{\text {a }}$ With 0.2 offset per ASTM B241/B241M-96.
${ }^{\mathrm{b}}$ With $0.5 \%$ offset per ASTM B188-96.

The angle of the total force below horizontal is

$$
\begin{equation*}
\theta=\tan ^{-1}\left[\frac{F_{\mathrm{G}}}{F_{\mathrm{w}}+F_{\mathrm{SC}}}\right] \tag{23}
\end{equation*}
$$

The total force on a conductor in a vertical configuration is

$$
\begin{equation*}
F_{\mathrm{T}}=\sqrt{\left(F_{\mathrm{w}}\right)^{2}+\left(F_{\mathrm{G}}+F_{\mathrm{SC}}\right)^{2}} \tag{24}
\end{equation*}
$$

where

$$
\begin{aligned}
& F_{\mathrm{T}}=\text { total unit force, } \mathrm{lbf} / \mathrm{ft}[\mathrm{~N} / \mathrm{m}] \\
& F_{\mathrm{W}}=\text { wind unit force, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}] \\
& F_{\mathrm{G}}=\text { total bus unit weight, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}] \\
& F_{\mathrm{SC}}=\text { short-circuit unit force, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]
\end{aligned}
$$

The angle of the force below horizontal is

$$
\begin{equation*}
\theta=\tan ^{-1}\left[\frac{F_{\mathrm{G}}+F_{\mathrm{SC}}}{F_{\mathrm{w}}}\right] \tag{25}
\end{equation*}
$$

### 11.2.3 Allowable span length for fiber stress

The maximum allowable span length for fiber stress may be calculated for any given conductor, total force, and allowable stress. If the conductor cross section is not symmetrical about the direction of the total force, calculations should be made for the conductor section modulus in the direction of the total force.

If the end conditions of the bus span are unknown, then Equation (26) should be used.

### 11.2.3.1 Two pinned ends

For a single span with two pinned ends, the allowable span length is calculated with Equation 26.

$$
\begin{equation*}
L_{\mathrm{S}}=C \sqrt{\frac{8 F_{\mathrm{A}} S}{F_{\mathrm{T}}}} \tag{26}
\end{equation*}
$$

where

$$
\begin{aligned}
& L_{S}=\text { maximum allowable length, } \mathrm{cm}[\mathrm{in}] \\
& C=3.16 \text { for Metric units }[3.46 \text { for English units }] \\
& F_{A}=\text { maximum allowable stress, } \mathrm{kPa}^{2}[\mathrm{lbf} / \mathrm{in}] \\
& S=\text { section modulus, } \mathrm{cm}^{3}\left[\mathrm{in}^{3}\right] \\
& F_{T}=\text { total force, } \mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]
\end{aligned}
$$

The maximum bending moment will occur at the middle of the span.

### 11.2.3.2 Single-span bus, two fixed ends

For a span with two fixed ends, the allowable span-length equation based on fiber stress is:

$$
\begin{equation*}
L_{\mathrm{S}}=C \sqrt{\frac{12\left(F_{\mathrm{A}}\right)(S)}{F_{\mathrm{T}}}} \tag{27}
\end{equation*}
$$

where

$$
C=3.16 \text { for Metric units [3.46 for English units] }
$$

All other variables have been defined previously.

### 11.2.3.3 Single-span bus, one pinned end, one fixed end

For a single span with one pinned end and one fixed end, the maximum allowable span based on fiber stress may be calculated as follows:

$$
\begin{equation*}
L_{\mathrm{S}}=C \sqrt{\frac{8\left(F_{\mathrm{A}}\right)(S)}{F_{\mathrm{T}}}} \tag{28}
\end{equation*}
$$

where
$C=3.16$ for Metric units [3.46 for English units]

All other variables have been defined previously.
The maximum bending moment will occur at the fixed end of the span.

### 11.2.3.4 Continuous-span bus

Equation (29a), Equation (29b), and Equation (29c), as follows, may be used to calculate the maximum allowable span length based on fiber stress for a different number of spans.

| Number of <br> Spans | Equation | Equation <br> Number |
| :--- | :--- | :--- |
| Two-Span Bus | $L_{\mathrm{S}}=C \sqrt{\frac{8\left(F_{\mathrm{A}}\right)(S)}{F_{\mathrm{T}}}}$ | $(29 \mathrm{a})$ |
| Three-Span Bus | $L_{\mathrm{S}}=C \sqrt{\frac{10\left(F_{\mathrm{A}}\right)(S)}{F_{\mathrm{T}}}}$ | $(29 \mathrm{~b})$ |
| Four-Span Bus | $L_{\mathrm{S}}=C \sqrt{\frac{28\left(F_{\mathrm{A}}\right)(S)}{F_{\mathrm{T}}}}$ |  |

NOTE-The allowable span length is limited by the maximum fiber stress that occurs at the second support from each end. Equation ( 29 c ) can be used conservatively for continuous bus with more than a four-span length. $L_{\mathrm{S}}, F_{\mathrm{A}}, F_{\mathrm{T}}$, and $S$ are as defined earlier.

### 11.3 Maximum allowable span length

The maximum allowable span length $L_{\mathrm{A}}$ is equal to $L_{\mathrm{S}}$ or $L_{\mathrm{D}}$, whichever is shorter.

## 12. Insulator strength considerations

Since the forces on the bus conductors are transmitted to the insulators, the strength of the insulators must be considered. With various bus configurations, insulators may be required to withstand cantilever, compressive, tensile, and torsional forces. Only cantilever forces have been considered in this guide. However, other forces (tension, torsion, and compression) may be critical, requiring consideration in the design.

### 12.1 Insulator cantilever forces

The insulator cantilever force for some common bus arrangements can be given as a function of the effective conductor span length supported by the insulator and the external forces on the bus and insulator. The external forces are
a) The fault current force on the bus
b) The wind force on the bus and insulator
c) The gravitational forces on the bus, insulator or concentrated masses, or both

The effective conductor span length $L_{\mathrm{E}}$ depends on the span length and the bus-support conditions. Use Table 5 to find $L_{\mathrm{E}}$ for each particular support condition and the number of spans. If the support conditions are not known, then take the support condition that yields the maximum span length $L_{\mathrm{S}}$ calculated in Clause 11.

## Table 5-Maximum effective bus span length $L_{E}$ supported by insulator for common bus arrangements ${ }^{\text {a }}$

| Bus <br> Configuration | Support Conditions |  |  |  |  | Maximum Span Length $\boldsymbol{L}_{\mathrm{E}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | S1 | S2 | S3 | S4 | S5 |  |
| Single-Span | P | P |  |  |  | (1/2)L |
| Single-Span | P | F |  |  |  | (5/8)L (Max at S2) |
| Single-Span | F | F |  |  |  | (1/2)L |
| Two Cont.-Span | P | C | P |  |  | (5/4)L (Max at S2) |
| Two Cont.-Span | P | F | F |  |  | (9/8)L (Max at S2) |
| Two Cont.-Span | F | F | F |  |  | $L$ (Max at S2) |
| Three Cont.-Span | P | C | C | P |  | 11/10 L (Max at S2) |
| Four Cont.-Span | P | C | C | C | P | 32/28 L (Max at S2 |

${ }^{\mathrm{a}} L=$ Bus Span Length - Equal Spans for two or more spans.
$L_{\mathrm{E}}=$ Maximum Effective Span Length.
$\mathrm{P}=$ Pinned Support
F = Fixed Support
C $=$ Mid-Support of Continuous Span

NOTE-This table is applicable only to equal span bus arrangement. The mid-support of a continuous bus has only reaction force, but no moment although the continuous bus conductor has a moment at the support point. See Annex F for the method of calculating insulator cantilever forces for individual support of all continuous-bus arrangements. For continuous spans of more than the spans shown, use the equation for the largest span shown for the same end conditions.

### 12.1.1 Bus short-circuit current force

The short-circuit current force transmitted to the bus-support fitting can be calculated using Equation (30).

$$
\begin{equation*}
F_{\mathrm{SB}}=L_{\mathrm{E}} F_{\mathrm{SC}} \tag{30}
\end{equation*}
$$

where
$F_{\mathrm{SB}}=$ bus fault current force transmitted to bus-support fitting, $\mathrm{N}[\mathrm{lbf}]$
$L_{\mathrm{E}}=$ effective bus span length, $\mathrm{m}[\mathrm{ft}]$ (See Table 5)
$F_{\mathrm{SC}}=$ fault current unit force as calculated in Clause $10, \mathrm{~N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]$
If the end conditions are unknown, then the fixed end conditions at the bus-support fitting in question and pinned end conditions at the opposite ends of the adjacent spans will yield the maximum effective bus span length. The adjacent bus span lengths $L 1$ and $L 2$ should be equal to or less than the maximum allowable span length $L_{\mathrm{D}}$ calculated in Clause 11.

### 12.1.2 Bus wind force

The unit wind force associated with the bus span is the same as that described in Clause 9. The wind force transmitted to the bus-support fitting can be calculated using Equation (31).

$$
\begin{equation*}
F_{\mathrm{WB}}=L_{\mathrm{E}} F_{\mathrm{W}} \tag{31}
\end{equation*}
$$

where
$F_{\mathrm{WB}}=$ bus wind force transmitted to bus-support fitting, N [lbf]
$L_{\mathrm{E}}=$ effective bus span length, $\mathrm{m}[\mathrm{ft}]$ (see Table 5)
$F_{\mathrm{W}}=$ wind unit force on the bus, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]$

### 12.1.3 Insulator wind force

The wind force on the bus-support insulator is a function of
a) The insulator dimensions
b) The wind speed
c) The gust factor
d) The radial ice thickness
e) The mounting height
f) Exposure to wind

The wind force acting on the center of an insulator can be calculated using Equation (32).

$$
\begin{equation*}
F_{\mathrm{WI}}=C C_{\mathrm{D}} K_{\mathrm{Z}} G_{\mathrm{F}} V^{2}\left(D_{\mathrm{i}}+2 r_{\mathrm{I}}\right) H_{\mathrm{i}} \tag{32}
\end{equation*}
$$

where
$F_{\mathrm{WI}}=$ wind force on insulator, N [lbf]
$C=6.13 \times 10-3$ for Metric units [ $2.132 \times 10^{-4}$ for English]
$C_{\mathrm{D}}=$ drag coefficient
$K_{\mathrm{Z}}=$ height and exposure factor
$G_{\mathrm{F}}=$ gust factor
$V=$ wind speed at 30 ft [ 9.1 m ] above ground, $\mathrm{km} / \mathrm{h}$ [ $\mathrm{mi} / \mathrm{h}$ ]
$D_{\mathrm{i}}=$ effective insulator diameter, $\mathrm{cm}[\mathrm{in}]$
$r_{\mathrm{I}}=$ radial ice thickness, cm [in]
$H_{\mathrm{i}}=$ insulator height, cm [in] (see Fig 5)
$r_{\mathrm{I}}, K_{\mathrm{Z}}, G_{\mathrm{F}}$, and $V$ are the same factors used for the wind force on the bus conductor (see Clause 9). $C_{\mathrm{D}}$ is usually considered as unity.

The effective insulator diameter $D_{\mathrm{i}}$ is usually considered as the insulator diameter over the skirts. For tapered insulators the effective diameter is the average diameter and can be calculated using Equation (33).

$$
\begin{equation*}
D_{\mathrm{i}}=\frac{D_{1}+D_{2}+\ldots \ldots \ldots D_{\mathrm{n}}}{n} \tag{33}
\end{equation*}
$$

where
$D_{1}, D_{2}$, and $D_{\mathrm{n}}=$ outside diameters of each subassembly for the $1 \mathrm{st}, 2 \mathrm{nd}$, and nth sections of the insulator (see Figure 5).

The total wind force $F_{\text {WI }}$ on a uniform-diameter insulator acts at the center of the insulator (see Figure 5). For a tapered insulator, the total wind force is usually considered acting at the center $H_{\mathrm{i}} / 2$ since the resulting error is of small magnitude and is conservative.

### 12.1.4 Gravitational forces

In some rigid-bus structure configurations, the insulator may be subjected to cantilever gravitational forces. These forces should be added vectorially to the fault current and wind forces. These gravitational forces will be due to the mass of the supported rigid bus, the mass of the insulator itself or other concentrated masses, or both.

The effective weight of the bus mass transmitted to the bus-support fitting can be determined using Equation (34).

$$
\begin{equation*}
F_{\mathrm{GB}}=L_{\mathrm{E}} F_{\mathrm{G}} \tag{34}
\end{equation*}
$$

where
$F_{\mathrm{GB}}=$ effective weight of bus transmitted to bus-support fitting, N [lbf]
$L_{\mathrm{E}}=$ effective bus span length, $\mathrm{m}[\mathrm{ft}]$ (see Table 5)
$F_{\mathrm{G}}=$ total bus unit weight, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{ft}]$

If the bus span is subjected to concentrated loads, the force transmitted to the bus-support fitting should be analyzed more thoroughly.

The weight of the insulator $F_{\mathrm{GI}}$ should be included in the total cantilever force if the insulator is not mounted vertically.

### 12.1.5 Total insulator cantilever load

The total cantilever load on an insulator is the summation of the cantilever forces acting on the insulator multiplied by their overload factors.

The total cantilever load on a vertically-mounted insulator supporting a horizontal bus (see Figure 5) can be calculated using Equation (35).

$$
\begin{equation*}
F_{\mathrm{IS}}=K_{1}\left[\frac{F_{W I}}{2}+\frac{\left(H_{\mathrm{i}}+H_{\mathrm{f}}\right) F_{\mathrm{WB}}}{H_{\mathrm{i}}}\right]+K_{2}\left[\frac{\left(H_{\mathrm{i}}+H_{\mathrm{f}}\right) F_{\mathrm{SB}}}{H_{\mathrm{i}}}\right] \tag{35}
\end{equation*}
$$

where
$F_{\text {IS }}=$ total cantilever load acting at end of insulator, $\mathrm{N}[\mathrm{lbf}]$
$F_{\mathrm{WI}}=$ wind force on the insulator, N [lbf]
$F_{\mathrm{SB}}=$ short-circuit current force transmitted to bus-support fitting, N [lbf]
$F_{\mathrm{WB}}=$ bus wind force transmitted to the bus-support fitting, N [lbf]
$H_{\mathrm{i}}=$ insulator height, cm [in]
$H_{\mathrm{f}}=$ bus centerline height above the insulator, cm [in]
$K_{1}=$ overload factor applied to wind forces
$K_{2}=$ overload factor applied to short-circuit current forces


Figure 5-Vertically mounted insulator cantilever forces

The total cantilever load on a horizontally mounted insulator with a horizontal bus (see Figure 6) can be calculated using Equation (36).

$$
\begin{equation*}
F_{\mathrm{IS}}=K_{3}\left[\frac{F_{\mathrm{GI}}}{2}+\frac{\left(H_{\mathrm{i}}+H_{\mathrm{f}}\right) F_{\mathrm{GB}}}{H_{\mathrm{i}}}\right]+K_{2}\left[\frac{\left(H_{\mathrm{i}}+H_{\mathrm{f}}\right) F_{\mathrm{SB}}}{H_{\mathrm{i}}}\right] \tag{36}
\end{equation*}
$$

where

$$
\begin{aligned}
& F_{I S}=\text { total cantilever load acting at end of insulator, } \mathrm{N}[\mathrm{lbf}] \\
& F_{G I}=\text { weight of insulator, } \mathrm{N}[\mathrm{lbf}] \\
& F_{G B}=\text { effective weight of bus transmitted to bus-support fitting, } \mathrm{N}[\mathrm{lbf}] \\
& F_{S B}=\text { short-circuit current force transmitted to bus-support fitting, } \mathrm{N}[\mathrm{lbf}] \\
& H_{i}=\text { insulator height, } \mathrm{cm}[\mathrm{in}] \\
& H_{f}=\text { bus centerline distance beyond insulator, } \mathrm{cm}[\mathrm{in}] \\
& K_{2}=\text { overload factor applied to fault current forces } \\
& K_{3}=\text { overload factor applied to gravitational forces }
\end{aligned}
$$

Equations (34), (35), and (36) cover the most common bus and insulator configurations. The designer should examine each configuration to ensure the proper summation of forces acting on the insulator.


Figure 6-Horizontally mounted insulator cantilever forces

### 12.2 Insulator force overload factors

Porcelain, unlike metal, is very brittle. The yield and tensile strengths of porcelain have identical values. Because porcelain cannot yield without cracking, an overload factor should be applied to the loads on the insulator.

A conservative value of 2.5 is recommended for overload factors $K_{1}$ and $K_{3}$ (wind and gravitational forces) by some US insulator manufacturers.

The value of overload factor $K_{2}$ (fault current forces) depends upon the natural frequencies of the insulator, of the insulator/mounting structure combination, and of the conductor span. Since the force $F_{\mathrm{SC}}$ is conservative, a value of 1.0 can be used for $K_{2}$ if
a) The natural frequency of the insulator, together with the effective weight of the conductor span $f_{\mathrm{i}}$, is less than one half the short-circuit current-force frequency, that is

$$
\begin{equation*}
f_{\mathrm{i}}<\frac{120}{2} \mathrm{~Hz} \text { for a } 60 \mathrm{~Hz} \text { system } \tag{37}
\end{equation*}
$$

where
$f_{\mathrm{i}}=$ natural frequency of insulator with effective weight of conductor span, Hz
b) The natural frequencies of the insulator/mounting structure combination $f_{\mathrm{s} 1}$ and $f_{\mathrm{s} 2}$ and the natural frequency of the conductor span $f_{\mathrm{b}}$ differ by a factor of at least two, that is

$$
\begin{align*}
& \frac{f_{\mathrm{s} 1}}{f_{\mathrm{b}}}<\frac{1}{2} \text { or } \frac{f_{\mathrm{s} 1}}{f_{\mathrm{b}}}>2  \tag{37a}\\
& \frac{f_{\mathrm{s} 2}}{f_{\mathrm{b}}}<\frac{1}{2} \text { or } \frac{f_{\mathrm{s} 2}}{f_{\mathrm{b}}}>2 \tag{37b}
\end{align*}
$$

where

$$
\begin{aligned}
& f_{\mathrm{s} 1}=\text { first natural frequency of insulator/mounting structure combination, } \mathrm{Hz} \\
& f_{\mathrm{s} 2}=\text { second natural frequency of insulator/mounting structure combination, } \mathrm{Hz} \\
& f_{\mathrm{b}}=\text { natural frequency of the conductor span, } \mathrm{Hz}
\end{aligned}
$$

If either of these conditions is not satisfied, a dynamic study should be made to determine an appropriate overload factor, or an overload factor of 2.5 should be used.

The natural frequency of the insulator together with the effective weight of the conductor span can be calculated using Equation (38).

$$
\begin{equation*}
f_{\mathrm{i}}=\frac{1}{2 \pi} \sqrt{\frac{K_{\mathrm{i}} \mathrm{~g}}{0.226 F_{\mathrm{GI}}+F_{\mathrm{GB}}}} \tag{38}
\end{equation*}
$$

where

```
\(f_{\mathrm{i}}=\) natural frequency of insulator with effective weight of conductor span, Hz
\(K_{\mathrm{i}}=\) insulator cantilever spring constant, \(\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{in}]\)
\(\mathrm{g}=\) gravitational constant, \(9.81 \mathrm{~m} / \mathrm{s}^{2}\left[386 \mathrm{in} / \mathrm{s}^{2}\right]\)
\(F_{\mathrm{GI}}=\) weight of insulator, \(\mathrm{N}[\mathrm{lbf}]\)
\(F_{\mathrm{GB}}=\) effective weight of bus transmitted to bus-support fitting, N [lbf]
```

The natural frequencies of the insulator/mounting structure combination $f_{s 1}$ and $f_{s 2}$ can be calculated using Equations (39) and (40).

$$
\begin{align*}
& f_{\mathrm{s} 1}=\frac{1}{2 \pi} \sqrt{\frac{K_{\mathrm{i}}+K_{\mathrm{s}}}{2 m_{1}}+\frac{K_{\mathrm{i}}}{2 m_{2}}-\frac{1}{2} \sqrt{\left(\frac{K_{\mathrm{i}}+K_{\mathrm{s}}}{m_{1}}+\frac{K_{\mathrm{i}}}{m_{2}}\right)^{2}-\frac{4 K_{\mathrm{i}} K_{\mathrm{s}}}{m_{1} m_{2}}}}  \tag{39}\\
& f_{\mathrm{s} 2}=\frac{1}{2 \pi} \sqrt{\frac{K_{\mathrm{i}}+K_{\mathrm{s}}}{2 m_{1}}+\frac{K_{\mathrm{i}}}{2 m_{2}}-\frac{1}{2} \sqrt{\left(\frac{K_{\mathrm{i}}+K_{\mathrm{s}}}{m_{1}}+\frac{K_{\mathrm{i}}}{m_{2}}\right)^{2}-\frac{4 K_{\mathrm{i}} K_{\mathrm{s}}}{m_{1} m_{2}}}} \tag{40}
\end{align*}
$$

where
$F_{\text {s1 }}=$ first natural frequency of insulator/mounting structure combination, Hz
$F_{\mathrm{s} 2}=$ second natural frequency of insulator/mounting structure combination, Hz
$K_{\mathrm{i}}=$ insulator cantilever spring constant, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{in}]$
$K_{\mathrm{S}}=$ mounting structure cantilever spring constant, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{in}]$

$$
\begin{aligned}
& m_{1}=\frac{0.333 F_{\mathrm{GS}}+0.5 F_{\mathrm{GI}}}{\mathrm{~g}} \\
& m_{2}=\frac{F_{\mathrm{GB}}+0.226 F_{\mathrm{GS}}}{\mathrm{~g}}
\end{aligned}
$$

where

$$
\begin{aligned}
& F_{\mathrm{GS}}=\text { weight of mounting structure, } \mathrm{N}[\mathrm{lbf}] \\
& F_{\mathrm{GI}}=\text { weight of insulator, } \mathrm{N}[\mathrm{lbf}] \\
& F_{\mathrm{GB}}=\text { weight of bus, } \mathrm{N}[\mathrm{lbf}] \\
& \mathrm{g}=\text { gravitational constant, } 9.81 / \mathrm{s}^{2}\left[386 \mathrm{in} / \mathrm{s}^{2}\right]
\end{aligned}
$$

The cantilever spring constant for the insulator can be obtained from insulator manufacturers. The cantilever spring constant for a single-phase mounting structure with a constant cross section can be calculated using Equation (41).

$$
\begin{equation*}
K_{\mathrm{S}}=C \frac{3 E J}{H_{\mathrm{s}}{ }^{3}} \tag{41}
\end{equation*}
$$

where
$K_{\mathrm{S}}=$ support cantilever spring constant, $\mathrm{N} / \mathrm{m}[\mathrm{lbf} / \mathrm{in}]$
$C=0.01$ for Metric units [1 for English units]
$E=$ modulus of elasticity, $\mathrm{N} / \mathrm{m}\left[\mathrm{lbf} / \mathrm{in}^{2}\right]$
$J=$ cross-sectional moment of inertia, $\mathrm{cm}^{4}\left[\mathrm{in}^{4}\right]$
$H_{\mathrm{s}}=$ mounting structure length, cm [in]

### 12.3 Minimum insulator cantilever strength

The minimum published insulator cantilever strength required is

$$
\begin{equation*}
S_{\mathrm{I}} \geq F_{\mathrm{IS}} \tag{42}
\end{equation*}
$$

where
$S_{\mathrm{I}}=$ minimum published insulator cantilever strength, $\mathrm{N}[\mathrm{lbf}]$
$F_{\text {IS }}=$ total cantilever load acting at end of insulator, N [lbf]

## 13. Conductor thermal expansion considerations

When the temperature of a bus conductor is changed, a corresponding change in length results. This change in length can be calculated as

$$
\begin{equation*}
\Delta L=\frac{\alpha L_{\mathrm{i}}\left(T_{\mathrm{f}}-T_{\mathrm{i}}\right)}{1+\alpha T_{\mathrm{i}}} \tag{43}
\end{equation*}
$$

where
$\Delta L=$ change in span length, $\mathrm{m}[\mathrm{ft}]$
$\alpha=$ coefficient of thermal expansion, $1 /{ }^{\circ} \mathrm{C}$
$T_{\mathrm{i}}=$ initial installation temperature, ${ }^{\circ} \mathrm{C}$
$T_{\mathrm{f}}=$ final temperature, ${ }^{\circ} \mathrm{C}$
$L_{\mathrm{i}}=$ span length at the initial temperature, $\mathrm{m}[\mathrm{ft}]$

### 13.1 Thermal loads

If the ends of the conductor are fixed, preventing expansion or contraction, and the conductor temperature is changed, compressive or tensile forces will result. These forces can be computed as

$$
\begin{equation*}
F_{\mathrm{TE}}=C A E \frac{\Delta L}{L_{1}}=C A E \alpha\left(T_{\mathrm{i}}-T_{\mathrm{f}}\right) \tag{44}
\end{equation*}
$$

where

```
\(F_{\mathrm{TE}}=\) thermal force, \(\mathrm{N}[\mathrm{lbf}]\)
\(C=0.1\) for Metric units [1 for English units]
\(A=\) cross-sectional area of the conductor, \(\mathrm{cm}^{2}\left[\mathrm{in}^{2}\right]\)
\(E=\) modulus of elasticity, \(\mathrm{kPa}\left[\mathrm{lbf} / \mathrm{in}^{2}\right]\)
\(\Delta L=\) change in span length, \(\mathrm{m}[\mathrm{ft}]\)
\(L_{\mathrm{i}}=\) span length at the initial temperature, \(\mathrm{m}[\mathrm{ft}]\)
\(\alpha=\) coefficient of thermal expansion, \(1 /{ }^{\circ} \mathrm{C}\)
\(T_{\mathrm{i}}=\) initial installation temperature, \({ }^{\circ} \mathrm{C}\)
\(T_{\mathrm{f}}=\) final temperature, \({ }^{\circ} \mathrm{C}\)
```

The force calculated using Equation (44) does not consider the flexibility of mounting structures or bus structure. Since this flexibility will allow some expansion or contraction of the bus conductor, the forces experienced will be less than the force calculated above.

### 13.2 Expansion fittings

Since the thermal forces exerted on the bus conductor are independent of span length, provisions should be made for expansion in any bus-conductor span. These provisions may be made with expansion fittings for long buses, or by considering deflection of a bus conductor, bus-conductor bends, insulators, or mounting structures for short buses.

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## Annex A

## (informative)

## Letter symbols for quantities

| Symbol | Meaning |
| :---: | :---: |
| A | cross-sectional area, $\mathrm{cm}^{2}\left[\mathrm{in}^{2}\right]$ |
| C | temperature, ${ }^{\circ} \mathrm{C}$ |
| $C_{\text {D }}$ | drag coefficient |
| D | conductor spacing, center-to-center, cm [in] |
| $D_{\text {i }}$ | effective insulator diameter, cm [in] |
| d | conductor outside diameter, cm [in] |
| $E$ | modulus of elasticity kPA [lbf/in ${ }^{2}$ ] |
| F | temperature, ${ }^{\circ} \mathrm{F}$ |
| $F$ | skin-effect coefficient |
| $F_{\text {A }}$ | maximum allowable stress, $\mathrm{Nm}^{2}\left[\mathrm{lbf} / \mathrm{in}^{2}\right]$ |
| $F_{\text {c }}$ | conductor unit weight, $\mathrm{Nm}[\mathrm{lbf} / \mathrm{ft}]$ |
| $F_{\text {D }}$ | damping material unit weight, $\mathrm{Nm}[\mathrm{lbf} / \mathrm{ft}]$ |
| $F_{\text {G }}$ | total bus unit weight, Nm [lbf/ft] |
| $F_{\text {GB }}$ | effective weight of bus transmitted to bus-support fitting, N [lbf] |
| $F_{\text {GI }}$ | weight of insulator, N [lbf] |
| $F_{\text {GS }}$ | weight of mounting structure, N [lbf] |
| $F_{\text {I }}$ | ice unit weight, $\mathrm{Nm}[\mathrm{lbf} / \mathrm{ft}]$ |
| $F_{\text {is }}$ | total cantilever load acting at end of insulator, N [lbf] |
| $F_{\text {SB }}$ | short-circuit current-force transmitted to bus-support fitting, N [lbf] |
| $F_{\text {SC }}$ | fault current unit force, Nm [lbf/ft] |
| $F_{\text {T }}$ | total unit force on the bus, $\mathrm{Nm}[\mathrm{lbf} / \mathrm{ft}]$ |
| $F_{\text {TE }}$ | thermal force, N [lbf] |
| $F_{\text {W }}$ | wind unit force on the bus, $\mathrm{Nm}[\mathrm{lbf} / \mathrm{ft}]$ |
| $F_{\text {WB }}$ | bus wind force transmitted to bus-support fitting, N [lbf] |
| $F_{\text {WI }}$ | wind force on insulator, N [lbf] |
| $f_{\text {a }}$ | maximum aeolian vibration frequency, Hz |
| $f_{\mathrm{b}}$ | natural frequency of bus span, Hz |
| $f_{\text {i }}$ | natural frequency of insulator together with effective weight of bus span, Hz |
| $f_{\text {s1 }}, f_{\text {s2 }}$ | natural frequencies of insulator together with mounting structure, Hz |
| $G$ | conductivity, \%IACS |
| g | gravitational constant |
| $G_{\text {F }}$ | gust factor |
| $H_{\text {f }}$ | bus centerline distance above top of insulator, cm [in] |
| $H_{\mathrm{i}}$ | insulator height, cm [in] |
| $H_{\text {s }}$ | mounting structure height, cm [in] |
| I | current, A, rms |
| $I_{\text {sc }}$ | symmetrical short-circuit current, A, rms |
| $J$ | moment of inertia of cross-sectional area, $\mathrm{cm}^{4}\left[\mathrm{in}^{4}\right]$ |
| K | constant used in span natural frequency calculation and dependent upon end conditions |
| $K_{\text {f }}$ | mounting structure flexibility factor |
| $K_{\text {i }}$ | insulator cantilever spring constant, Nm [lbf/in] |
| $K_{\text {s }}$ | mounting structure cantilever spring constant, Nm [lbf/in] |
| $K_{\text {Z }}$ | height and exposure factor |
| $K_{1}, K_{2}, K_{3}$ | insulator overload factors |


| $L$ | span length, m [ft] |
| :---: | :---: |
| $L_{\text {A }}$ | maximum allowable bus span length, m [ft] |
| $L_{\text {D }}$ | maximum allowable bus span length based on vertical deflection, cm [in] |
| $L_{\mathrm{E}}$ | effective bus span length, m [ft] |
| $L_{\text {i }}$ | span length at initial temperature $T_{i}, \mathrm{~m}[\mathrm{ft}]$ |
| $L_{\text {S }}$ | maximum allowable bus span length based on fiber stress, cm [in] |
| $L_{1}, L_{2}$ | adjacent bus span lengths, m [ft] |
| lbf | pound force [N] |
| lbf | pound mass [N] |
| m | mass per unit length, $\mathrm{kg} / \mathrm{m}[\mathrm{lbm} / \mathrm{ft}]$ |
| $q_{\text {c }}$ | convective heat loss, W/m [W/ft] |
| $q_{\text {cond }}$ | conductive heat loss, W/m [W/ft] |
| $q_{\mathrm{r}}$ | radiation heat loss, W/m [W/ft] |
| $q_{\text {s }}$ | solar heat gain, W/m [W/ft] |
| $R$ | conductor direct-current resistance, S/m [S/ft] |
| $r_{\text {I }}$ | radial ice thickness, w/m [in] |
| $S$ | section modulus, $\mathrm{cm}^{3}\left[\mathrm{in}^{3}\right]$ |
| $S_{\text {I }}$ | minimum published insulator cantilever strength, N [lbf] |
| $T_{\text {f }}$ | final conductor temperature, ${ }^{\circ} \mathrm{C}$ |
| $T_{\mathrm{i}}$ | initial conductor temperature, ${ }^{\circ} \mathrm{C}$ |
| $t$ | time, s |
| V | wind speed, $\mathrm{km} / \mathrm{h}[\mathrm{mi} / \mathrm{h}]$ |
| $W_{\text {I }}$ | ice weight, $\mathrm{N} / \mathrm{cm}^{3}\left[\mathrm{lbf} / \mathrm{in}^{3}\right]$ |
| $Y_{\text {A }}$ | maximum allowable deflection, in [cm] |
| $Y_{\text {B }}$ | maximum allowable deflection as a fraction of span length |
| $\alpha$ | coefficient of thermal expansion |
| $\Delta L$ | $L$ change in span length, m [ft] |
| $\theta$ | angle of total force below horizontal, degrees |
| $\lambda$ | ratio of span length to vertical dimension of bus conductor |
| $\Gamma$ | multiplying factor based on type of short-circuit current |

## Annex B

## (informative)

## Bus-conductor ampacity

The bus-ampacity data included in this annex have been taken from Thermal Consideration for Outdoor Bus-Conductor Design Ampacity Tables, Substation Committee of the IEEE Power Engineering Society. ${ }^{8}$

Table B.1-Single aluminum rectangular bar AC ampacity, with sun (55.0\% conductivity)

| Size (in) | Emissivity $=\mathbf{0 . 2 0}$Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.50$Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 0.250 by 4.000 | 1130 | 1298 | 1441 | 1566 | 1678 | 1872 | 2039 | 1200 | 1394 | 1560 | 1707 | 1839 | 2073 | 2278 |
| 0.250 by 5.000 | 1320 | 1517 | 1685 | 1833 | 1965 | 2195 | 2393 | 1413 | 1644 | 1841 | 2016 | 2174 | 2455 | 2703 |
| 0.250 by 6.000 | 1497 | 1723 | 1915 | 2084 | 2235 | 2500 | 2729 | 1615 | 1881 | 2109 | 2311 | 2495 | 2821 | 3110 |
| 0.375 by 4.000 | 1385 | 1593 | 1769 | 1924 | 2063 | 2304 | 2510 | 1464 | 1704 | 1909 | 2091 | 2254 | 2544 | 2799 |
| 0.375 by 5.000 | 1608 | 1851 | 2057 | 2239 | 2401 | 2686 | 2931 | 1714 | 1998 | 2241 | 2456 | 2651 | 2997 | 3302 |
| 0.375 by 6.000 | 1815 | 2091 | 2326 | 2533 | 2718 | 3044 | 3326 | 1950 | 2275 | 2554 | 2801 | 3026 | 3426 | 3782 |
| 0.375 by 8.000 | 2202 | 2540 | 2829 | 3084 | 3313 | 3718 | 4070 | 2395 | 2800 | 3148 | 3458 | 3740 | 4247 | 4700 |
| 0.500 by 4.000 | 1589 | 1829 | 2034 | 2213 | 2374 | 2654 | 2895 | 1672 | 1951 | 2189 | 2399 | 2590 | 2926 | 3223 |
| 0.500 by 5.000 | 1835 | 2115 | 2353 | 2562 | 2750 | 3079 | 3364 | 1949 | 2276 | 2556 | 2805 | 3030 | 3430 | 3785 |
| 0.500 by 6.000 | 2071 | 2388 | 2659 | 2897 | 3111 | 3487 | 3814 | 2216 | 2590 | 2912 | 3197 | 3456 | 3918 | 4330 |
| 0.500 by 8.000 | 2511 | 2899 | 3231 | 3524 | 3788 | 4255 | 4662 | 2721 | 3186 | 3587 | 3943 | 4268 | 4851 | 5374 |
| 0.625 by 4.000 | 1776 | 2047 | 2277 | 2479 | 2660 | 2977 | 3249 | 1861 | 2177 | 2446 | 2683 | 2898 | 3278 | 3614 |
| 0.625 by 5.000 | 2034 | 2347 | 2613 | 2847 | 3058 | 3427 | 3747 | 2152 | 2519 | 2833 | 3111 | 3363 | 3812 | 4210 |
| 0.625 by 6.000 | 2286 | 2639 | 2940 | 3206 | 3445 | 3865 | 4231 | 2437 | 2855 | 3213 | 3531 | 3820 | 4337 | 4798 |
| 0.625 by 8.000 | 2760 | 3190 | 3558 | 3884 | 4177 | 4696 | 5151 | 2982 | 3498 | 3942 | 4337 | 4698 | 5347 | 5929 |
| 0.625 by 10.000 | 3190 | 3690 | 4120 | 4501 | 4845 | 5457 | 5996 | 3483 | 4091 | 4615 | 5084 | 5513 | 6238 | 6987 |
| 0.625 by 12.000 | 3560 | 4123 | 4608 | 5039 | 5430 | 6126 | 6744 | 3924 | 4615 | 5212 | 5748 | 6240 | 7131 | 7941 |
| 0.750 by 4.000 | 1935 | 2232 | 2486 | 2708 | 2907 | 3256 | 3557 | 2021 | 2368 | 2664 | 2926 | 3163 | 3582 | 3953 |
| 0.750 by 5.000 | 2216 | 2559 | 2851 | 3108 | 3340 | 3746 | 4098 | 2336 | 2740 | 3085 | 3391 | 3668 | 4162 | 4601 |
| 0.750 by 6.000 | 2472 | 2856 | 3184 | 3474 | 3735 | 4195 | 4597 | 2627 | 3083 | 3474 | 3821 | 4137 | 4702 | 5207 |
| 0.750 by 8.000 | 2984 | 3452 | 3852 | 4207 | 4527 | 5094 | 5592 | 3214 | 3776 | 4260 | 4691 | 5085 | 5793 | 6430 |
| 0.750 by 10.000 | 3518 | 4072 | 4548 | 4969 | 5350 | 6026 | 6622 | 3832 | 4505 | 5086 | 5605 | 6079 | 6935 | 7708 |
| 0.750 by 12.000 | 3875 | 4491 | 5021 | 5492 | 5919 | 6682 | 7359 | 4260 | 5015 | 5669 | 6255 | 6793 | 7768 | 8655 |

[^4]Table B.2-Single aluminum rectangular bar AC ampacity, without sun (55.0\% conductivity)

| Size (in) | Emissivity $=\mathbf{0 . 2 0}$ <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.50$Temperature Rise Above $40^{\circ}$ C Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 0.250 by 4.000 | 1158 | 1322 | 1462 | 1585 | 1695 | 1887 | 2052 | 1265 | 1449 | 1608 | 1749 | 1877 | 2105 | 2306 |
| 0.250 by 5.000 | 1354 | 1546 | 1711 | 1856 | 1986 | 2213 | 2409 | 1492 | 1710 | 1899 | 2068 | 2221 | 2494 | 2737 |
| 0.250 by 6.000 | 1538 | 1757 | 1945 | 2111 | 2260 | 2521 | 2747 | 1708 | 1959 | 2177 | 2372 | 2549 | 2867 | 3150 |
| 0.375 by 4.000 | 1423 | 1625 | 1798 | 1950 | 2086 | 2324 | 2528 | 1553 | 1780 | 1975 | 2149 | 2308 | 2589 | 2838 |
| 0.375 by 5.000 | 1654 | 1890 | 2092 | 2270 | 2429 | 2710 | 2952 | 1821 | 2087 | 2319 | 2526 | 2714 | 3050 | 3349 |
| 0.375 by 6.000 | 1869 | 2136 | 2366 | 2569 | 2751 | 3072 | 3350 | 2073 | 2378 | 2644 | 2882 | 3099 | 3488 | 3835 |
| 0.375 by 8.000 | 2271 | 2598 | 2880 | 3130 | 3355 | 3753 | 4102 | 2552 | 2931 | 3262 | 3560 | 3833 | 4324 | 4767 |
| 0.500 by 4.000 | 1638 | 1871 | 2070 | 2246 | 2403 | 2679 | 2917 | 1786 | 2047 | 2273 | 2474 | 2657 | 2984 | 3273 |
| 0.500 by 5.000 | 1893 | 2164 | 2396 | 2601 | 2786 | 3109 | 3390 | 2082 | 2388 | 2654 | 2892 | 3109 | 3497 | 3843 |
| 0.500 by 6.000 | 2137 | 2444 | 2708 | 2941 | 3152 | 3522 | 3844 | 2369 | 2719 | 3024 | 3297 | 3546 | 3995 | 4396 |
| 0.500 by 8.000 | 2595 | 2970 | 3294 | 3580 | 3840 | 4298 | 4701 | 2912 | 3347 | 3726 | 4068 | 4381 | 4946 | 5457 |
| 0.625 by 4.000 | 1836 | 2098 | 2322 | 2520 | 2697 | 3008 | 3277 | 2002 | 2295 | 2549 | 2775 | 2981 | 3349 | 3675 |
| 0.625 by 5.000 | 2104 | 2406 | 2665 | 2894 | 3100 | 3463 | 3778 | 2313 | 2654 | 2951 | 3216 | 3458 | 3893 | 4281 |
| 0.625 by 6.000 | 2365 | 2706 | 2999 | 3259 | 3493 | 3906 | 4267 | 2620 | 3008 | 3346 | 3650 | 3928 | 4428 | 4877 |
| 0.625 by 8.000 | 2859 | 3274 | 3632 | 3949 | 4237 | 4747 | 5196 | 3206 | 3686 | 4106 | 4484 | 4831 | 5459 | 6027 |
| 0.625 by 10.000 | 3307 | 3790 | 4207 | 4579 | 4917 | 5518 | 6050 | 3748 | 4313 | 4809 | 5257 | 5669 | 6420 | 7102 |
| 0.625 by 12.000 | 3696 | 4239 | 4709 | 5129 | 5512 | 6196 | 6805 | 4227 | 4869 | 5434 | 5945 | 6418 | 7282 | 8072 |
| 0.750 by 4.000 | 2006 | 2293 | 2539 | 2756 | 2951 | 3293 | 3589 | 2188 | 2509 | 2787 | 3035 | 3262 | 3666 | 4026 |
| 0.750 by 5.000 | 2298 | 2628 | 2912 | 3163 | 3389 | 3788 | 4135 | 2526 | 2899 | 3224 | 3515 | 3780 | 4257 | 4684 |
| 0.750 by 6.000 | 2564 | 2934 | 3253 | 3535 | 3791 | 4243 | 4639 | 2838 | 3260 | 3628 | 3959 | 4262 | 4808 | 5299 |
| 0.750 by 8.000 | 3097 | 3548 | 3937 | 4283 | 4596 | 5153 | 5644 | 3472 | 3992 | 4448 | 4859 | 5237 | 5921 | 6542 |
| 0.750 by 10.000 | 3655 | 4188 | 4649 | 5060 | 5433 | 6097 | 6684 | 4140 | 4763 | 5311 | 5805 | 6260 | 7088 | 7841 |
| 0.750 by 12.000 | 4030 | 4622 | 5136 | 5595 | 6013 | 6762 | 7429 | 4605 | 5305 | 5921 | 6480 | 6996 | 7940 | 8804 |

Table B.3-Aluminum tubular bus - Schedule 40 AC ampacity

| SPS ${ }^{\text {a }}$ Size | OD | Wall Thickness | Emissivity $=0.20$, With SunTemperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity = 0.20, Without Sun Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | (in) | (in) | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 1.0 | 1.315 | 0.133 | 591 | 688 | 770 | 840 | 903 | 1011 | 1102 | 638 | 728 | 804 | 871 | 931 | 1035 | 1123 |
| 1.5 | 1.900 | 0.145 | 837 | 978 | 1097 | 1199 | 1290 | 1447 | 1580 | 914 | 1043 | 1153 | 1250 | 1336 | 1486 | 1614 |
| 2.0 | 2.375 | 0.154 | 1035 | 1213 | 1362 | 1490 | 1605 | 1802 | 1969 | 1139 | 1300 | 1438 | 1558 | 1666 | 1854 | 2015 |
| 2.5 | 2.875 | 0.203 | 1377 | 1618 | 1818 | 1992 | 2147 | 2413 | 2640 | 1527 | 1743 | 1928 | 2090 | 2235 | 2488 | 2705 |
| 3.0 | 3.500 | 0.216 | 1666 | 1962 | 2208 | 2422 | 2612 | 2940 | 3220 | 1861 | 2126 | 2351 | 2550 | 2728 | 3038 | 3305 |
| 3.5 | 4.000 | 0.226 | 1897 | 2239 | 2523 | 2770 | 2989 | 3367 | 3690 | 2132 | 2435 | 2695 | 2923 | 3127 | 3484 | 3792 |
| 4.0 | 4.500 | 0.237 | 2134 | 2523 | 2847 | 3127 | 3376 | 3807 | 4175 | 2412 | 2755 | 3049 | 3307 | 3539 | 3945 | 4295 |
| 5.0 | 5.563 | 0.258 | 2636 | 3127 | 3536 | 3890 | 4204 | 4748 | 5213 | 3010 | 3439 | 3807 | 4131 | 4422 | 4933 | 5374 |
| 6.0 | 6.625 | 0.280 | 3153 | 3752 | 4250 | 4681 | 5063 | 5726 | 6294 | 3633 | 4152 | 4597 | 4990 | 5343 | 5963 | 6500 |
| 8.0 | 8.625 | 0.322 | 4142 | 4954 | 5629 | 6213 | 6731 | 7631 | 8404 | 4843 | 5538 | 6135 | 6662 | 7138 | 7975 | 8703 |
| SPS ${ }^{\text {a }}$ Size | OD | Wall <br> Thickness |  |  | issivit ure $\mathbf{R}$ | $\begin{gathered} =0.50, \\ \text { Abov } \end{gathered}$ | $\begin{aligned} & V \text { ith } \mathrm{St} \\ & 40^{\circ} \mathrm{C} \end{aligned}$ | bient |  |  |  | sivity ure $\mathbf{R}$ | $\begin{aligned} & \text { 0.50, } \\ & \text { e Abo } \end{aligned}$ | $\begin{aligned} & \text { thout } \\ & \mathbf{4 0}^{\circ} \mathrm{C} \end{aligned}$ | mbient |  |
|  |  |  | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 1.0 | 1.315 | 0.133 | 572 | 690 | 788 | 872 | 948 | 1078 | 1190 | 686 | 785 | 870 | 945 | 1013 | 1133 | 1238 |
| 1.5 | 1.900 | 0.145 | 805 | 981 | 1127 | 1252 | 1363 | 1556 | 1723 | 992 | 1136 | 1260 | 1370 | 1469 | 1645 | 1800 |
| 2.0 | 2.375 | 0.154 | 991 | 1217 | 1402 | 1561 | 1703 | 1949 | 2161 | 1244 | 1425 | 1581 | 1720 | 1845 | 2068 | 2264 |
| 2.5 | 2.875 | 0.203 | 1314 | 1623 | 1876 | 2094 | 2287 | 2623 | 2914 | 1677 | 1921 | 2132 | 2320 | 2490 | 2793 | 3060 |
| 3.0 | 3.500 | 0.216 | 1582 | 1969 | 2284 | 2555 | 2795 | 3214 | 3576 | 2056 | 2357 | 2617 | 2848 | 3059 | 3434 | 3766 |
| 3.5 | 4.000 | 0.226 | 1796 | 2248 | 2614 | 2929 | 3208 | 3694 | 4116 | 2366 | 2712 | 3012 | 3280 | 3523 | 3957 | 4342 |
| 4.0 | 4.500 | 0.237 | 2015 | 2534 | 2954 | 3315 | 3635 | 4192 | 4675 | 2686 | 3080 | 3421 | 3726 | 4004 | 4500 | 4940 |
| 5.0 | 5.563 | 0.258 | 2474 | 3142 | 3680 | 4141 | 4550 | 5262 | 5880 | 3375 | 3872 | 4304 | 4690 | 5041 | 5671 | 6232 |
| 6.0 | 6.625 | 0.280 | 2943 | 3771 | 4435 | 5003 | 5506 | 6382 | 7144 | 4098 | 4703 | 5230 | 5701 | 6131 | 6902 | 7591 |
| 8.0 | 8.625 | 0.322 | 3830 | 4982 | 5899 | 6681 | 7373 | 8581 | 9633 | 5515 | 6334 | 7048 | 7688 | 8274 | 9328 | 10274 |

[^5]Table B.4-Aluminum tubular bus-Schedule 80 AC ampacity

| $\begin{gathered} \text { Size } \\ \text { SPS }^{\text {a }} \end{gathered}$ | OD | Wall <br> Thickness | Emissivity $=0.20$, With SunTemperature Rise Above $40{ }^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.20$, Without SunTemperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| (in) | (in) | (in) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 1.0 | 1.315 | 0.179 | 672 | 783 | 875 | 956 | 1027 | 1149 | 1253 | 726 | 828 | 915 | 991 | 1059 | 1177 | 1277 |
| 1.5 | 1.900 | 0.200 | 967 | 1131 | 1267 | 1385 | 1490 | 1671 | 1825 | 1056 | 1205 | 1332 | 1444 | 1543 | 1716 | 1864 |
| 2.0 | 2.375 | 0.218 | 1212 | 1420 | 1595 | 1745 | 1879 | 2110 | 2306 | 1334 | 1523 | 1684 | 1825 | 1952 | 2172 | 2360 |
| 2.5 | 2.875 | 0.276 | 1580 | 1855 | 2086 | 2285 | 2462 | 2768 | 3029 | 1751 | 1999 | 2211 | 2397 | 2564 | 2855 | 3104 |
| 3.0 | 3.500 | 0.300 | 1930 | 2273 | 2559 | 2807 | 3028 | 3408 | 3733 | 2157 | 2463 | 2725 | 2955 | 3161 | 3522 | 3832 |
| 3.5 | 4.000 | 0.318 | 2210 | 2608 | 2940 | 3228 | 3483 | 3925 | 4303 | 2484 | 2838 | 3140 | 3406 | 3645 | 4062 | 4422 |
| 4.0 | 4.500 | 0.337 | 2499 | 2954 | 3334 | 3663 | 3955 | 4460 | 4893 | 2824 | 3226 | 3570 | 3873 | 4146 | 4622 | 5034 |
| 5.0 | 5.563 | 0.375 | 3104 | 3683 | 4165 | 4583 | 4954 | 5598 | 6150 | 3544 | 4050 | 4484 | 4867 | 5212 | 5816 | 6340 |
| 6.0 | 6.625 | 0.432 | 3801 | 4525 | 5127 | 5649 | 6113 | 6919 | 7610 | 4379 | 5007 | 5546 | 6022 | 6450 | 7204 | 7859 |
| 8.0 | 8.625 | 0.500 | 4927 | 5898 | 6706 | 7407 | 8031 | 9118 | 10056 | 5761 | 6592 | 7308 | 7943 | 8516 | 9528 | 10413 |
| Size |  | Wall |  | $\begin{array}{r} \text { En } \\ \text { emper } \end{array}$ | ssivity ure Ri | 0.50,W <br> Above | ith Sun $40^{\circ} \mathrm{CAm}$ | bient |  |  | Emis | $\begin{aligned} & \text { ivity }=1 \\ & \text { ure } \mathrm{Ri} \end{aligned}$ | .50, Wi Abov | $\begin{aligned} & \text { hout S } \\ & 40^{\circ} \mathrm{C} \end{aligned}$ | mbient |  |
| $\text { SPS }^{\mathbf{a}}$ | OD | Thickness | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| (in) | (in) | (in) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 1.0 | 1.315 | 0.179 | 650 | 785 | 896 | 992 | 1078 | 1226 | 1353 | 780 | 893 | 989 | 1075 | 1152 | 1289 | 1408 |
| 1.5 | 1.900 | 0.200 | 930 | 1134 | 1302 | 1446 | 1575 | 1798 | 1990 | 1146 | 1312 | 1455 | 1582 | 1697 | 1901 | 2079 |
| 2.0 | 2.375 | 0.218 | 1161 | 1425 | 1642 | 1829 | 1994 | 2282 | 2531 | 1457 | 1669 | 1851 | 2014 | 2161 | 2422 | 2652 |
| 2.5 | 2.875 | 0.276 | 1507 | 1862 | 2152 | 2402 | 2624 | 3009 | 3343 | 1923 | 2203 | 2445 | 2661 | 2856 | 3204 | 3512 |
| 3.0 | 3.500 | 0.300 | 1833 | 2282 | 2647 | 2961 | 3240 | 3725 | 4146 | 2382 | 2731 | 3032 | 3301 | 3545 | 3981 | 4366 |
| 3.5 | 4.000 | 0.318 | 2092 | 2619 | 3046 | 3413 | 3739 | 4307 | 4799 | 2756 | 3160 | 3510 | 3822 | 4106 | 4613 | 5063 |
| 4.0 | 4.500 | 0.337 | 2358 | 2967 | 3459 | 3882 | 4257 | 4911 | 5479 | 3144 | 3606 | 4006 | 4364 | 4690 | 5272 | 5789 |
| 5.0 | 5.563 | 0.375 | 2912 | 3700 | 4335 | 4879 | 5362 | 6204 | 6937 | 3974 | 4560 | 5069 | 5525 | 5941 | 6687 | 7352 |
| 6.0 | 8.625 | 0.432 | 3547 | 4548 | 5350 | 6037 | 6647 | 7711 | 8638 | 4940 | 5672 | 6309 | 6880 | 7401 | 8339 | 9178 |
| 8.0 | 8.625 | 0.500 | 4556 | 5931 | 7028 | 7965 | 8797 | 10252 | 11526 | 6561 | 7541 | 8396 | 9166 | 9871 | 11145 | 12293 |



Table B.5-Single aluminum angle bus AC ampacity (55.0\% conductivity)

|  | Emissivity $=0.20$, With SunTemperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=\mathbf{0 . 2 0}$, Without Sun <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Size (in) | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 3.250 by 3.250 by 0.250 | 1588 | 1857 | 2083 | 2279 | 2454 | 2757 | 3016 | 1734 | 1980 | 2191 | 2376 | 2542 | 2831 | 3081 |
| 4.000 by 4.000 by 0.250 | 1835 | 2153 | 2420 | 2652 | 2859 | 3217 | 3525 | 2022 | 2311 | 2557 | 2775 | 2970 | 3312 | 3608 |
| 4.000 by 4.000 by 0.375 | 2178 | 2557 | 2875 | 3153 | 3400 | 3831 | 4201 | 2401 | 2744 | 3039 | 3299 | 3533 | 3943 | 4300 |
| 4.500 by 4.500 by 0.375 | 2343 | 2757 | 3104 | 3408 | 3678 | 4150 | 4558 | 2597 | 2970 | 3291 | 3574 | 3829 | 4279 | 4670 |
| 5.000 by 5.000 by 0.375 | 2518 | 2969 | 3347 | 3677 | 3972 | 4488 | 4934 | 2806 | 3210 | 3557 | 3865 | 4143 | 4633 | 5061 |

Emissivity=0.50, WithSun
Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient

Emissivity $=0.50$, Without Sun Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient

| Size (in) | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 3.250 by 3.250 by 0.250 | 1550 | 1889 | 2169 | 2412 | 2628 | 3007 | 3336 | 1902 | 2180 | 2420 | 2634 | 2828 | 3174 | 3481 |
| 4.000 by 4.000 by 0.250 | 1786 | 2194 | 2530 | 2821 | 3080 | 3535 | 3931 | 2236 | 2564 | 2848 | 3102 | 3334 | 3747 | 4114 |
| 4.000 by 4.000 by 0.375 | 2120 | 2606 | 3007 | 3354 | 3664 | 4208 | 4685 | 2654 | 3045 | 3385 | 3688 | 3965 | 4461 | 4904 |
| 4.500 by 4.500 by 0.375 | 2277 | 2813 | 3254 | 3637 | 3979 | 4580 | 5108 | 2885 | 3312 | 3683 | 4016 | 4320 | 4866 | 5356 |
| 5.000 by 5.000 by 0.375 | 2443 | 3032 | 3516 | 3936 | 4311 | 4973 | 5555 | 3130 | 3595 | 4000 | 4363 | 4696 | 5295 | 5833 |

Table B.6-Double aluminum angle bus AC ampacity

|  | Emissivity $=\mathbf{0 . 2 0}$, With Sun <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.20$, Without Sun <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Size (in) | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 3.250 by 3.250 by 0.250 | 2875 | 3370 | 3794 | 4166 | 4501 | 5086 | 5590 | 3045 | 3513 | 3917 | 4276 | 4600 | 5170 | 5663 |
| 4.000 by 4.000 by 0.250 | 3361 | 3949 | 4451 | 4892 | 5289 | 5984 | 6583 | 3579 | 4131 | 4608 | 5032 | 5415 | 6090 | 6675 |
| 4.000 by 4.000 by 0.375 | 3952 | 4646 | 5240 | 5764 | 6236 | 7065 | 7784 | 4208 | 4860 | 5426 | 5929 | 6385 | 7191 | 7893 |
| 4.500 by 4.500 by 0.375 | 4340 | 5109 | 5766 | 6346 | 6868 | 7786 | 8581 | 4636 | 5356 | 5980 | 6536 | 7040 | 7930 | 8707 |
| 5.000 by 5.000 by 0.375 | 4739 | 5585 | 6307 | 6945 | 7519 | 8528 | 9403 | 5077 | 5866 | 6552 | 7162 | 7715 | 8693 | 9546 |
|  | Emissivity $=0.50$, With Sun <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.50$, Without Sun <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| Size (in) | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 3.250 by 3.250 by 0.250 | 2832 | 3407 | 3893 | 4318 | 4700 | 5370 | 5953 | 3247 | 3749 | 4187 | 4578 | 4933 | 5566 | 6122 |
| 4.000 by 4.000 by 0.250 | 3306 | 3996 | 4577 | 5086 | 5542 | 6345 | 7044 | 3835 | 4432 | 4952 | 5416 | 5839 | 6593 | 7258 |
| 4.000 by 4.000 by 0.375 | 3887 | 4702 | 5389 | 5992 | 6535 | 7492 | 8329 | 4510 | 5215 | 5830 | 6382 | 6885 | 7785 | 8582 |
| 4.500 by 4.500 by 0.375 | 4265 | 5173 | 5938 | 6609 | 7213 | 8277 | 9209 | 4983 | 5764 | 6446 | 7057 | 7615 | 8614 | 9499 |
| 5.000 by 5.000 by 0.375 | 4653 | 5658 | 6503 | 7245 | 7911 | 9087 | 10117 | 5472 | 6331 | 7081 | 7755 | 8369 | 9470 | 10447 |

Table B.7-Aluminum integral web channel bus AC ampacity

| Size | Emissivity $=0.20$, With Sun <br> Temperature Rise Above $40{ }^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.20$, Without Sun Temperature Rise Above $40{ }^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| (in) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 4.000 by 4.000 by 0.250 | 2395 | 2948 | 3404 | 3799 | 4150 | 4761 | 5286 | 2949 | 3402 | 3795 | 4144 | 4461 | 5022 | 551 |
| 4.000 by 4.000 by 0.312 | 2603 | 3206 | 3703 | 4134 | 4517 | 5183 | 5757 | 3213 | 3706 | 4133 | 4514 | 4860 | 5472 | 600 |
| 6.000 by 4.000 by 0.375 | 3391 | 4168 | 4812 | 5370 | 5867 | 6734 | 7483 | 4161 | 4800 | 5356 | 5851 | 6301 | 7099 | 779 |
| 6.000 by 5.000 by 0.375 | 3558 | 4420 | 5129 | 5743 | 6289 | 7241 | 8062 | 4483 | 5175 | 5778 | 6316 | 6805 | 7674 | 843 |
| 6.000 by 6.000 by 0.550 | 4287 | 5335 | 6200 | 6950 | 7621 | 8795 | 9816 | 5412 | 6254 | 6990 | 7649 | 8250 | 9325 | 1027 |
| 8.000 by 5.000 by 0.500 | 4617 | 5695 | 6588 | 7365 | 8058 | 9272 | 10326 | 5699 | 6582 | 7351 | 8039 | 8666 | 9783 | 1077 |
| 8.000 by 8.000 by 0.500 | 5849 | 7228 | 8375 | 9374 | 10271 | 11846 | 13223 | 7212 | 8345 | 9335 | 10224 | 11036 | 12491 | 1378 |
| 12.000 by 12.000 by 0.625 | 8610 | 10614 | 12296 | 13774 | 15108 | 17477 | 19574 | 10466 | 12138 | 13608 | 14936 | 16156 | 18361 | 2034 |


| Size | Emissivity $=0.50$, With Sun <br> Temperature Rise Above $40{ }^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.50$, Without Sun Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| (in) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 4.000 by 4.000 by 0.250 | 2463 | 3102 | 3627 | 4081 | 4486 | 5198 | 5819 | 3208 | 3707 | 4143 | 4535 | 4894 | 5538 | 611 |
| 4.000 by 4.000 by 0.312 | 2677 | 3376 | 3948 | 4444 | 4887 | 5665 | 6345 | 3497 | 4041 | 4517 | 4944 | 5336 | 6039 | 666 |
| 6.000 by 4.000 by 0.375 | 3572 | 4470 | 5211 | 5856 | 6434 | 7453 | 8349 | 4568 | 5280 | 5905 | 6467 | 6982 | 7911 | 874 |
| 6.000 by 5.000 by 0.375 | 3718 | 4722 | 5544 | 6256 | 6893 | 8014 | 8999 | 4929 | 5701 | 6379 | 6990 | 7551 | 8563 | 947 |
| 6.000 by 6.000 by 0.550 | 4403 | 5646 | 6661 | 7540 | 8329 | 9722 | 10954 | 5963 | 6904 | 7733 | 8483 | 9174 | 10428 | 1156 |
| 8.000 by 5.000 by 0.500 | 4886 | 6145 | 7185 | 8091 | 8906 | 10347 | 11622 | 6308 | 7300 | 8172 | 8960 | 9685 | 10999 | 1218 |
| 8.000 by 8.000 by 0.500 | 5922 | 7594 | 8963 | 10152 | 11219 | 13110 | 14786 | 7990 | 9262 | 10384 | 11400 | 12338 | 14044 | 1559 |
| 12.000 by 12.000 by 0.625 | 8584 | 11093 | 13153 | 14949 | 16570 | 19463 | 22058 | 11724 | 13621 | 15304 | 16839 | 18264 | 20878 | 2327 |

Table B.8-Single copper rectangular bar AC ampacity, with sun (99.0\% conductivity)

| Size (in) | Emissivity $=0.35$Temperature Rise Above $40^{\circ}$ C Ambient |  |  |  |  |  |  | Emissivity $=\mathbf{0 . 8 5}$Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 0.250 by 4.000 | 1516 | 1751 | 1951 | 2127 | 2286 | 2564 | 2806 | 1661 | 1948 | 2194 | 2412 | 2611 | 2965 | 3281 |
| 0.250 by 5.000 | 1764 | 2040 | 2276 | 2484 | 2671 | 3002 | 3291 | 1955 | 2296 | 2589 | 2850 | 3088 | 3515 | 3898 |
| 0.250 by 6.000 | 2010 | 2327 | 2599 | 2838 | 3054 | 3437 | 3773 | 2250 | 2646 | 2987 | 3290 | 3568 | 4067 | 4517 |
| 0.375 by 4.000 | 1824 | 2112 | 2356 | 2572 | 2766 | 3107 | 3405 | 1985 | 2337 | 2638 | 2906 | 3149 | 3584 | 3973 |
| 0.375 by 5.000 | 2122 | 2458 | 2746 | 3000 | 3229 | 3633 | 3988 | 2337 | 2754 | 3112 | 3430 | 3721 | 4243 | 4712 |
| 0.375 by 6.000 | 2407 | 2792 | 3121 | 3412 | 3675 | 4141 | 4552 | 2679 | 3159 | 3573 | 3942 | 4279 | 4887 | 5436 |
| 0.375 by 8.000 | 2934 | 3409 | 3816 | 4178 | 4505 | 5089 | 5608 | 3319 | 3922 | 4442 | 4908 | 5335 | 6109 | 6813 |
| 0.500 by 4.000 | 2083 | 2415 | 2699 | 2948 | 3173 | 3569 | 3915 | 2253 | 2662 | 3011 | 3321 | 3603 | 4108 | 4560 |
| 0.500 by 5.000 | 2404 | 2790 | 3120 | 3412 | 3675 | 4141 | 4551 | 2633 | 3113 | 3524 | 3890 | 4224 | 4826 | 5367 |
| 0.500 by 6.000 | 2717 | 3156 | 3532 | 3865 | 4166 | 4701 | 5174 | 3007 | 3558 | 4031 | 4453 | 4839 | 5536 | 6167 |
| 0.500 by 8.000 | 3312 | 3853 | 4317 | 4730 | 5105 | 5774 | 6369 | 3729 | 4417 | 5011 | 5542 | 6030 | 6916 | 7723 |
| 0.625 by 4.000 | 2253 | 2617 | 2928 | 3203 | 3451 | 3889 | 4274 | 2423 | 2873 | 3258 | 3599 | 3911 | 4469 | 4971 |
| 0.625 by 5.000 | 2619 | 3045 | 3409 | 3731 | 4023 | 4540 | 4996 | 2854 | 3384 | 3840 | 4245 | 4615 | 5282 | 5885 |
| 0.625 by 6.000 | 2951 | 3433 | 3847 | 4213 | 4546 | 5137 | 5662 | 3251 | 3857 | 4378 | 4843 | 5269 | 6040 | 6739 |
| 0.625 by 8.000 | 3598 | 4192 | 4702 | 5156 | 5568 | 6306 | 6966 | 4034 | 4791 | 5443 | 6028 | 6565 | 7541 | 8433 |
| 0.625 by 10.000 | 4179 | 4875 | 5474 | 6009 | 6496 | 7372 | 8158 | 4752 | 5648 | 6424 | 7121 | 7763 | 8936 | 10012 |
| 0.625 by 12.000 | 4758 | 5555 | 6244 | 6860 | 7422 | 8435 | 9348 | 5474 | 6511 | 7411 | 8222 | 8970 | 10339 | 11601 |
| 0.750 by 4.000 | 2455 | 2857 | 3199 | 3502 | 3775 | 4258 | 4683 | 2626 | 3125 | 3550 | 3928 | 4271 | 4888 | 5443 |
| 0.750 by 5.000 | 2834 | 3300 | 3699 | 4051 | 4370 | 4937 | 5438 | 3073 | 3656 | 4155 | 4599 | 5005 | 5737 | 6398 |
| 0.750 by 6.000 | 3204 | 3732 | 4185 | 4587 | 4951 | 5600 | 6177 | 3513 | 4179 | 4752 | 5262 | 5729 | 6575 | 7343 |
| 0.750 by 8.000 | 3881 | 4527 | 5082 | 5576 | 6026 | 6831 | 7551 | 4334 | 5159 | 5870 | 6507 | 7092 | 8157 | 9130 |
| 0.750 by 10.000 | 4509 | 5265 | 5917 | 6498 | 7029 | 7982 | 8840 | 5109 | 6085 | 6929 | 7687 | 8386 | 9662 | 10835 |
| 0.750 by 12.000 | 5119 | 5983 | 6729 | 7396 | 8006 | 9107 | 10100 | 5869 | 6995 | 7971 | 8850 | 9661 | 11147 | 12519 |

Table B.9-Single copper rectangular bar AC ampacity, without sun (99.0\% conductivity)

| Size (in) | Emissivity $=0.35$ <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=\mathbf{0 . 8 5}$ <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 0.250 by 4.000 | 1577 | 1802 | 1996 | 2168 | 2322 | 2595 | 2833 | 1793 | 2059 | 2290 | 2498 | 2688 | 3030 | 3337 |
| 0.250 by 5.000 | 1838 | 2102 | 2330 | 2533 | 2715 | 3039 | 3323 | 2114 | 2429 | 2705 | 2953 | 3180 | 3592 | 3965 |
| 0.250 by 6.000 | 2098 | 2401 | 2663 | 2896 | 3107 | 3481 | 3811 | 2436 | 2801 | 3121 | 3410 | 3675 | 4157 | 4595 |
| 0.375 by 4.000 | 1908 | 2182 | 2418 | 2627 | 2816 | 3150 | 3442 | 2167 | 2489 | 2770 | 3023 | 3254 | 3672 | 4049 |
| 0.375 by 5.000 | 2221 | 2542 | 2819 | 3065 | 3288 | 3683 | 4032 | 2550 | 2932 | 3266 | 3568 | 3844 | 4347 | 4802 |
| 0.375 by 6.000 | 2522 | 2889 | 3206 | 3488 | 3744 | 4199 | 4603 | 2924 | 3364 | 3751 | 4099 | 4421 | 5006 | 5539 |
| 0.375 by 8.000 | 3081 | 3532 | 3924 | 4274 | 4593 | 5164 | 5673 | 3628 | 4179 | 4665 | 5105 | 5513 | 6258 | 6941 |
| 0.500 by 4.000 | 2189 | 2505 | 2777 | 3018 | 3236 | 3623 | 3962 | 2485 | 2855 | 3179 | 3470 | 3737 | 4221 | 4658 |
| 0.500 by 5.000 | 2527 | 2894 | 3211 | 3493 | 3749 | 4204 | 4606 | 2900 | 3335 | 3717 | 4062 | 4379 | 4956 | 5480 |
| 0.500 by 6.000 | 2858 | 3275 | 3636 | 3958 | 4251 | 4773 | 5237 | 3310 | 3810 | 4250 | 4647 | 5014 | 5683 | 6294 |
| 0.500 by 8.000 | 3489 | 4002 | 4448 | 4847 | 5211 | 5863 | 6447 | 4103 | 4729 | 5281 | 5782 | 6246 | 7097 | 7879 |
| 0.625 by 4.000 | 2379 | 2724 | 3021 | 3286 | 3526 | 3953 | 4330 | 2700 | 3104 | 3458 | 3778 | 4071 | 4604 | 5088 |
| 0.625 by 5.000 | 2765 | 3168 | 3517 | 3828 | 4111 | 4615 | 5062 | 3171 | 3650 | 4069 | 4449 | 4799 | 5437 | 6019 |
| 0.625 by 6.000 | 3117 | 3573 | 3969 | 4323 | 4645 | 5222 | 5736 | 3607 | 4154 | 4636 | 5072 | 5475 | 6213 | 6889 |
| 0.625 by 8.000 | 3804 | 4365 | 4854 | 5291 | 5691 | 6411 | 7057 | 4469 | 5153 | 5757 | 6307 | 6816 | 7752 | 8615 |
| 0.625 by 10.000 | 4423 | 5081 | 5654 | 6169 | 6642 | 7496 | 8266 | 5262 | 6073 | 6792 | 7448 | 8057 | 9182 | 10225 |
| 0.625 by 12.000 | 5042 | 5795 | 6454 | 7046 | 7591 | 8578 | 9473 | 6060 | 7000 | 7835 | 8597 | 9308 | 10622 | 11845 |
| 0.750 by 4.000 | 2605 | 2983 | 3310 | 3601 | 3865 | 4335 | 4750 | 2956 | 3400 | 3789 | 4139 | 4462 | 5049 | 5582 |
| 0.750 by 5.000 | 3006 | 3445 | 3825 | 4164 | 4473 | 5024 | 5515 | 3446 | 3967 | 4425 | 4839 | 5221 | 5918 | 6555 |
| 0.750 by 6.000 | 3397 | 3895 | 4328 | 4715 | 5067 | 5699 | 6263 | 3929 | 4526 | 5052 | 5529 | 5970 | 6778 | 7518 |
| 0.750 by 8.000 | 4117 | 4726 | 5256 | 5732 | 6167 | 6951 | 7655 | 4834 | 5575 | 6231 | 6827 | 7381 | 8399 | 9340 |
| 0.750 by 10.000 | 4787 | 5499 | 6122 | 6681 | 7195 | 8123 | 8963 | 5690 | 6569 | 7348 | 8059 | 8721 | 9944 | 11078 |
| 0.750 by 12.000 | 5439 | 6253 | 6965 | 7607 | 8198 | 9269 | 10242 | 6532 | 7547 | 8449 | 9273 | 10043 | 11468 | 12795 |

Table B.10-Copper tubular bus-Schedule 40 AC ampacity

| Size |  | Wall | Emissivity $=0.35$, With SunTemperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.35$, Without SunTemperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| SPS ${ }^{\text {a }}$ | OD | Thickness | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| (in) | (in) | (in) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 1.0 | 1.315 | 0.127 | 771 | 912 | 1029 | 1131 | 1220 | 1375 | 1506 | 878 | 1002 | 1107 | 1200 | 1283 | 1428 | 1552 |
| 1.5 | 1.900 | 0.150 | 1131 | 1347 | 1526 | 1681 | 1818 | 2054 | 2255 | 1313 | 1499 | 1658 | 1798 | 1923 | 2143 | 2332 |
| 2.0 | 2.375 | 0.157 | 1383 | 1656 | 1881 | 2075 | 2246 | 2543 | 2796 | 1628 | 1859 | 2056 | 2231 | 2387 | 2661 | 2899 |
| 2.5 | 2.875 | 0.188 | 1755 | 2111 | 2403 | 2655 | 2878 | 3264 | 3594 | 2091 | 2389 | 2644 | 2868 | 3071 | 3426 | 3734 |
| 3.0 | 3.500 | 0.219 | 2214 | 2675 | 3054 | 3380 | 3669 | 4169 | 4597 | 2673 | 3054 | 3381 | 3670 | 3930 | 4388 | 4787 |
| 4.0 | 4.500 | 0.250 | 2870 | 3492 | 4002 | 4441 | 4829 | 5502 | 6080 | 3530 | 4035 | 4470 | 4855 | 5202 | 5815 | 6350 |
| 6.0 | 6.625 | 0.250 | 3903 | 4807 | 5544 | 6177 | 6737 | 7708 | 8545 | 4955 | 5669 | 6285 | 6831 | 7324 | 8199 | 8968 |
| 8.0 | 8.625 | 0.313 | 5281 | 6570 | 7617 | 8514 | 9308 | 10687 | 11880 | 6871 | 7868 | 8728 | 9493 | 10187 | 11422 | 12512 |
| Size |  | Wall | Emissivity $=0.85$, Without Sun Temperature Rise Above $40{ }^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.85$, With SunTemperature Rise Above $40{ }^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| $\text { SPS }^{\mathbf{a}}$ | OD | Thickness | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| (in) | (in) | (in) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 1.0 | 1.315 | 0.127 | 726 | 916 | 1069 | 1199 | 1315 | 1515 | 1688 | 978 | 1120 | 1243 | 1353 | 1452 | 1629 | 1786 |
| 1.5 | 1.900 | 0.150 | 1054 | 1354 | 1593 | 1797 | 1977 | 2289 | 2559 | 1482 | 1698 | 1886 | 2054 | 2206 | 2479 | 2722 |
| 2.0 | 2.375 | 0.157 | 1279 | 1665 | 1970 | 2230 | 2458 | 2855 | 3200 | 1851 | 2122 | 2358 | 2569 | 2762 | 3106 | 3414 |
| 2.5 | 2.875 | 0.188 | 1611 | 2123 | 2526 | 2867 | 3168 | 3689 | 4142 | 2394 | 2747 | 3054 | 3328 | 3579 | 4030 | 4433 |
| 3.0 | 3.500 | 0.219 | 2014 | 2692 | 3221 | 3668 | 4062 | 4745 | 5339 | 3083 | 3539 | 3936 | 4292 | 4618 | 5204 | 5729 |
| 4.0 | 4.500 | 0.250 | 2579 | 3517 | 4242 | 4852 | 5389 | 6321 | 7131 | 4112 | 4723 | 5256 | 5736 | 6175 | 6968 | 7682 |
| 6.0 | 6.625 | 0.250 | 3425 | 4848 | 5925 | 6827 | 7617 | 8988 | 10182 | 5863 | 6741 | 7509 | 8201 | 8836 | 9988 | 11031 |
| 8.0 | 8.625 | 0.313 | 4543 | 6632 | 8190 | 9488 | 10624 | 12596 | 14315 | 8220 | 9459 | 10545 | 11527 | 12431 | 14075 | 15569 |

${ }^{\text {a }}$ SPS $=$ Standard pipe size
Table B.11-Copper tubular bus-Schedule 80 AC ampacity

| Size |  | Wall | Emissivity = 0.35, With Sun <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity = 0.35, Without Sun Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\text { SPS }^{\mathbf{a}}$ | OD | Thickness | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| (in) | (in) | (in) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 1.0 | 1.315 | 0.182 | 903 | 1069 | 1206 | 1325 | 1430 | 1611 | 1765 | 1029 | 1174 | 1297 | 1406 | 1503 | 1673 | 1818 |
| 1.5 | 1.900 | 0.203 | 1289 | 1536 | 1741 | 1917 | 2073 | 2343 | 2573 | 1498 | 1710 | 1891 | 2051 | 2194 | 2445 | 2661 |
| 2.0 | 2.375 | 0.221 | 1610 | 1928 | 2190 | 2416 | 2616 | 2962 | 3258 | 1895 | 2164 | 2395 | 2598 | 2780 | 3100 | 3377 |
| 2.5 | 2.875 | 0.280 | 2093 | 2517 | 2866 | 3168 | 3434 | 3896 | 4292 | 2493 | 2848 | 3153 | 3422 | 3664 | 4089 | 4459 |
| 3.0 | 3.500 | 0.304 | 2536 | 3065 | 3501 | 3876 | 4209 | 4785 | 5279 | 3062 | 3500 | 3876 | 4209 | 4508 | 5036 | 5497 |
| 4.0 | 4.500 | 0.341 | 3256 | 3963 | 4543 | 5043 | 5486 | 6255 | 6917 | 4004 | 4580 | 5075 | 5513 | 5910 | 6610 | 7224 |
| 6.0 | 6.625 | 0.437 | 4789 | 5906 | 6820 | 7606 | 8306 | 9525 | 10584 | 6081 | 6965 | 7730 | 8411 | 9030 | 10132 | 11108 |
| 8.0 | 8.625 | 0.500 | 6076 | 7571 | 8790 | 9841 | 10776 | 12412 | 13842 | 7906 | 9066 | 10073 | 10973 | 11794 | 13265 | 14579 |
| Size |  | Wall |  | $\begin{array}{r} \mathbf{E} \\ \text { emper } \end{array}$ | issivity re Ris | $=0.85,$ <br> Above | With Su $40^{\circ} \mathrm{C}$ An | ient |  |  | $\begin{array}{r} \text { En } \\ \text { Temper } \end{array}$ | ssivity $=$ ture Ri | $0.85, \mathrm{~W}$ <br> Abov | ithout S $40{ }^{\circ} \mathrm{C}$ A | un <br> mbient |  |
| $\text { SPS }^{\mathbf{a}}$ | OD | Thickness | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| (in) | (in) | (in) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| 1.0 | 1.315 | 0.182 | 851 | 1073 | 1252 | 1405 | 1540 | 1775 | 1978 | 1146 | 1313 | 1457 | 1585 | 1701 | 1909 | 2092 |
| 1.5 | 1.900 | 0.203 | 1202 | 1544 | 1817 | 2050 | 2255 | 2612 | 2920 | 1690 | 1937 | 2151 | 2343 | 2517 | 2829 | 3106 |
| 2.0 | 2.375 | 0.221 | 1489 | 1938 | 2294 | 2597 | 2863 | 3326 | 3728 | 2155 | 2471 | 2746 | 2992 | 3216 | 3619 | 3978 |
| 2.5 | 2.875 | 0.280 | 1921 | 2532 | 3013 | 3420 | 3780 | 4404 | 4946 | 2855 | 3276 | 3642 | 3971 | 4271 | 4810 | 5293 |
| 3.0 | 3.500 | 0.304 | 2308 | 3086 | 3693 | 4207 | 4659 | 5446 | 6130 | 3532 | 4056 | 4512 | 4922 | 5296 | 5972 | 6579 |
| 4.0 | 4.500 | 0.341 | 2926 | 3992 | 4816 | 5511 | 6122 | 7186 | 8113 | 4664 | 5360 | 5967 | 6513 | 7015 | 7921 | 8739 |
| 6.0 | 6.625 | 0.437 | 4203 | 5956 | 7288 | 8407 | 9391 | 11107 | 12612 | 7195 | 8282 | 9236 | 10099 | 10894 | 12343 | 13664 |
| 8.0 | 8.625 | 0.500 | 5277 | 7642 | 9452 | 10967 | 12300 | 14629 | 16679 | 9457 | 10899 | 12170 | 13324 | 14391 | 16346 | 18140 |

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Table B.12-Double copper channel bus AC ampacity

|  | Emissivity $=0.35$, With Sun Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity = 0.35, Without Sun Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Size (in) | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 3.000 by 1.313 by 0.216 | 2785 | 3347 | 3819 | 4232 | 4601 | 5246 | 5801 | 3178 | 3671 | 4098 | 4478 | 4822 | 5430 | 5961 |
| 4.000 by 1.750 by 0.240 | 3697 | 4470 | 5118 | 5684 | 6190 | 7075 | 7841 | 4283 | 4951 | 5531 | 6048 | 6517 | 7348 | 8076 |
| 4.000 by 1.750 by 0.338 | 4106 | 4969 | 5695 | 6331 | 6902 | 7906 | 8780 | 4757 | 5504 | 6155 | 6737 | 7267 | 8212 | 9044 |
| 5.000 by 2.188 by 0.338 | 4967 | 6040 | 6942 | 7731 | 8440 | 9686 | 10772 | 5827 | 6746 | 7548 | 8266 | 8920 | 10087 | 11117 |
| 6.000 by 2.688 by 0.384 | 5932 | 7235 | 8332 | 9293 | 10159 | 11686 | 13025 | 6995 | 8107 | 9079 | 9953 | 10751 | 12182 | 13453 |
|  | Emissivity $=0.85$, With Sun <br> Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  | Emissivity $=0.85$, Without Sun Temperature Rise Above $40^{\circ} \mathrm{C}$ Ambient |  |  |  |  |  |  |
| Size (in) | 30 | 40 | 50 | 60 | 70 | 90 | 110 | 30 | 40 | 50 | 60 | 70 | 90 | 110 |
| 3.000 by 1.313 by 0.216 | 2733 | 3430 | 4003 | 4499 | 4941 | 5718 | 6395 | 3504 | 4053 | 4533 | 4963 | 5356 | 6061 | 6689 |
| 4.000 by 1.750 by 0.240 | 3619 | 4593 | 5390 | 6078 | 6693 | 7772 | 8714 | 4764 | 5514 | 6171 | 6762 | 7303 | 8276 | 9145 |
| 4.000 by 1.750 by 0.338 | 4019 | 5106 | 5998 | 6771 | 7464 | 8685 | 9759 | 5290 | 6129 | 6867 | 7532 | 8143 | 9248 | 10241 |
| 5.000 by 2.188 by 0.338 | 4851 | 6222 | 7341 | 8310 | 9177 | 10706 | 12052 | 6526 | 7565 | 8480 | 9306 | 10065 | 11440 | 12680 |
| 6.000 by 2.688 by 0.384 | 5770 | 7460 | 8836 | 10029 | 11099 | 12990 | 14663 | 7888 | 9154 | 10271 | 11283 | 12217 | 13915 | 15455 |

## Annex C

(informative)

## Thermal considerations for outdoor bus-conductor design

By the Substation Committee of the IEEE Power Engineering Society. ${ }^{9}$

## C. 1 Abstract

Outdoor rigid bus design is based on several limiting criteria. This paper brings to a single source the thermal considerations of rigid bus design namely, transfer of heat and properties of material. It concerns itself with aluminum alloys, copper and copper alloys and the currently acceptable shapes. Historically thermal designs have been conservative. This paper will allow the engineer to re-examine the factors involved in increased current loadings of rigid bus and possibly determine new thermal limits.

## C. 2 Introduction

Thermal considerations entering into the design of bus conductors for outdoor substations fall into two general categories, transfer of heat and properties of materials. Each of these subjects will be considered in detail in this paper. The first, transfer of heat to and from the conductor, is relatively independent of the material and is mainly a function of the geometry of the conductor, proximity to other surfaces or conductors, atmospheric conditions, and geographic location. The most important element in the computation is the estimate of forced convection arising from wind currents. A method is given here to compute heat losses due to forced and natural convection and radiation and heat gained from the sun. Using the formulas provided it is possible to calculate the current carrying capacity of any conductor corresponding to a given temperature rise. Examples are provided showing methods for calculating the ampacity of conventional types of bus conductors, e.g., bar, tube, channel, angle, integral web, etc.

The second subject, properties of materials, includes the effects of temperature and outdoor exposure on the mechanical strength, electrical resistivity, dimensional stability, and surface condition of the conductor. Aluminum alloys, copper, and copper alloys are included in the discussion and tabulations. No attempt has been made to consider the relative merits of the conductors. Instead, technical information is provided which must be coupled with economic factors when optimizing design and selecting materials.

## C. 3 Heat transfer

Usually well over half the heat generated by resistance losses in a bus conductor is removed from the surface by convection of the surrounding air. The remainder is given off by radiation from external surfaces. Unfortunately, it is not at all convenient to run controlled outdoor tests to determine the appropriate heat transfer coefficients. As a result there is very little independent support for the formulas found in the literature.

A variety of formulas can be found for the sizes of conductors of interest. All show that convective heat transfer out-of-doors exceeds that in the indoors when it is assumed that the wind velocity is 2 feet per second (fps). However, the difference between the indoor and outdoor rating is often not very great. If a slower

[^6]wind velocity is assumed, the outdoor heat losses may be calculated as lower than those indoors. This is not plausible. It is therefore, concluded that assumption of a 2 fps wind is a conservative, yet realistic approach, and it will be used in the examples given herein.

The difference between indoor and outdoor convection losses are found to diminish with increasing conductor size and increasing temperature rise. This is because an increase in the temperature rise leads to natural drafts which can be as effective as a slight breeze in promoting heat transfer. Similarly, with large conductors, the assumed 2 fps wind speed is so low as to add very little benefit over natural convection.

For the purpose of calculating ampacity, conditions which are least advantageous for convection must be considered. Thus, it is assumed that there is only a 2 fps wind. (See note 1 following the references of this annex) It is to be expected that when the flow is at an angle or normal to the surface, heat transfer will increase. Likewise, it is wise to stipulate that the emissivity is a low value when there is no solar heating. This will provide the most conservative ampacity rating. In contrast, when there can be considerable solar heating a high value of emissivity essentially equal to solar absorptivity may give the most conservative ampacity rating.

In connection with this last point, it should be noted that solar heating of the conductor always diminishes ampacity and can result in outdoor current ratings which are lower than indoor ratings. This is less likely on smaller conductors for which forced (outdoor) convective heat transfer coefficients are relatively high. However, for large conductors with high absorptivity, the heat gain from solar radiation can exceed the improvement in convective heat transfer due to the wind effect and ratings are reduced accordingly.

## C.3.1 Assumptions

Some assumptions will be made about the properties of air in order to reduce the number of terms which must be carried through the computations. These approximations will have negligible effect on the accuracy of the calculated ampacity. First, it is assumed that the properties of air are constant and may be evaluated at mid-range temperatures. This is reasonable because variations in heat capacity, conductivity, density and viscosity of air tend to compensate for one another and have very little net effect on heat transfer over the temperature range of interest. For example, the Prandtl number of air, $\mathrm{C} \rho^{\mu / \mathrm{k}}$ is commonly taken as 0.74 over a wide range of ordinary temperatures and pressures.

The properties used are as follows:

```
\(\mathrm{C} \rho \quad\) heat capacity of air \(=0.235 \mathrm{btu} / \mathrm{lb} .-^{\circ} \mathrm{F}\)
\(\mathrm{k} \quad\) thermal conductivity of air \(=.018 \mathrm{btu} / \mathrm{hr}^{-\mathrm{ft}^{2}-{ }^{\circ} \mathrm{F}, ~}\)
\(\mathrm{C} \rho^{\mu / \mathrm{k}}\) Prandtl number of air \(=0.74\)
\(\rho^{\mathrm{a}} \quad\) density of air \(=0.062 \mathrm{lbs} / \mathrm{cu} \mathrm{Ft}\).
\(\mu / \rho \mathrm{a}\) kinematic viscosity \(=0.9 \mathrm{ft}^{2} / \mathrm{sec}\)
```

As a result, only the temperature difference between the conductor and the surrounding air is important in calculating convective heat losses. For example, the convection losses calculated for a $40^{\circ} \mathrm{C}$ temperature rise apply equally for a $70^{\circ} \mathrm{C}$ conductor in $30^{\circ} \mathrm{C}$ air or an $85^{\circ} \mathrm{C}$ conductor in $45^{\circ} \mathrm{C}$ air.

One might expect that the ampacities in the above instances would be different because the resistivities at $70^{\circ} \mathrm{C}$ and $85^{\circ} \mathrm{C}$ are different. However, it will be seen that the radiation losses which increase with the absolute temperature rather than the temperature difference tend to offset the rise in resistivities. As a result, ampacities based on the $40^{\circ} \mathrm{C}$ ambient apply quite well to ambients from about $20^{\circ} \mathrm{C}$ to $50^{\circ} \mathrm{C}$. Thus, for any temperature rise there is a single ampacity, (irrespective of the ambient) and it is usually not necessary to calculate a different ampacity for each ambient temperature and temperature rise.

## C.3.2 Computation method

The general approach suggested for calculating the ampacity of any outdoor bus conductor is summarized below. A detailed explanation of each item follows.

Step by step the procedure is as follows:

1) Identify all exterior surfaces which should be treated as flat planes subject to forced convection.
2) Identify any exterior surfaces which should be treated as cylindrical surfaces subject to forced convection.
3) Identify any surfaces which may be shielded from the wind and only lose heat via natural convection (the same as indoors).
4) Identify surfaces which will lose heat also by radiation.
5) Ascertain the orientation and location of the conductors in determining the projected area exposed to solar heat gain.
6) For each of the appropriate areas (items 1,2 and 3 ) compute the total convective heat losses, $q_{\mathrm{c}}$.
7) For the appropriate values of emittance and area (item 4) compute the total heat lost through radiation, $q_{\mathrm{r}}$.
8) Consider the projected area, latitude, altitude, seasonal factors, absorptivity, etc. and compute the solar heat gain, $q_{\mathrm{s}}$.
9) Sum the heat gain and loss terms and, for the appropriately temperature compensated values of resistance $(R)$ and skin effect coefficient $(F)$, compute ampacity using the general formula

$$
I=\sqrt{\frac{q_{\mathrm{c}}+q_{\mathrm{r}}-q_{\mathrm{s}}}{R F}}
$$

where

$$
\begin{array}{ll}
I & =\text { current for the allowable temperature rise, amps. } \\
q_{\mathrm{c}} & =\text { convective heat loss, watts/ft. } \\
q_{\mathrm{r}} & =\text { radiation loss, watts/ft. } \\
q_{\mathrm{s}} & =\text { solar heat gain, watts/ft. } \\
R & =\text { direct current resistance at the operating temperature, ohms/ft. } \\
F & \text { skin effect coefficient for } 60 \text { cycle current. }
\end{array}
$$

The following is an analysis of each of the individual operations. It will show that the basic equations can be reduced to easy to handle forms.

## C.3.2 1. Forced convection over flat surfaces

When air flows parallel to and over a flat planar surface the following equation may be used to calculate the heat transfer coefficient:

$$
h=0.66(L v \rho a / \mu)^{-1 / 2}(C \rho \mu / k)^{-2 / 3}(C \rho v \rho a)
$$

where

$$
\begin{array}{ll}
h & =\text { heat transfer coefficient, } \mathrm{btu} / \mathrm{hr}^{\circ} \mathrm{F} \mathrm{ft}^{2} \\
L & =\text { length of flow path over conductor (normally the width or thickness), in feet } \\
v & =\text { air velocity, feet/hour }
\end{array}
$$

The total heat lost (in watts/ft) from the surface due to forced convection is

$$
q_{\mathrm{c}}=0.00367 h A \Delta T
$$

where

```
qc = convection losses, watts/ft.
h = heat transfer coefficient BTU/hr ' F ft }\mp@subsup{}{}{2
A = area of flat surfaces, square inches/linear foot
\DeltaT = temperature differences between the surface of the conductor and surrounding air, }\mp@subsup{}{}{\circ}\textrm{C
```

At elevations above sea level multiply $q_{\mathrm{c}}$ by $P^{0.5}$ where $P$ is the air pressure in atmospheres. This will reduce the convective coefficient for lower pressures.

For the properties of air noted earlier,

$$
q_{\mathrm{c}}=0.0085 \sqrt{\frac{V}{l}} A \Delta T
$$

where

$$
\begin{array}{ll}
v & =\text { air velocity, feet/sec. } \\
l & =\text { length of surface over which air flows, inches }(=12 \mathrm{~L}) \\
\text { For } v & =2 \text { feet per second } \\
q_{\mathrm{c}}= & 0.012 A \sqrt{\frac{1}{l}} \Delta T
\end{array}
$$

This simplified formula applies to air flow parallel to the surface. Outdoors air flow is seldom unidirectional and cannot always be parallel to the surface. However, it is assumed that air circulating around the conductor will be in more turbulent flow and provide on the average greater heat transfer than would be calculated using the above equation.

The convective loss formula above must be applied to each flat surface of the conductor. For example, consider a rectangular conductor $6^{\prime \prime} \mathrm{x} 1 / 2^{\prime \prime}$ operating at $100^{\circ} \mathrm{C}$ in a $40^{\circ} \mathrm{C}$ ambient. For the 6 -inch faces $\mathrm{A}=2 \times 6 \times 12=144 \mathrm{in}^{2} / \mathrm{ft}$. Then

$$
q_{\mathrm{c} 6}=\frac{(0.0120)(144)(60)}{6^{1 / 2}}
$$

or

$$
q_{\mathrm{c} 6}=42.3 \mathrm{watts} / \mathrm{ft} .
$$

For the $1 / 2$-inch edges, $\mathrm{A}=2 \times(1 / 2) \times 12=12 \mathrm{in}^{2} / \mathrm{ft}$ and $\sqrt{(1 / 2)}=.707$.

Then

$$
\begin{aligned}
& q_{\mathrm{c}}(1 / 2)=12.2 \text { watts } / \mathrm{ft} \\
& q_{\mathrm{c}}=q_{\mathrm{c}}(1 / 2)+q_{\mathrm{c} 6}=54.5 \text { watts } / \mathrm{ft}
\end{aligned}
$$

Note that for a 6-inch square tube the convective heat loss would have been twice $q_{\mathrm{c} 6}$ calculated above or 84 watts/ft. The heat loss per unit area, $q_{\mathrm{c}} / A$, is $84 / 288$ or 0.29 watts $/ \mathrm{in}^{2}$. It will be interesting to compare this value with that calculated for a 6 -inch cylindrical pipe by a different method in the next section.

## C.3.2 2. Forced convection over cylindrical surfaces

From McAdams [2] ${ }^{10}$ text or Perry's Handbook [1] heat transfer for a cylindrical shape at least 1-inch in diameter may be estimated as follows when there is a 2 fps wind and 1 atmosphere pressure

$$
q_{\mathrm{c}}=0.010 d^{-0.4} A \Delta T
$$

where

$$
\begin{array}{ll}
d & =\text { diameter of the cylinder, inches } \\
A & =\text { Surface area in } 2 / \mathrm{ft} \\
\Delta T & =\text { Difference in temperature in }{ }^{\circ} \mathrm{C} \text { between conductor surface and ambient air temperature }
\end{array}
$$

Thus, for a hypothetical pipe with an O.D. of six inches and $\Delta \mathrm{T}=60^{\circ} \mathrm{C}$

A $\quad=6 \times 3.14 \times 12=226 \mathrm{in}^{2} / \mathrm{ft}$


$$
\begin{aligned}
q_{\mathrm{c}} & =(0.010)\left(6^{-0.4}\right)(226)(60) \\
& =\frac{(0.010)(226)(60)}{2.04} \\
& =66.8 \mathrm{watts} / \mathrm{ft}
\end{aligned}
$$

The heat transfer per unit area is $q_{\mathrm{c}} / A$ or .298 watts $/ \mathrm{in}^{2}$. This value is virtually identical to that calculated for the square tube of the same major dimension and may be taken as an indication of the credibility of both methods.

It is of interest to make the comparison between square tubes and pipes for conductors of other size.

|  | $\boldsymbol{q}_{\mathbf{c}} / \mathbf{A}$, watts/in |
| :---: | :---: | :---: |

[^7]It is seen that for the larger bus conductors the heat transfer efficiency of the pipe is about the same as that of the square tube. In fact they are identical at about 6 inches. Note that the heat transfer efficiency decreases with increasing size of the conductor.

## C.3.2 3. Natural convection for flat and cylindrical surfaces

Some surfaces on conductors or in arrays of conductors may be shielded from direct exposure to wind. Assuming that there is nevertheless sufficient space for natural convection to occur, such surfaces may be treated as though convective losses outdoor would be the same as natural convective losses indoors. For such shielded surfaces heat losses are calculated using generally accepted equations for natural convection.

Examples of areas requiring such treatment are the spaces between double angles, double channels, or parallel rectangular conductors. The use of the natural convection equations is probably justified when the space between conductors is greater than $20 \%$ of the major dimension of the conductor or 1 -inch, whichever is smaller. This estimate of the permissible spacing is based on the fact that the boundary layer for mass transfer is, very roughly, $10 \%$ of the length of the flow path. When the spacing between conductors is greater than the major dimension of the conductor, then the forced convection formulas given above may apply.

Because of the restricted flow away from the interior surfaces of integral web conductors, it is suggested that the natural convection loss formulas given here for surfaces facing down be applied to all interior surfaces.

The appropriate natural convection formulas are as follows:

Vertical or upward facing surfaces and cylinders

$$
q_{\mathrm{c}}=0.0022 \Delta T^{1.25} l^{-0.25} A
$$

Surfaces facing down

$$
q_{\mathrm{c}}=0.0011 \Delta T^{1.25} l^{-0.25} A
$$

where

| $\Delta T$ | $=$ difference in temperature between conductor surface and ambient air temperature in ${ }^{\circ} \mathrm{C}$ |
| :--- | :--- |
| $l$ | $=$ length of conductor surface (width or thickness) in inches $(12 \mathrm{~L})$ |
| $A$ | $=$ conductor surface area in inches ${ }^{2} /$ foot |
| $q_{\mathrm{c}}$ | $=$ conductive heat loss in watts/linear foot |

For 3,6 , and 9 inch wide vertical surfaces at a $60^{\circ} \mathrm{C}$ temperature difference

$$
\begin{aligned}
3^{\prime \prime} q_{\mathrm{c}} / A & =\frac{0.0022(60)^{1.25}}{3^{0.25}} \\
& =0.28 \text { watts } / \mathrm{in}^{2} \\
6^{\prime \prime} q_{\mathrm{c}} / A & =0.234 \text { watts } / \mathrm{in}^{2} \\
9^{\prime \prime} q_{\mathrm{c}} / A & =0.21 \text { watts } / \mathrm{in}^{2}
\end{aligned}
$$

[^8]When surfaces face downward the heat transfer per unit area is only half the value calculated in the above example.

Considering some other temperature differences, we get the following comparison between forced convection and natural convection.

Example
For a 6-inch flat conductor

| $\boldsymbol{q}_{\mathbf{c}} / \mathbf{A}$, watts/in ${ }^{\mathbf{2}}$ |  |  |  |
| :--- | :---: | :---: | :---: |
|  | $\Delta \boldsymbol{T}=\mathbf{8 0}{ }^{\circ} \mathbf{C}$ | $\Delta \boldsymbol{T}=\mathbf{6 0}{ }^{\circ} \mathbf{C}$ | $\Delta \boldsymbol{T}=\mathbf{4 0}{ }^{\circ} \mathbf{C}$ |
| Forced convection <br> (outdoor) | .390 | .293 | .195 |
| Natural convection <br> (indoor or confined <br> spaces) | .335 | .234 | .141 |
| Indoor/outdoor | .86 | .795 | .725 |

Thus, for large conductors and large temperature rises the calculated benefit of the 2 fps wind of heat transfer outdoors over natural convection on favorably oriented surfaces indoors is only $10-20 \%$. The effect on ampacity will be even less and may be as low as only 2 or $3 \%$ for large conductors and high temperatures.

## C.3.2 4. Radiation loss

The basic Stefan-Boltzmann equation for radiation from a surface (or narrow slits, which are treated as black bodies) is as follows:

$$
q_{\mathrm{r}}=36.9 \times 10^{-12} \varepsilon A\left(T_{\mathrm{c}}^{4}-T_{\mathrm{c}}^{4}\right)
$$

where
$\varepsilon \quad$ = emissivity corresponding to the temperatures of interest. Here is assumed emissivity at $T_{\mathrm{c}}$ equals

$T_{\mathrm{c}} \quad \begin{aligned} & \text { absorptivity of energy spectrum at } T_{\mathrm{a}} . \text { This is usually a good approximation. } \\ & T_{\mathrm{a}} \quad \text { temperature of conductor, }{ }^{\circ} \text { Kelvin } \\ & q_{\mathrm{r}} \quad \text { temperature of surrounding bodies, }{ }^{\circ} \text { Kelvin } \\ & \text { radiation loss watts/linear foot }\end{aligned}$

Typical values of $\varepsilon$ for bus conductors are in the range of 0.3 to 0.9 . A value of 0.5 would apply to heavily weathered aluminum while $0.8-0.85$ is appropriate for copper which has achieved a dense green or blackbrown patina. High values of emittance may be achieved also with special paints, coatings or wrappings on the conductor. While high emittance improves heat dissipation via radiation it would also increase heat gain via solar absorption.

## Example

Consider the conductor of emittance equal of 0.5 operating at $100^{\circ} \mathrm{C}\left(373^{\circ} \mathrm{K}\right)$ in an environment of $40^{\circ} \mathrm{C}$ ( $313^{\circ} \mathrm{K}$ ) then

$$
\begin{aligned}
& q_{\mathrm{r}} / A=\left(36.9 \times 10^{-12}\right)(0.5)\left(373^{4}-313^{4}\right) \\
& q_{\mathrm{r}} / A=.18 \text { watts } / \mathrm{in}^{2}
\end{aligned}
$$

By comparing this figure to the forced convective losses calculated earlier it can be seen that radiation losses may make up 30-40\% of the total heat losses. For large conductors with high emissivity, losses by radiation may exceed those due to convection.

## C.3.2 5. Solar heat gain

The heat gained from incident solar radiation is estimated as follows:

$$
q_{\mathrm{s}}=0.00695 \varepsilon^{6} Q_{\mathrm{s}} A^{9} K(\sin \theta)
$$

where

```
\(\varepsilon^{6} \quad=\) coefficient of solar absorption, usually somewhat higher than emmitance, but generally taken as
        equal to that used for radiation loss
    \(\theta=\) effective angle of incidence of sun, \(\cos ^{-1}\left[\cos H_{c} \cos \left(Z_{c}-Z_{1}\right)\right]\)
    \(q_{\mathrm{s}} \quad=\) solar heat gain in watts/linear foot
```

where

```
\(H_{\mathrm{c}} \quad=\) altitude of sun, degrees
\(Z_{c} \quad=\) azimuth of sun, degrees
\(Z_{1} \quad=\) azimuth of conductor line, degrees
    = 0 or 180 for N-S
    \(=90\) or 270 for E-W
\(A^{9} \quad=\) projected area of conductor, square inches per foot (area casting shadow)
\(Q_{\mathrm{s}} \quad=\) total solar and sky radiated heat on a surface normal to sun's rays, watts/sq.ft
\(K \quad=\) heat multiplying factors for high altitudes
```

In cases where solar heat input is high, it is important to consider whether solar heating will peak during the time the maximum current load is on the circuit. If not, the estimate of the solar load should be reduced accordingly in order to arrive at the most cost-effective conductor size.

The projected area of a flat surface is the area of its shadow on a plane normal to the direction of the sun's rays, e.g., per foot of conductor.

$$
A^{9}=12 \sin \zeta \times \text { conductor size }
$$

where
$\zeta \quad=$ angle between plane of the conductor surface and sun's altitude
For a vertical surface

$$
\zeta \quad=90-H_{\mathrm{c}}
$$

For a horizontal surface
$\zeta=H_{\mathrm{c}}$

Table C.1-Data for calculating solar heat gain

| Altitude and Azimuth in Degrees of the Sun at Various Latitudes ${ }^{314}$ Declination 23.0 ${ }^{\circ}$ Northern Hemisphere • June 10 and July 3 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Degrees North Latitude | 10:00A.M. |  | 12:00 N. |  | 2:00 P.M. |  |
|  | $\boldsymbol{H}_{\text {c }}$ | $Z_{\text {c }}$ | $\boldsymbol{H}_{\text {c }}$ | $Z_{\text {c }}$ | $\boldsymbol{H}_{\text {c }}$ | $Z_{\text {c }}$ |
| 20 | 62 | 78 | 87 | 0 | 62 | 282 |
| 25 | 62 | 88 | 88 | 180 | 62 | 272 |
| 30 | 62 | 98 | 83 | 180 | 62 | 262 |
| 35 | 61 | 107 | 78 | 180 | 61 | 253 |
| 40 | 60 | 115 | 73 | 180 | 60 | 245 |
| 45 | 57 | 122 | 68 | 180 | 57 | 238 |
| 50 | 54 | 128 | 63 | 180 | 54 | 232 |
| 60 | 47 | 137 | 53 | 180 | 47 | 223 |
| 70 | 40 | 143 | 43 | 180 | 40 | 217 |

$H_{\mathrm{c}}=62^{\circ}$

Table C.1-Data for calculating solar heat gain (Continued)

| Total Heat Received by a Surface <br> at Sea Level Normal to the Sun's Rays ${ }^{\text {a }}$ |  |  |
| :---: | :---: | :---: |
| Qs watts/sq ft <br> Dolar Altitude <br> Degrees HC | Clear Atmosphere | Industrial <br> Atmosphere |
| 5 | 21.7 | 12.6 |
| 10 | 40.2 | 22.3 |
| 15 | 54.2 | 30.5 |
| 20 | 64.4 | 39.2 |
| 25 | 71.5 | 46.6 |
| 30 | 77.0 | 53.0 |
| 35 | 81.5 | 57.5 |
| 40 | 84.8 | 61.5 |
| 45 | 87.4 | 64.5 |
| 50 | 90.0 | 67.5 |
| 60 | 92.9 | 71.6 |
| 70 | 95.0 | 75.2 |
| 80 | 95.8 | 77.4 |
| 90 | 96.4 | 78.9 |
|  |  |  |

${ }^{\text {a }}$ See Reference 5 at the end of this annex.

# Table C.1-Data for calculating solar heat gain (Continued) 

| Solar Heat Multiplying Factors (K) <br> for High Altitudes ${ }^{\mathbf{a}}$ |  |
| ---: | :---: |
| Elevation above <br> Sea Level, feet | ${\text { Multiplier for } \boldsymbol{Q}_{\mathbf{s}}}^{0} \quad$ |
| 5,000 | 1.00 |
| 10,000 | 1.15 |
| 15,000 | 1.25 |

${ }^{\text {a }}$ See Reference 6 at the end of this annex.

## Examples of Solar Heating <br> Example 1

Assume conductors are in an industrial area on a E-W line at $30^{\circ} \mathrm{N}$ latitude at 5000 foot elevation. If maximum current is required, at 10:00 a.m. from Table C.13.

$$
\begin{aligned}
& Z_{\mathrm{c}}=98^{\circ} \\
& Q_{\mathrm{s}} \text { industrial }=72.3 \text { watts } / \mathrm{ft}^{2} \\
& K=1.15 \text { at } 5,000 \text { feet }
\end{aligned}
$$

Then,

$$
\begin{aligned}
& \theta=\cos ^{-1}[\cos (62) \cos (98-270)] \\
& \therefore \theta=117.5^{\circ} \\
& \sin \theta=0.885
\end{aligned}
$$

For a cylinder, the projected area is $12 \mathrm{~d}\left(\mathrm{in}^{2} / \mathrm{ft}\right)$. Then for a 6 -inch cylinder with $\varepsilon=0.8$.

$$
\begin{aligned}
& q_{\mathrm{s}}=(0.00695)(0.8)(72.3)(12 \times 6)(0.885)(1.15) \\
& q_{\mathrm{s}}=29.5 \text { watts/ft. }
\end{aligned}
$$

## Example 2

Compute typical 10:00 a.m. summertime solar radiation incident on a $6 \times 1 / 2$-inch rectangular bus conductor running E-W at $45^{\circ} \mathrm{N}$ latitude in a clear atmosphere at 5000 feet.

From Table C.1, $H_{\mathrm{c}}=57^{\circ}$
Projected area equals

$$
\begin{array}{ll}
A^{\prime} & =12\left[6 \sin 33^{\circ}+1 / 2 \sin 57^{\circ}\right] \\
A^{\prime} & =44.28 \text { in }^{2} / \mathrm{ft} . \\
\theta & =\cos ^{-1}\left[\left(\cos 57^{\circ}\right)\left[\cos \left(122^{\circ}-270^{\circ}\right)\right]\right] \\
\theta & =\cos ^{-1}[(0.545)(0.53)]=\cos ^{-1}(.-293) \\
\theta & =107^{\circ}
\end{array}
$$

$$
\begin{array}{ll}
\sin \theta & =.96 \\
q_{\mathrm{s}} & =(0.00695)(0.5)(92)(44.28)(1.15)(.96) \\
q_{\mathrm{s}} & =15.6 \text { watts/ft. }
\end{array}
$$

For comparison, consider the radiation loss for the same conductor at $80^{\circ} \mathrm{C}$ with $40^{\circ}$ ambient.

$$
\begin{aligned}
& q_{\mathrm{r}}=\left(36.9 \times 10^{-12}\right)(0.5)(12)(12+1)\left(353^{4}-312^{4}\right) \\
& =17.0 \mathrm{watt} / \mathrm{ft}
\end{aligned}
$$

This is a case where emissivity (absorptivity) is of minor importance in the rating of a bus conductor. In contrast, at a lower altitude and for a greater temperature rise, high emissivity would provide for improved ampacity. It should be noted that except during periods of peak solar loads, high emissivity provides the lowest operating temperatures and therefore the least power loss.

## C.3.2 6. Summation of convective losses

For each of the conventional types of bus conductor, the convective loss areas for which the formulas given in items 1, 2, and 3 apply are as follows.

| Shape | Area for Forced Convection | Area for Natural Convection | Summation of Convection Losses |
| :---: | :---: | :---: | :---: |
| Single Rectangle | 24 (l+t) | 0 | $0.288 \Delta \mathrm{~T}\left(l^{1 / 2}+\mathrm{t}^{1 / 2}\right)$ |
| Multiple (N) Rectangles | $24(l+\mathrm{Nt})$ | $24 l(\mathrm{~N}-1)$ | $\begin{aligned} & 0.288 \Delta \mathrm{~T}\left(l^{1 / 2}+\mathrm{Nt}^{1 / 2}\right)+ \\ & 0.0528 \Delta \mathrm{~T}^{1.25} l^{75}(\mathrm{~N}-1) \end{aligned}$ |
| Round Tube or Bar | $12 \pi \mathrm{~d}$ | 0 | $0.377 \Delta \mathrm{~T} \mathrm{~d}^{0.6}$ |
| Square Tube | $48 l$ | 0 | $0.576 \Delta \mathrm{~T} l^{1 / 2}$ |
| Rectangular Tube ( $l \times \mathrm{w}$ ) | 24(l+w) | 0 | $0.288 \Delta \mathrm{~T}\left(l^{1 / 2}+\mathrm{w}^{1 / 2}\right)$ |
| Universal Angle ( $l \times \mathrm{w}$ )(ignoring thickness) | 24(l+w) | 0 | $0.288 \Delta \mathrm{~T}\left(l^{1 / 2}+\mathrm{w}^{1 / 2}\right)$ |
| Double Angles (for 2 angles) | 24(l+w) | 24(l+w)* | $\begin{aligned} & 0.288 \Delta \mathrm{~T}\left(l^{1 / 2}+\mathrm{w}^{1 / 2}\right)+ \\ & 0.0462 \Delta \mathrm{~T}^{1.25}\left(l^{75}+\mathrm{w}^{25}\right) \end{aligned}$ |
| Single Channel | 24(l+2w) | 0 | $0.288 \Delta \mathrm{~T}\left(l^{1 / 2}+2 \mathrm{w}^{1 / 2}\right)$ |
| Double Channel | $24(l+2 \mathrm{w})$ | 24(l+2w)* | $\begin{aligned} & 0.288 \Delta \mathrm{~T}\left(l^{1 / 2}+2 \mathrm{w}^{1 / 2}\right) \\ & 0.0462 \Delta \mathrm{~T}^{1.25}\left(l^{75}+2 \mathrm{w}^{75}\right) \end{aligned}$ |
| Integral Web | 24(l+2w) | $24(a+2 b+2 c)^{* *}$ | $\begin{aligned} & 0.288 \Delta \mathrm{~T}\left(l^{1 / 2}+2 \mathrm{w}^{1 / 2}\right) \\ & 0.0264 \mathrm{~T}^{1.25}\left(\mathrm{a}^{75}+\right. \\ & \left.2 \mathrm{~b}^{75}+2 \mathrm{c}^{.25}\right) \end{aligned}$ |

* Average over all surfaces on interior assuming equivalent of 3 favorably oriented surfaces and 1 unfavorable $\left[\frac{3(0.0022)+(0.0011)}{4}\right] 24=0.0462$
** Due to overhang count all interior surfaces as unfavorably oriented for natural convection.


## C.3.2 7. Summation of radiation losses

For each of the conventional bus conductors the areas which radiate energy are as follows

| Shape | Surface Area of Material | Areas Which Behave as Black Body Slit or Hole ( $\varepsilon^{\prime}=1$ ) | Summation of Radiation $\operatorname{Loss} \div\left(T_{\mathrm{c}}^{4}-T_{\mathrm{s}}^{4}\right) \times 10^{-12}$ |
| :---: | :---: | :---: | :---: |
| Single Rectangle | $24(l+\mathrm{t})$ | 0 | $8868(l+\mathrm{t})$ |
| Multiple (N) Rectangles (Spacing = S) | $24(l+\mathrm{Nt})$ | ( $\mathrm{N}-1$ )24S | $886 \varepsilon(l+N \mathrm{Nt}+886(\mathrm{~N}-1) \mathrm{S}$ |
| Round Tube or Bar Square Tube | $\begin{aligned} & 12 \pi \mathrm{~d} \\ & 48 l \end{aligned}$ | $\begin{aligned} & 0 \\ & 0 \end{aligned}$ | $\begin{aligned} & 1,390 \varepsilon \mathrm{~d} \\ & 1,772 \varepsilon l \end{aligned}$ |
| Rectangular Tube $(l \times \mathrm{w})$ | $24(l+w)$ | 0 | $8868(l+\mathrm{w})$ |
| Universal Angle | $24(l+\mathrm{w})$ | 0 | $8868(l+\mathrm{w})$ |
| Double Angle (Two <br> Angles) $($ Spacing $=\mathbf{S})$ | $24(l+\mathrm{w})$ | 24S | $886 \varepsilon(l+\mathrm{w}+\mathrm{S} / \varepsilon)$ |
| Channel | $24(l+2 \mathrm{w})$ | 0 | $886 \varepsilon(l+2 \mathrm{w})$ |
| Double Channel (Two Channels) (Spacing = S | $24(l+2 \mathrm{w})$ | 0 | $886 \varepsilon(l+2 \mathrm{w}+\mathrm{S} / \varepsilon)$ |
| Integral Web (overall dimensions $l \times \mathrm{g}$ ) | $24(l+\mathrm{g})$ | 0 | $886 \varepsilon(l+\mathrm{g})$ |

## C.3.2 8. Summation of solar radiation gains

The effective projected area for each of the conventional shapes is given below. Only direct solar radiation has been considered. A smaller amount of energy is radiated from the sky. However, it has been ignored here. If data is available for the particular ocation, sky radiation impinging on other surfaces may be added to the overall energy balance.

| Shape | Effective Projected Area |
| :--- | :--- |
| Single Rectangle | $12\left[l \sin \left(90-H_{\mathrm{c}}\right)+\mathrm{t} \sin H_{\mathrm{c}}\right]$ |
| Multiple $(\mathrm{N})$ Rectangles | $12\left[l \sin \left(90-H_{\mathrm{c}}\right)+(\mathrm{Nt}+(\mathrm{N}-1) \mathrm{S} / \varepsilon) \sin H_{\mathrm{c}}\right]$ |
| Round Tube or Bar | 12 d |
| Square Tube | $12 l\left[\sin \left(90-H_{\mathrm{c}}\right)+\sin H_{\mathrm{c}}\right]$ |
| Rectangular Tube $(l \times \mathrm{w})$ | $12\left[l \sin \left(90-H_{\mathrm{c}}\right)+\mathrm{w} \sin H_{\mathrm{c}}\right]$ |
| Universal Angle | $12\left[l \sin \left(90-H_{\mathrm{c}}\right)+\mathrm{w} \sin H_{\mathrm{c}}\right]$ |
| Double Angle | $12\left[l \sin \left(90-H_{\mathrm{c}}\right)+\mathrm{w} \sin H_{\mathrm{c}}\right]$ |
| Channel | $12\left[l \sin \left(90-H_{\mathrm{c}}\right)+\mathrm{w} \sin H_{\mathrm{c}}\right]$ |
| Double Channel | $12\left[l \sin \left(90-H_{\mathrm{c}}\right)+(2 \mathrm{w}+\mathrm{S} / \varepsilon) \sin H_{\mathrm{c}}\right]$ |
| Integral Web | $\left.12\left[l \sin \left(90-H_{\mathrm{c}}\right)+2 \mathrm{w} \sin H_{\mathrm{c}}+[(\mathrm{g}-2 \mathrm{w}) / \varepsilon)\right] \sin H_{\mathrm{c}}\right]$ |

## C.3.2 9. Computation of ampacity

The ampacity computation requires dividing the sum of the heat losses by the product of the resistance $(R)$ and the skin effect factor $(F)$.

Resistance increases with increasing temperature and this must be accounted for in the calculation. Skin effect factors are a function of resistance, frequency and geometry. The factors are readily available for simple shapes. Calculating skin effect factors for complex shapes is beyond the scope of this paper and no guidance will be offered except that the factors can be significant and should be included when calculations are performed. The skin effect factors decrease slightly with increasing temperature and should be adjusted accordingly. This subject is discussed in the section on properties of materials.

As shown in the section on properties of materials, the resistance at any temperature may be calculated as follows:

For copper and copper alloys

$$
R=\frac{8.145 \times 10^{-4}}{C^{\prime} A_{2}}\left[1+\frac{0.00393 C^{\prime}}{100}\left(T_{2}-20\right)\right]
$$

For aluminum alloys

$$
R=\frac{8.145 \times 10^{-4}}{C^{\prime} A_{2}}\left[1+\frac{0.00403 C}{61}\left(T_{2}-20\right)\right]
$$

where

$$
\begin{array}{ll}
C^{\prime} & =\text { conductivity as } \% \text { IACS } \\
A_{2} & =\text { cross-sectional area, square inches } \\
T_{2} & =\text { conductor temperature },{ }^{\circ} \mathrm{C}
\end{array}
$$

## Example

Compute the 60 cycle outdoor ampacity of a $12^{\prime \prime}$ by $1 / 4^{\prime \prime}$ copper conductor operating with a temperature rise of $65^{\circ} \mathrm{C}$ above a $40^{\circ} \mathrm{C}$ ambient. Assume $\varepsilon=0.5$, no solar heating, $C^{\prime}=98 \%$ IACS and $F=1.28$

$$
\begin{aligned}
& q_{\mathrm{c}}=(0.288)(65)\left[12^{1 / 2}+(1 / 4)^{1 / 2}\right] \\
& q_{\mathrm{c}}=74 \mathrm{watts} / \mathrm{ft} \\
& q_{\mathrm{r}}=(886)(0.5)(12.250)\left(10^{-12}\right)\left(378^{4}-313^{4}\right) \\
& q_{\mathrm{r}}=58.2 \mathrm{watts} / \mathrm{ft} \\
& q_{\mathrm{c}}+q_{\mathrm{r}}=132.2 \mathrm{watts} / \mathrm{ft}
\end{aligned}
$$

$$
\begin{aligned}
R & =\frac{8.145 \times 10^{-4}}{(98)(12)(1 / 4)}\left[1+\frac{0.00393(98)}{100}(105-20)\right] \\
& =3.68 \times 10^{-4} \mathrm{ohms} / \mathrm{ft} \\
R F & =(3.68)(1.28) \times 10^{-4}=4.7 \times 10^{-4} \mathrm{ohms} / \mathrm{ft} \\
I & =\left[\left(q_{\mathrm{c}}+q_{\mathrm{r}}\right) / R F\right]^{1 / 2}=10^{3}(132.2 / 4.7)^{1 / 2} \\
I & =5,310 \mathrm{amps}
\end{aligned}
$$

## C. 4 Properties of materials

## C.4.1 Thermal expansion

Bus conductors expand as their temperatures rise. The amount of expansion may be calculated by multiplying the coefficients below by the increase in temperature. The base temperature corresponding to zero expansion is the installation temperature not the ambient temperature.

## Material

Table C.2-Thermal expansion multiplication coefficients

|  | Average Coefficient of Thermal Expansion for the Range Indicated |  |
| :---: | :---: | :---: |
|  | $\begin{gathered} \text { in/in } .^{\circ} \mathbf{F} \\ \left(68-212^{\circ} F\right) \end{gathered}$ | $\begin{gathered} \text { in } / \mathbf{i n}^{\circ} \mathrm{C} \\ \left(20-100^{\circ} \mathrm{C}\right) \end{gathered}$ |
| Aluminum and Alloys | $13.0 \times 10^{-6}$ | $23.4 \times 10^{-6}$ |
| Copper and Alloys | $9.22 \times 10^{-6}$ | $16.6 \times 10^{-6}$ |
| Steel | $6.3 \times 10^{-6}$ | $11.4 \times 10^{-6}$ |
| Concrete | 3.5 to $8 \times 10^{-6}$ | 6.3 to $14.4 \times 10^{-6}$ |

Example
What is the total thermal expansion of a 15 -foot run of copper bus conductor installed on a concrete pad at $20^{\circ} \mathrm{C}$ and operating at $50^{\circ} \mathrm{C}$ over a $40^{\circ} \mathrm{C}$ ambient (i.e. at $90^{\circ} \mathrm{C}$ )

For the bus conductor
$\Delta$ copper $=$ total expansion $=(12)(15)\left(16.8 \times 10^{-6}\right)(70)=0.211$ inches
For the concrete pad (assume coefficient expansion $=10 \times 10^{-6}$ )
$\Delta$ concrete $=(12)(15)\left(10 \times 10^{-6}\right)(20)=0.036$ inches

Net amount of restraint on bus conductor is the difference between the expansion of the bus and the concrete pad
$\Delta$ net $=0.211-0.036=0.175$ inches
The strain on the copper (assuming massive rigid pad) is

$$
\frac{\Delta n e t}{L^{\prime}}=\frac{0.175}{12 \times 15}=0.001 \text { inches } / \text { inch }
$$

where
$L^{\prime} \quad=$ length of restrained conductor in same units as $\Delta$ net

## C.4.2 Stresses and forces due to thermal expansion

When a material is totally restrained from expanding or contracting normally as temperatures change, stresses are induced to account for the effective change in length.

The stress, $S$, is

$$
S=\frac{E \Delta n e t}{L^{\prime}}
$$

where

$$
E \quad=\text { modulus of elasticity }
$$

For the materials of construction
Table C.3-Modulus of elasticity

|  | E, modulus of elasticity, $\times \mathbf{1 0}^{\mathbf{4}} \mathbf{~ p s i}$ |  |  |
| :--- | :---: | :---: | :---: |
|  | $\mathbf{2 0}{ }^{\circ} \mathbf{C}$ | $\mathbf{5 0}^{\circ} \mathbf{C}$ | $\mathbf{1 0 0}{ }^{\circ} \mathbf{C}$ |
| Aluminum | 10 | 10 | 10 |
| Copper | 17 | 16.5 | 16 |
| Steel | 30 | 30 | 30 |
| Concrete | 3 to 5 | 3 to 5 | 3 to 5 |

Example
For the example above

$$
\mathrm{S}=17 \times 10^{6} \times 10^{-3}=17,000 \mathrm{psi}
$$

The total load is $S \times A_{2}$
where

$$
A_{2}=\text { cross-sectional area, sq. inches }
$$

For $6^{\prime \prime} \times 1 / 2^{\prime \prime}$ bus conductor the associated load on the bus supports in the above case would be 51,000 pounds. In practice this high load would not be generated. Complete restraint is unlikely due to bending, sliding, or plastic deformation of the conductors. However, to be sure loads are not excessive it is suggested that expansion joints be provided to minimize thermally generated stresses.

## C.4.3 Maximum operating stresses

Metals may deform plastically to accommodate thermal stresses and strains and reduce other applied loads. While bus conductor alloys can deform appreciably it is suggested that stresses be maintained below levels at which plastic deformation is expected. If the loads will be applied occasionally and for only a short time the maximum stress should be below the yield strength. It must be remembered that, at the yield stress of a material, a small amount of deformation (less than $1 / 2$ percent) occurs. For extended operation or negligible deformation lower stresses must be employed to avoid creep, relaxation or fatigue damage. To provide a margin of safety designers may limit stresses to $2 / 3$ the values given below in Table C.4.

Table C.4-Operating stresses

| Representative Yield Strength Levels, psi |  |  |  |
| :---: | ---: | ---: | ---: |
|  | $\mathbf{2 0}^{\circ} \mathbf{C}$ | $\mathbf{1 0 0}^{\circ} \mathbf{C}$ | $\mathbf{1 5 0}^{\circ} \mathbf{C}$ |
| Aluminum Alloys |  |  |  |
| 6101-T6 | 25000 | 22300 | 16900 |
| 6063-T6 | 25000 | 22700 | 16200 |
| Copper (Hard) | 25000 | 22000 | 20000 |
| Copper (Soft) | 9000 | 9000 | 9000 |
| Maximum Stresses for Continuous Operation, psi |  |  |  |
|  |  |  |  |
| Aluminum Alloys | $\mathbf{2 0}^{\circ} \mathbf{C}$ | $\mathbf{1 0 0}{ }^{\circ} \mathbf{C}$ | $\mathbf{1 5 0}^{\circ} \mathbf{C}$ |
| 6101-T6 | 15000 | 13380 | 10140 |
| 6063-T6 | 15000 | 13620 | 9720 |
| Copper (Hard) | 9000 | 9000 | 8700 |
| Copper (Soft) | 5100 | 4800 | 4700 |

The above strength levels apply to the usual conductor materials. Special alloys of aluminum or copper and coppers with small additions of silver may be used where higher strength or resistance to relaxation or softening are required.

## C.4.4 Resistance

Resistance of bus conductors increases with increasing temperature. For aluminum and copper alloys, resistance at an elevated temperature $\left(T_{2}\right)$ may be expressed in terms of resistance at $20^{\circ} \mathrm{C}$ as follows

$$
R_{\mathrm{T} 2}=R_{20}\left[1+\alpha\left(T_{2}-20\right)\right]
$$

where
$\propto \quad=$ temperature coefficient of resistance for a base of $20^{\circ} \mathrm{C}\left(\mathrm{ohms} / \mathrm{ohms}^{-}{ }^{\circ} \mathrm{C}\right)$

$$
\begin{aligned}
R_{20} & =\text { resistance at } 20^{\circ} \mathrm{C} \text { per unit length in ohms/foot } \\
& =\rho A_{2}
\end{aligned}
$$

where

$$
\begin{array}{ll}
\rho & =\text { resistivity, ohm-in }{ }^{3} / \mathrm{ft} \\
\mathrm{~A}_{2} & =\text { cross-sectional area of conductor at } 20^{\circ} \mathrm{C}, \text { in sq. in. }
\end{array}
$$

The temperature coefficient of resistance for copper of conductivity equal to $100 \%$ of the International Annealed Copper Standard (IACS) is $0.00393 /{ }^{\circ} \mathrm{C}$ and for aluminum of conductivity equal to $61 \%$ IACS it is $0.00403 /{ }^{\circ} \mathrm{C}$. For copper and aluminum conductors of other conductivities the following relations may be written for
$\mathrm{C}^{\prime}=\%$ conductivity (as \% IACS)
$\alpha c u=\frac{0.00393 C^{\prime}}{100}$
$\alpha \mathrm{al}=\frac{0.00403 C^{\prime}}{61}$

The above relations give the following for copper

$$
R_{\mathrm{T} 2}=\rho / A_{2}\left[1+\frac{0.00393 C^{\prime}}{100}\left(T_{2}-20\right)\right]
$$

and for aluminum

$$
R_{\mathrm{T} 2}=\rho / A_{2}\left[1+\frac{0.00403 C^{\prime}}{61}\left(T_{2}-20\right)\right]
$$

For copper of $100 \%$ IACS conductivity the resistivity, $\rho$, is $8.145 \times 10^{-6} \mathrm{ohm}-\mathrm{in}^{2} / \mathrm{ft}$. Then

## Copper

$$
R_{\mathrm{T} 2}=\frac{8.145 \times 10^{-6}}{C^{\prime} A_{2}}\left[1+\frac{0.00393 C^{\prime}}{61}\left(T_{2}-20\right)\right]
$$

Aluminum

$$
R_{\mathrm{T} 2}=\frac{8.145 \times 10^{-6}}{C^{\prime} A_{2}}\left[1+\frac{0.00403 C^{\prime}}{100}\left(T_{2}-20\right)\right]
$$

## C.4.5 Emissivity and absorptivity

For ordinary calculations the emissivity and absorptivity of a bus conductor are taken as equal. Strictly speaking, since they apply to different energy spectra they are not equal, but for practical purposes the error is small.

For conditions of interest,

## Table C.5-Emissivity and absorptivity of material

| $\varepsilon=$ Emissivity, absorptivity |  |  |
| :--- | :--- | :--- |
|  | Copper | Aluminum |
| Clean Mill Finish | 0.1 | 0.1 |
| Light Tarnish <br> (recent outdoor <br> installation or indoor) | $0.3-0.4$ | 0.2 |
| After Extended <br> Outdoor Exposure | $0.7-0.85$ | $0.3-0.5$ |
| Painted Black | $0.9-0.95$ | $0.9-0.95$ |

## C.4.6 Skin effect

For common conductor shapes plots are available which provide skin effect coefficients as a function of current frequency and resistivity. When such plots are available the variation in skin effect with temperature may be determined by computing the resistivity of the shape at various temperatures and determining the associated skin effect coefficients. When only a single value of the skin effect coefficient is suitable or when a convenient equation is needed for computer calculations, the following procedure may be used to obtain a conservative (slightly) high estimate of the skin effect coefficient at a higher temperature.

For

$$
\begin{array}{ll}
F_{1} & =\text { skin effect coefficient at temperature } T_{1} \\
F_{2} & =\text { skin effect coefficient at temperature } T_{2}
\end{array}
$$

Then

$$
F_{2}=F_{1}+\frac{\Delta F}{\Delta T}\left(T_{2}-T_{1}\right)
$$

Normally the skin effect coefficient is given as a function of Frequency/Resistivity $\times 10^{3}$ which we will define as X for convenience here. Then

$$
\begin{aligned}
& \frac{\Delta F}{\Delta T} \approx \frac{d F}{d t}=\frac{d F}{d X} \frac{d X}{d R} \frac{d R}{d t} \\
& \frac{d X}{d R}=-\frac{X}{2 R} \\
& \frac{d R}{d t} \approx R \alpha
\end{aligned}
$$

A conservative estimate of $d F / d X$ is always $(F-1) / X$. Then

$$
\frac{\Delta F}{\Delta T} \approx-\frac{(F-1)}{X} \times \frac{1}{R} \times \frac{X}{R} \times R \alpha
$$

$$
\approx-1 / 2 \alpha(F-1)
$$

Therefore

$$
F_{2}=F_{1}-\frac{1}{2}\left(T_{2}-T_{1}\right)\left(F_{1}-1\right) \alpha
$$

## C. 5 Ampacity tabulations

The procedures described herein have been used to calculate ampacity tables which are a separate document.

## C. 6 References

[1] Chemical Engineer's Handbook, J. H. Perry, ed. McGraw-Hill Book Company, 1950. Chapter 6 by McAdams, W. H.
[2] McAdams, W. H., Heat Transmission, McGraw-Hill Book Co., N.Y., 1954.
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[4] Sight Reduction Tables for Air Navigation, U. S. Navy Hydrographic Office, H. O. Publication No. 249, Vols. II and III.
[5] Heating, Ventilating and Air-Conditioning Guide 1956, American Society of Heating and Air-Conditioning Engineers.
[6] Yellot, J. I., "Power from Solar Energy," ASME Transactions Vol. 79, No. 6, AUgust, 1957, pp. 13491357.

Note 1
The wind is considered a forced draft with the air circulating parallel to each surface of the conductor and perpendicular to the length

This paper is part of the work of a task force of the IEEE Substations Committee's Working Group 69.1 "Rigid Bus Design Criteria for Outdoor Substations." Messrs. Bleshman, Pemberton, Craig and Prager are members of that task force.

Discussion
W. H. Dainwood, J. E. Holladay, and S. W. Kercel (Tennessee Valley Authority, Knoxville, TN: The authors should be commended on this paper in which they have presented a very sophisticated method of calculating the temperature rise for a certain value of current. It should become an important reference for design of rigid bus systems.

We are utilizing a procedure for calculating temperature rise that is similar to the authors' approach. However, at the present our computerized procedure is limited to tubular and solid round conductors. We use the equations for heat loss which are in the Westinghouse Electrical Transmission and Distribution Reference Book, copyright 1964, Fourth Edition, Fifth Printing. Also used as a reference is the book Elements of Power System Analysis, second edition, by William D. Stevenson, Jr. As with the equations in this paper, the ones we use express current as a function of temperature rise. Primarily, we are interested in specifying a value of current and determining the temperature rise. To do this, we use the Newton-Raphson technique to solve the
equation which expresses the current as a function of temperature rise. Have the authors considered this approach?

We would suggest that the authors include, under "PROPERTIES OF MATERIALS" No. 6, Skin Effect, the method for calculating the skin effect ratio defined as $\frac{A C \text { resistance }}{\text { DC resistance. }}$. It appears the authors have given a conservative method for estimating the skin effect ratio. This estimate approach seems to be somewhat in disagreement with the statement in the $A B S T R A C T$ which says, "This paper will allow the engineer to reexamine the factors involved in increased current loadings of rigid bus and possibly determine new thermal limits." If the object of the paper is to move away from conservative estimates and look at what is actually happening, then it appears that more explicit equations for skin effect could also be presented. We feel that this would further enhance a very significant paper.

The following is an extract from the computer program which we have developed:
Calculation of skin effect ratio:

The literature defines a quantity in

$$
m=\sqrt{\frac{4 \pi \omega}{\rho}}
$$

where
$\begin{array}{ll}\omega & =2 \pi f \\ \rho & =D C \text { resistivity in } \mu \text { ohm }-m\end{array}$
Stevenson demonstrates (Power System Analysis, pages 81-82):

$$
m r=.0636 \sqrt{\frac{F}{R o}}
$$

where

$$
\begin{array}{ll}
\mathrm{F} & =60 \mathrm{~Hz} \\
\mathrm{Ro} & =\mathrm{ohm} / \mathrm{mil}(\mathrm{DC})
\end{array}
$$

or

$$
m r=.0636 \sqrt{\frac{F}{R o 5280}}
$$

Ro ohm/ft (DC)

For a solid round conductor of radius $r$

Now it follows from this that:

$$
m=\frac{m r}{r}=\frac{.0636}{r} \sqrt{\frac{F}{R o 5280}}
$$

Where Ro is the DC resistance in $\Omega / \mathrm{ft}$ of a solid conductor of radius $r$.

By calculating $m$ by this formula, you can be sure the units will come out right.
The ratio is (Electrical Coils and Conductors, page 172):

$$
\frac{R_{A C}}{R_{D C}}=\operatorname{Re}\left[\frac{\alpha(q-r)(q+r)}{2 r}\left(\frac{I_{o}(\alpha r) K_{o^{\prime}}(\alpha q)-K_{o}(\alpha q r) I_{o}(\alpha q)}{I_{o^{\prime}}(\alpha r) K_{o^{\prime}}(\alpha q)-K_{o^{\prime}}(\alpha r) I_{o^{\prime}}(\alpha q)}\right)\right]
$$

where

$$
\begin{aligned}
& \alpha \quad=\sqrt{j} \mathrm{~m} \\
& \mathrm{j} \quad=\sqrt{-1} \\
& \mathrm{I}_{\mathrm{O}}(\alpha r)=\text { ber } \mathrm{mr}+\mathrm{j} \text { bei } \mathrm{mr} \\
& I_{o}(\alpha q)=\text { ber } m q+j \text { bei } m q \\
& K_{o}(\alpha r)=\text { ker } m r+j \text { kei } m r \\
& K_{\mathrm{O}}(\alpha q)=\text { ker } m q+j \text { kei } m q \\
& \mathrm{I}_{\mathrm{o}}{ }^{\prime}(\alpha r)=\mathrm{e}^{-\mathrm{j} \pi / 4}\left(\text { ber }^{\prime} \mathrm{mr}+\mathrm{j} \text { bei' } \mathrm{mr}\right. \text { ) } \\
& I_{o}{ }^{\prime}(\alpha q)=e^{-j \pi / 4}\left(\text { ber' }^{\prime} m q+j\right. \text { bei' mq) } \\
& \mathrm{K}_{\mathrm{o}}{ }^{\prime}(\alpha r)=\mathrm{e}^{-\mathrm{j} \pi / 4}\left(\mathrm{ker}^{\prime} \mathrm{mr}+\mathrm{j} k \mathrm{kei}^{\prime} \mathrm{mr}\right) \\
& \mathrm{K}_{\mathrm{o}}{ }^{\prime}(\alpha q)=\mathrm{e}^{-\mathrm{j} \pi / 4}\left(\mathrm{ker}^{\prime} \mathrm{mq}+\mathrm{j} k \mathrm{kei}^{\prime} \mathrm{mq}\right)
\end{aligned}
$$

Where the following bessel functions are defined by infinite series:
ber ax $=1-\frac{(a x / 2)^{4}}{(2!)^{2}}+\frac{(a x / 2)^{8}}{(4!)^{2}}-\ldots$
bei ax $=\frac{(a x / 2)^{2}}{(1!)^{2}}-\frac{(a x / 2)^{6}}{(3!)^{2}}+\frac{(a x / 2)^{10}}{(5!)^{2}}-\ldots$
ber' $^{\prime}$ ax $=-\frac{2 a(a x / 2)^{3}}{(2!)^{2}}+\frac{4 a(a x / 2)^{7}}{(4!)^{2}}-\ldots$
bei' $\mathrm{ax}=\frac{a(a x / 2)}{(1!)^{2}}-\frac{3 a(a x / 2)^{5}}{(3!)^{2}}+\ldots$
ker ax $=-\left(\ln \frac{a x}{2}+C\right)$ ber ax $+\frac{\pi}{4}$ bei ax $-\lambda 2+\lambda 4-\ldots$
kei ax $=-\left(\ln \frac{a x}{2}+C\right)$ bei ax $-\frac{\pi}{4}$ ber ax $+\lambda 1-\lambda 3+\ldots$
where $C=.57721566490153286061$

$$
\lambda K=\left(\frac{(a x / 2)^{2 k}}{(K!)^{2}}\right)\left(1+\frac{1}{2}+\frac{1}{3}+\ldots \frac{1}{K}\right)
$$

$$
\begin{aligned}
& \text { ker' }^{\prime} \mathrm{ax}=-\left(\ln \frac{a x}{2}+C\right) \text { ber'ax }-\frac{1}{x} \text { ber ax }+\frac{\pi}{4} \text { bei'ax }-\lambda 2^{\prime}+\lambda 4^{\prime}-\ldots \\
& \text { kei ax }=-\left(\ln \frac{a x}{2}+C\right) \text { bei'ax }-\frac{1}{x} \text { bei ax }-\frac{\pi}{4} \text { ber'ax }+\lambda 1^{\prime}=\lambda 3^{\prime}+\ldots \\
& \lambda^{\prime} K=\frac{2 K}{x} \lambda K
\end{aligned}
$$

M. Prager, D. L. Pemberton, A. G. Craig, and N. A. Bleshman: The authors thank Messrs. Dainwood, Holladay, and Kercel for their timely comments. The formulas presented in the paper were selected as the first stage in a program to develop ampacity tables for commercial bus conductors. Such tables may be used alternatively to determine the allowable current for a specified temperature rise or the temperature rise for a specified current. Many of these tables have been prepared based on these formulas and they will be the subject of a forthcoming paper.

When a quick estimate of the temperature rise for 2 given current is needed the following procedure may be used without the need for a computer. The temperature term in the expression for radiation loss (i.e., $\mathrm{T}_{2}{ }^{4}$ $\mathrm{T}_{1}{ }^{4}$ ) may be approximated by $1.6 \times 10^{8} \Delta \mathrm{~T}$ and the exponential term in the natural convection equations ( $\Delta \mathrm{T}^{1.25}$ ) may be approximated by $2.8 \Delta \mathrm{~T}$. Substituting these terms into the general expression relating current to heat loss and resistance $\mathrm{I}=\sqrt{\frac{q c+q r-q s}{R F}}$ provides an equation in which the unknown $(\Delta \mathrm{T})$ appears to the first power. The solution is then easily obtained by solving that equation.

In using the expression suggested by Messrs. Dainwood et. al. to calculate the skin effect ratio, it must be remembered that the temperature coefficient should be included in the resistivity term. The authors took the approach that since the skin effect ratios for conductors were usually available at one temperature, such data could be modified conveniently by the method shown in the text $\mathrm{F}_{2}=\mathrm{F}_{1}-1 / 2\left(\mathrm{~T}_{2}-\mathrm{T}_{1}\right)\left(\mathrm{F}_{1}-1\right) \alpha$ The error introduced is negligible for practical purposes.

## Annex D

## (informative)

## Calculation of surface voltage gradient

The allowable surface voltage gradient $E_{\mathrm{o}}$ for equal radio-influence (RI) generation for smooth, circular conductors is a function of bus-conductor diameter, barometric pressure, and operating temperature. It may be calculated as follows:

$$
\begin{equation*}
E_{\mathrm{o}}=\delta g_{\mathrm{o}} \tag{D1}
\end{equation*}
$$

where
$E_{\mathrm{o}}=$ allowable surface voltage gradient, $\mathrm{kV} \mathrm{rms} / \mathrm{cm}$
$g_{\mathrm{o}}=$ allowable surface voltage gradient under standard conditions for equal radio-influence generation and for circular conductors, $\mathrm{kV} \mathrm{rms} / \mathrm{cm}$ (see Figure D.1)
$\delta=\frac{7.05 b}{459+F}$
where

$$
\begin{aligned}
& \delta=\text { air density factor } \\
& b=\text { barometric pressure, } \mathrm{cm} \text { of } H_{\mathrm{g}} \\
& F=\text { temperature },{ }^{\circ} \mathrm{F}
\end{aligned}
$$



Figure D.1-Allowable surface voltage gradient for equal radio-influence generation under standard conditions versus bus diameter

The temperature to be used in Equation (D1) is generally considered to be the conductor operating temperature. Table D. 1 gives standard barometric pressure corrected for various altitudes above sea level.

Table D.1-Standard barometric pressure (for various altitudes)

| Altitude (ft) | Altitude (m) | Pressure <br> $\left(\mathbf{c m}\right.$ of $\left.\mathbf{H}_{\mathbf{g}}\right)$ |
| :---: | :---: | :---: |
| -1000 | -300 | 79.79 |
| -500 | -150 | 77.39 |
| 0 | 0 | 76.00 |
| 1000 | 300 | 73.30 |
| 2000 | 600 | 70.66 |
| 3000 | 900 | 68.10 |
| 4000 | 1201 | 65.63 |
| 5000 | 1501 | 63.22 |
| 6000 | 1801 | 60.91 |
| 8000 | 2402 | 56.44 |
| 10000 | 3003 | 52.27 |
| 15000 | 4504 | 42.88 |
| 20000 | 6006 | 34.93 |

The average and maximum surface voltage gradients at the surface of smooth circular conductors, at operating voltage, may be determined by the following formulae from NEMA CC 1-1993.


Figure D.2-For single conductor

Conductor


$$
\begin{aligned}
& E_{a}=\frac{V_{1}}{\frac{d}{2} L_{n}\left(\frac{4 h_{e}}{d}\right)} \\
& E_{m}=\frac{h_{e}}{h_{e}-\frac{d}{2}} E_{a} \\
& h_{e}=\frac{h D}{\sqrt{4 h^{2}+D^{2}}}
\end{aligned}
$$

Figure D.3-For three-phase conductor
where

```
\(h=\) distance from center of conductor to ground plane, cm [in]
\(h_{e}=\) equivalent distance from center of conductor to ground plane for three phase, cm [in]
\(d=\) diameter of the individual conductor, \(\mathrm{cm}[\mathrm{in}]\)
\(D=\) phase-to-phase spacing for three phase, \(\mathrm{cm}[\mathrm{in}]\)
\(V_{1}=\) line-to-ground test voltage, kV
\(E_{a}=\) average voltage gradient at the surface of the conductor, \(\mathrm{kV} / \mathrm{cm}[\mathrm{kV} / \mathrm{in}]\)
\(E_{m}=\) maximum voltage gradient at the surface of the conductor, \(\mathrm{kV} / \mathrm{cm}[\mathrm{kV} / \mathrm{in}]\)
```

NOTE $-V_{1}=110 \%$ of nominal operating line-to-ground voltage
For the three-phase configuration the center conductor has a gradient approximately $5 \%$ higher than the outside conductors. For bundled circular conductors, formulae for calculating the surface voltage gradient may be obtained from NEMA CC 1-1993.

For satisfactory operation, $E_{m}$ must be less an $E_{0}$.

## Annex E

(informative)

# Mechanical forces on current-carrying conductors 

By E. D. Charles ${ }^{11}$

## Synopsis

Following a brief review of the standard formulae in connection with the forces on current-carrying conductors the author examines the problem from a more general standpoint, and derives formulae for both the distribution and direction of forces on conductors lying at any angle in different planes. It is felt that these formulae [Equations (E4) and (E5)], which have not previously been published, will be of value for the following reasons: (i) The approximations obtained by application of the standard formulae to non-standard conductor arrangements may lead to serious over- or under-estimation of the true magnitude of mechanical forces and their moments. (ii) A precise knowledge of the direction of the resultant mechanical force is of considerable importance in determining the cantilever stress in the very long insulator stacks used for h.v. installations. (iii) The general formulae put forward are comprehensive in that all the standard formulae for the distribution and direction of forces may be readily obtained by suitable substitution.

List of symbols
$d \quad=$ shortest distance between centre-lines of two straight cylindrical conductors crossing each other obliquely in different planes, $m$
$d F \quad=$ mechanical force on element $d x$ of conductor, N
$F_{p}\left(=\frac{d F}{d x}\right)=$ mechanical force per unit length at point P on conductor, $\mathrm{N} / \mathrm{m}$
$F_{h}, F_{v}=$ horizontal and vertical components of $F_{p}, \mathrm{~N} / \mathrm{m}$
$I_{1} I_{2 x}=$ current in conductors, A
$\chi \quad=$ angle between direction of mechanical force on an element of conductor and the plane in which the conductor lies
$\alpha \quad=$ angle between conductor and the direction of the magnetic field in which it lies
$\beta \quad=$ angle between one conductor and the trace of the other in a plane perpendicular to the shortest distance between the two conductors

## E. 1 Introduction

A large number of papers have been written in connection with the forces of attraction and repulsion between current-carrying conductors. Following the work of Ampère, Laplace, Biot, and Savart, the underlying principles were well established, and a number of other investigators formulated methods of computing the forces in several practical arrangements of conductors lying in a plane or crossing each other at right angles.

In the paper a general formula is given from which may be calculated the distribution of mechanical forces along current carrying conductors which lie at any angle in different planes.

[^9]
## E. 2 Conductor arrangements

It is well known that adjacent current-carrying conductors experience a mechanical force which depends upon the magnitude of the current and the geometrical configuration of the conductors.

The forces which arise under short-circuit conditions may amount to several tons and must be taken into account in the design of conductors, insulators and their supporting structures. The calculation of the forces is a simple matter in the case of very long, straight, parallel busbars, because for all practical purposes the forces are uniformly distributed along the length of the conductors. At the extreme ends of the conductors the forces actually 'tail off' owing to the reduction of magnetic-field strength, but this so-called 'end effect' is only of importance in conductor arrangements in which short lengths, bends, taps, and cross-overs form part of the complete circuit. (Frick, [1]) ${ }^{12}$

A knowledge of the way in which the mechanical forces are distributed along a conductor is a first requirement in computing both the total force and the moment of these forces about a particular point.

The total force on a section of conductor is obtained by integrating the force per unit length over the section. In a similar manner, the moment of the force on a particular section of conductor about a specified point is found by integrating the product of force per unit length times distance to the point. The mathematical integration of the expression for force per unit length is possible only in simple arrangements such as those shown in Figure E.1, so that, in the general case, graphical methods of investigation must be adopted.

The mechanical force on a particular conductor forming part of a complete circuit is found by summing the component forces calculated for the individual conductor members making up the circuit. The conductor members are treated in pairs, each member being taken in combination with every other member, although it is often possible to neglect the more remote parts of the circuit when it is estimated that their effects are negligible compared with other component forces.

It is assumed that the conductors are of circular cross section and that the current is concentrated along the axis of the conductor. No error is introduced by this latter assumption, since, neglecting proximity effects with alternating current, the external magnetic field due to current in a cylindrical conductor does not depend upon the radius of the conductor. Proximity effects need not be considered where the clearance between two members is more than twice the diameter of the conductor.

When the conductors are near together, the mechanical forces in conductors of rectangular cross-section are different from those in conductors of circular cross-section, and for further information the reader should consult the references [2] through [5].

Methods of calculating electromagnetic forces are presented in textbooks and papers for the following cases, illustrated in Figures E.1a, E.1b and E.1c:
a) Parallel conductors
b) Right-angled cross-over conductors
c) Conductors at any angle lying in a plane

The formula for case c) was first introduced by Dunton in 1927. (Dunton, [E6])

For ease of calculation, a complete circuit is usually simplified by regarding it as a combination of the arrangement $a$ ), b) and $c$ ), and a further simplification is often obtained in arrangement $c$ ) by assuming that the angle between the conductors is a right angle. Although these approximations suffice in many practical

[^10]cases, problems may arise where greater accuracy is desired, and in these circumstances a more detailed calculation may be justified.

## E. 3 Skewed-conductor arrangements

It will be realized that $a$ ), b), and $c$ ) are special cases of a more general arrangement in which the axes of the conductors are two straight lines skewed in space at any angle relative to each other, as in Figure E.2.

The definition of two skew lines is that they neither intersect nor are parallel, although $a$ and $c$ of Figure E. 1 may be regarded as limiting cases.

In pure geometry it is shown that, if two lines JD and HA neither intersect nor are parallel, then (see Figure E.2)
(i) there is one straight line CB which is perpendicular to both the given lines
(ii) the length, $d$, of the common perpendicular is the shortest distance between the lines

It follows that JD and HA lie in two parallel planes separated by the distance CB. Thus the general case can be analyzed by using one conductor HA and the shortest distance CB to form the framework of reference shown in Figure E.3. The special cases shown in Figure E. 1 are obviously obtained from Figure E. 3 as follows:
a) $\beta=0^{\circ}$ (parallel)
b) $\beta=90^{\circ}$ (right-angled cross-over)
c) $d=0$ (any angle in a plane)


0


0


Figure E.1-Conductor arrangements-special cases


Figure E.2-Skewed conductors


Figure E.3-Skewed conductors-reference axes and dimensions

## E. 4 Distribution and direction of forces

It has already been pointed out that the mechanical forces experienced by current-carrying conductors are not uniformly distributed along their length, the degree of non-uniformity being more pronounced in the case of short lengths, bends and cross-overs. The direction of the forces depends upon the relative directions of the currents. In the parallel arrangement, the conductors are attracted when the currents are in the same direction and repelled when the currents are in opposite directions. Two circuits crossing obliquely attract each other when both the currents proceed from or to the apparent point of intersection but repel each other if one current proceeds from and the other towards that point.

Figure E. 4 shows the approximate distribution and the direction of forces in the three special cases $(a),(b)$, and (c) when the currents are flowing in directions such as to cause repulsion between the two conductors. If one of the currents is reversed, the direction of the forces will also be reversed. All forces are at right angles to the conductor. In these special cases it will be observed that the mechanical forces are uniplanar.


Figure E.4-Direction of forces-special cases

a Magnitude and direction of forces on conductor JD viewed axially
$b$ Orthogonal components of $F_{p}$
Figure E.5-Direction of forces-skewed conductors

Figure E. 5 shows the skewed-conductor arrangement carrying currents $I_{1}$ and $I_{2}$ in the directions shown. Consider an elemental portion $d x$ of the conductor JD at point $\mathrm{P}_{1}$. The direction of the magnetic flux at point $\mathrm{P}_{1}$ due to the current $I_{1}$ in conductor HA is normal to the plane $\mathrm{HP}_{1} \mathrm{~A}$. The mechanical force $F_{1}$ experienced by the element $d x$ is in a direction at right angles to both the flux at point $\mathrm{P}_{1}$ and to the conductor JD. The line from $\mathrm{P}_{1}$ representing the force $F_{1}$ therefore lies in the plane $\mathrm{HP}_{1} \mathrm{~A}$ and is radial to the conductor JD . In the same way, the forces experienced by an element of conductor JD at any other point such as $\mathrm{P}_{0}, \mathrm{P}_{2}, \mathrm{P}_{3}$, etc., are radial to JD and lie in the planes containing both the conductor HA and the point considered. The angles at which the forces $F_{0}, F_{1}, F_{2}$, etc., act depend upon the values of $x, d$ and $\beta$. Figure E.5a shows the magnitude and direction of the forces as viewed along the axis of the conductor. Figure E.5b shows the horizontal and vertical components of $F_{\mathrm{p}}$.

$$
F_{\mathrm{h}}=F_{\mathrm{p}} \cos X
$$

$$
F_{\mathrm{v}}=F_{\mathrm{p}} \sin X
$$

where, as shown in E.10.2, $X=\tan ^{-1}(d / x \tan \beta)$.

## E. 5 Development of general formula for the distribution of mechanical forces in current-carrying conductors

## E.5.1 Magnitude of force per unit length

In Figure E.6, HA and JD are two conductors of circular cross-section carrying currents $I_{1}$ and $I_{2}$ amperes, respectively.

Consider the force $d F_{s}$ on a length $d x$ of conductor JD at P due to current flowing in element $d s$ of conductor HA at G. According to Ampère's law, the force between the current elements $d x$ and $d s$ is

$$
\begin{equation*}
\frac{I_{1} I_{2} d x d s}{z^{2}} \sin \phi \sin \alpha \times 10^{-7} \tag{E1}
\end{equation*}
$$

where

$$
\begin{array}{ll}
z & =\text { length of the line PG } \\
\phi & =\text { angle between conductor HA and the line } z \\
\alpha & =\text { angle between the normal to the plane HPA at } \\
& =\text { P and the conductor JD. }
\end{array}
$$

Now GE $=d s \sin \phi$
Also GE $=z d \phi$
$\sin \phi=\frac{z d \phi}{d s}$
but $\frac{k}{z}=\sin \phi$, so that $\frac{d s}{z^{2}}=\frac{d \phi}{k}$
Substituting in Equation E1 we get

$$
\begin{equation*}
\frac{I_{1} I_{2} d x d \phi}{k} \sin \phi \sin \alpha \times 10^{-7} \tag{E2}
\end{equation*}
$$



Figure E.6-Skewed conductors- $\beta<90^{\circ}$, $\mathbf{x}$ and $\mathbf{m}$ positive

If this expression is integrated with respect to $\phi$ from $\phi=\phi_{A}$ (where G is at A ) to $\phi=\phi_{H}$ (where G is at H ), this will give the force $d F$ on the element $d x$ at P due to the current $I_{2}$ in this element and the current $I_{1}$ in the whole of conductor HA.

Thus

$$
\begin{aligned}
& d F=\frac{I_{1} I_{2} d x 10^{-7} \sin \alpha}{k} \int_{\phi A}^{\phi H} \sin \phi d \phi \\
& d F=\frac{I_{1} I_{2} d x 10^{-7} \sin \alpha}{k}(-\cos \phi) \begin{array}{l}
\phi H \\
\phi A
\end{array}
\end{aligned}
$$

i.e., the force $F_{p}$ per meter at P is

$$
\begin{equation*}
\frac{d F}{d x}=\frac{I_{1} I_{2} 10^{-7} \sin \alpha}{k}\left(\cos \phi_{A}-\cos \phi_{H}\right) \tag{E3}
\end{equation*}
$$

From the geometry of Figure E. 6 it is easy to show that

$$
\begin{aligned}
& \cos \phi_{A}=\frac{l-x \cos \beta}{\sqrt{\left[k^{2}+(l-x \cos \beta)^{2}\right]}} \\
& \cos \phi_{H}=\frac{-(m+x \cos \beta)}{\sqrt{\left[k^{2}+(m+x \cos \beta)^{2}\right]}}
\end{aligned}
$$

where $k=\sqrt{\left(d^{2}+x^{2} \sin ^{2} \beta\right)}$

It can also be shown that

$$
\sin \alpha=\sqrt{\frac{d^{2} \cos ^{2} \beta+x^{2} \sin ^{2} \beta}{d^{2}+x^{2} \sin ^{2} \beta}}(\text { see E.10.1) }
$$

and substituting these values of $\cos \phi_{\mathrm{A}}, \cos \phi_{\mathrm{H}}, k$ and $\sin \alpha$ in Equation (E3), we obtain

$$
\begin{align*}
F_{\mathrm{p}} & =\frac{I_{1} I_{2} 10^{-7} \sqrt{\left(d^{2} \cos ^{2} \beta+x^{2} \sin ^{2} \beta\right)}}{d^{2}+x^{2} \sin ^{2} \beta}  \tag{E4}\\
& \times\left\{\frac{l-x \cos \beta}{\sqrt{\left[d^{2}+x^{2} \sin ^{2} \beta+(l-x \cos \beta)^{2}\right]}}\right. \\
& \left.+\frac{m+x \cos \beta}{\sqrt{\left[d^{2}+x^{2} \sin ^{2} \beta+(m+x \cos \beta)^{2}\right]}}\right\}
\end{align*}
$$

Figures E. 7 and E. 8 show that Equation (E4) applies also to cases in which $\beta$ lies between $90^{\circ}$ and $180^{\circ}$ and when the dimensions $m$ and $x$ are negative.


Figure E.7-Skewed conductors- $\beta>\mathbf{9 0}^{\circ}$, $\mathbf{x}$ positive, $m$ negative


Figure E.8-Skewed conductors- $\beta<90^{\circ}$, $\mathbf{x}$ negative, m positive

## E.5.2 Direction of forces

It has already been stated in the discussion of Figure E. 5 that the force per unit length at various points along the conductor JD acts at right angles to JD (i.e., radially) and that the force vector lies in the plane containing conductor HA and the point considered. The angle which the force vector makes with the plane in which conductor JD lies is given by

$$
\begin{equation*}
X=\tan ^{-1}\left(\frac{d}{x \tan \beta}\right) \text { (see E.10.2) } \tag{E5}
\end{equation*}
$$

so that the orthogonal components of $\mathrm{F}_{p}$ are

$$
\begin{align*}
& F_{h}=F_{p} \cos X  \tag{E6}\\
& F_{v}=F_{p} \sin X \tag{E7}
\end{align*}
$$

## E. 6 Numerical example

To illustrate the use of Equations (E4) through (E7), consider the arrangement shown in Figure E.9. Two conductors are shown 30 cm and 80 cm long, crossing each other obliquely and forming part of a complete circuit carrying a current of $10^{4} \mathrm{~A}$. The shortest distance between the two conductors is along a line 10 cm long joining the middle point of the 80 cm conductor to a point 10 cm from one end of the 30 cm conductor.

The distribution of forces along the 80 cm conductor, computed from Equation (E4), is shown in Figure E. 10 for six different angles of cross-over. It should be remembered that the forces acting on each differential length of conductor are uniplanar only in the cases for $\beta=0^{\circ}$ (parallel) and $\beta=90^{\circ}$ (right- angle cross-over) as shown by Equation (E5).


Figure E.9-Skewed conductors-numerical example


Figure E.10-Distribution of mechanical forces on skewed conductors for various angles of cross-over


Figure E.11-70 ${ }^{\circ}$ cross-over
Note the transition in form of the curves from $\beta=0^{\circ}$ to $\beta=90^{\circ}$ in Figure E.10, the minimum points at $x=0$ disappearing on curves where $\beta<45^{\circ}$.

The component forces for the $70^{\circ}$ cross-over (Figure E.11) are plotted in Figure E. 12 from Equations (E6) and (E7), and it is these curves which would be used in calculating the total force and moments by graphical integration. The magnitude and direction of forces on supporting insulators may be deduced easily from the moments of the component forces by methods which are fully detailed in Frick, [1].


Figure E.12-Orthogonal components of mechanical forces on conductors with $70^{\circ}$ cross-over

## E. 7 Special conductor arrangements

By suitable substitutions in Equation (E4), formulae may be obtained for the distribution of mechanical force in special conductor arrangements which agree with those already published.

## E.7.1 Parallel conductors ( $\beta=0^{\circ}$ )

a) Short conductors (see Figure E.13)
$F_{p}=\frac{I_{1} I_{2} 10^{-7}}{d}\left\{\frac{l-x}{\sqrt{\left[d^{2}+(l-x)^{2}\right]}}+\frac{m+x}{\sqrt{\left[d^{2}+(m+x)^{2}\right]}}\right\}$


Figure E.13-Short parallel conductors
b) End of long conductor (see Figure E.14)


Figure E.14-End of long parallel conductor

If $d$ and $x$ are very small compared with $l$, and $m=0$
then $\frac{l-x}{\sqrt{\left[d^{2}+(l-x)^{2}\right]}}=1$
and

$$
\frac{m+x}{\sqrt{\left[d^{2}+(m+x)^{2}\right]}}=\frac{x}{\sqrt{\left(d^{2}+x^{2}\right)}}
$$

Then from Equation (E4)

$$
\begin{equation*}
F_{p}=\frac{I_{1} I_{2} 10^{-7}}{d}\left[1+\frac{x}{\sqrt{\left(d^{2}+x^{2}\right)}}\right] \tag{E9}
\end{equation*}
$$

c) Centre of long conductors
$m=l, x=0$, and $d$ is negligible compared with $l$. Then from Equation (E4),

$$
\begin{equation*}
F_{p}=\frac{2 I_{1} I_{2} 10^{-7}}{d} \tag{E9a}
\end{equation*}
$$

## E.7.2 Right-angled cross-over (see Figure E.15) ( $\beta=90^{\circ}$ )

$F_{p}=\frac{I_{1} I_{2} x 10^{-7}}{d^{2}+x^{2}}\left[\frac{l}{\sqrt{\left(d^{2}+x^{2}+l^{2}\right)}}+\frac{m}{\sqrt{\left(d^{2}+x^{2}+m^{2}\right)}}\right]$


Figure E.15-90 ${ }^{\circ}$ cross-over

## E.7.3 Right-angled bend (see Figure E.16) ( $\beta=90^{\circ} \mathrm{d}=0 \mathrm{~m}=0$ )

From Equation (E10)

$$
\begin{equation*}
F_{p}=\frac{I_{1} I_{2} 10^{-7}}{x}\left[\frac{l}{\sqrt{\left(x^{2}+l^{2}\right)}}\right] \tag{E11}
\end{equation*}
$$




Figure E.16-90 bend

## E.7.4 Any angle in a plane (see Figures E.17 and E.18) (d=0)

$$
\begin{equation*}
F_{p}=\frac{I_{1} I_{2} 10^{-7}}{x \sin \beta}\left[\frac{l-x \cos \beta}{\sqrt{\left(x^{2}+l^{2}-2 l x \cos \beta\right)}}+\frac{m+x \cos \beta}{\sqrt{\left(x^{2}+m^{2}+2 m x \cos \beta\right)}}\right] \tag{E12}
\end{equation*}
$$

In Figure E. 18, $m$ is negative and $\beta$ is in the second quadrant, so that $\cos \beta$ is also negative.
When $\mathrm{m}=0$ and $\beta=135^{\circ}$, then $\cos \beta=\frac{-\sqrt{2}}{2}$ and $\sin \beta=\frac{\sqrt{2}}{2}$
Let $l=\frac{x}{v}$


Figure E.17-Any angle in a plane $<90^{\circ}$


Figure E.18-Any angle in a plane $\mathbf{>} \mathbf{9 0}^{\circ}, \mathrm{m}$ negative
Substituting in equation 12 we obtain

$$
\begin{equation*}
F_{p}=\frac{I_{1} I_{2} 10^{-7}}{x}\left[\frac{\sqrt{2}+v}{\sqrt{\left(v^{2}+\sqrt{2 v}+1\right)}}-1\right] \tag{E12a}
\end{equation*}
$$

which agrees with the formula given by Van Asperen [4].
Similarly when $\mathrm{m}=0$ and $\beta=45^{\circ}, \cos \beta=\sin \beta=\frac{\sqrt{2}}{2}$ and $l=\frac{x}{v}$, giving

$$
\begin{equation*}
F_{p}=\frac{I_{1} I_{2} 10^{-7}}{x}\left[\frac{\sqrt{2}-v}{\sqrt{\left(v^{2}-\sqrt{2 v}+1\right)}}+1\right] \tag{E12b}
\end{equation*}
$$

## E.7.5 Forces at bends and corners of a conductor system

The standard Equations (E11) and (E12) for angled conductors lying in a plane do not take into account the non-uniform current distribution occurring near the junction of the conductors. As $x \rightarrow 0$ the current in the bend tapers off with a corresponding reduction in the mechanical forces in the vicinity of the corner. The problem is outside the scope of the paper, but an approximate solution may be obtained for a $90^{\circ}$ bend by assuming that the force starts at the point $x=0.779 r$, where r is the radius of the conductor. (Frick, [1])

## E. 8 Conclusions

So far as the author is aware, the general formulae developed in the paper have not previously been stated. They should prove useful to designers in circumstances where accuracy is important. In other cases, where
approximate methods are appropriate, the rigid formulae may serve as a guide to the percentage error involved.

## E. 9 References

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[4] Van Asperen, C. H., "Mechanical forces on busbars under short circuit conditions," Ibid. 1922, 42, p. 1091.
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## E. 10 Appendixes

## E.10.1 To determine the angle between conductor JD and the direction of the magnetic flux (see Figure E.19)

Consider point P on conductor JD. The direction of the magnetic flux $\phi$ at this point due to the current $I_{1}$ in conductor HA is normal to the plane BPA and is indicated by the line PT. It is required to find the angle TPJ in terms of $x, d$ and $\beta$.

Rectangular co-ordinate axes PX, PY, and PZ with P as the origin are reproduced in Figure E.19a, in which PC and PR represent the directions of the conductor JD and flux vector PT, respectively. From a well-known proposition in co-ordinate geometry the cosine of the angle between two lines is equal to the sum of the products of their respective direction-cosines.


Figure E.19-Angle between conductor and direction of magnetic flux

Thus, if $a, b$ and $c$ are the direction-cosines of PC and $a^{\prime}, b^{\prime}$ and $c^{\prime}$ are the direction-cosines of PR, we have $\cos \alpha=a a^{\prime}+b b^{\prime}+c c^{\prime}$
where

$$
\begin{array}{ll}
a=\cos \mathrm{XPC}=\sin \beta & a^{\prime}=\cos \mathrm{XPR}=\cos \theta \\
b=\cos \mathrm{YPC}=\cos \beta & b^{\prime}=\cos \mathrm{YPR}=0 \\
c=\cos \mathrm{ZPC}=0 & \mathrm{c}^{\prime}=\cos \mathrm{ZPR}=\sin \theta
\end{array}
$$

Then

$$
\begin{equation*}
\cos \alpha=\sin \beta \cos \theta \tag{E13}
\end{equation*}
$$

Referring to Figure E. 19 it is seen that

$$
\cos \theta=\frac{d}{\sqrt{\left(d^{2}+x^{2} \sin ^{2} \beta\right)}}
$$

Therefore

$$
\begin{align*}
& \cos \alpha=\frac{d \sin \beta}{\sqrt{\left(d^{2}+x^{2} \sin ^{2} \beta\right)}} \\
& \sin \alpha=\sqrt{\left(1-\frac{d^{2} \sin ^{2} \beta}{d^{2}+x^{2} \sin ^{2} \beta}\right)} \\
& \sin \alpha=\sqrt{\left(\frac{d^{2} \cos ^{2} \beta+x^{2} \sin ^{2} \beta}{d^{2}+x^{2} \sin ^{2} \beta}\right)} \tag{E14}
\end{align*}
$$

## E.10.2 To determine the angle between the direction of the mechanical force on an element of conductor JD at point $P$ and the normal to the conductor in the plane CPA' (see Figure E.20).

The mechanical force on an element of conductor JD due to the currents $I_{1}$ in HA and $I_{2}$ in JD is at right angles to the direction of the magnetic flux at point P. It lies, therefore, in the plane BPA and is represented by the line PF. Produce PF to cut HA in T. Since the force PF is at right angles to conductor JD, its trace PS in the plane CPA' is also normal to JD.

Then

$$
\begin{equation*}
\tan X=\frac{S T}{P S}=\frac{d}{x \tan \beta} \tag{E15}
\end{equation*}
$$



Figure E.20-Direction of mechanical force

## Annex F

(informative)

## Static analysis of substation rigid-bus structure

## F. 1 Introduction

Clause 11 of this guide provides good guidance for determining maximum span lengths of single-level bus conductors based on the allowable vertical deflection or conductors' fiber stress. However, it fails to address a two-level bus arrangement, the most common rigid-bus arrangement in a low-profile substation, where one bus at a lower-level bus supports the upper-level bus with an A-frame. In such an arrangement, the forces acting on the upper bus are transmitted to the lower bus as concentrated forces at each base of the A-frame. The lower bus may be subjected to severe stress due to these concentrated loads. This annex is intended to provide a simple statistical method for analyzing two-level bus configurations. Since the concern here is the higher fiber stress that can be developed at the base of the A-frame, only the fiber stress aspect is addressed.

## F. 2 Basis of static analysis

The bus conductors are subjected to uniformly distributed forces (weight, wind force, and fault current force) in vertical and horizontal directions. Some concentrated forces are transmitted through the A-frame from an upper bus as well. When external forces act upon the bus, a reaction (force) is developed at the insulator supports. These forces also create bending moments along the bus conductor. The statistical analysis of the bus structures determines the values of these unknown force reactions and moments.

The basis of the analysis for a bus conductor are the static equilibrium equations. Equilibrium equations relate the forces acting on the bus conductor with reactions and the bending moments developed in the bus conductor, and the deformation equations. The deformation equations are required to supplement the equilibrium equations to solve the statistically independent continuous bus structures. Most buses are supported by three or more supports. They normally form indeterminate continuous beams. The deformation equation that is applicable here for the continuous bus with simple end conditions is the Three-Moment Theorem. It provides the relations of moments at three supports of two adjacent spans of a continuous beam. (See Figure F.1.)


Figure F.1-Two adjacent spans of continuous beam

Using the Three-Moment Theorem and Figure F.1, the following equation can be derived:

$$
\begin{equation*}
M_{1} L_{1}+2 M_{2}\left(L_{1}+L_{2}\right)+M_{3} L_{2}=-\left[\frac{w\left(L_{1}^{3}+L_{2}^{3}\right)}{4}+\frac{\sum\left(P_{1} a_{1}\left(L_{1}{ }^{2}-a_{1}^{2}\right)\right.}{L_{1}}+\frac{\sum\left(P_{2} a_{2}\left(L_{2}{ }^{2}-a_{2}^{2}\right)\right.}{L_{2}}\right] \tag{F1}
\end{equation*}
$$

where
$M_{1}, M_{2}, M_{3}$ are moments at supports $1,2,3 \mathrm{in} \mathrm{lbf} / \mathrm{ft}$
$L_{1}, L_{2}$ are span lengths of two adjacent spans in ft
$w$ is uniformly distributed loads in $\mathrm{lbf} / \mathrm{ft}$
$a_{1}$ is the distance of concentrated $\operatorname{load}(\mathrm{s}) P_{1}$ from insulator 1 in feet
$b_{2}$ is the distance of concentrated $\operatorname{load}(\mathrm{s}) P_{2}$ from insulator 3 in feet

To apply the Three-Moment Theorem, two moments at the end supports must be known. Normally the continuous bus is assumed to be pinned at the ends; thus, the end moments are zero. When the bus extends beyond the end support, the moment become non-zero due to cantilever bus section. In this case the end moment can be solved using the moment equation for the cantilever section. Three-Moment Theorem cannot be applied to bus configurations with fixed ends of unknown moments.

The step-by-step procedure for analyzing the bus conductor using the Three-Moment Theorem and the equilibrium equations is described in F.3. Samples of generalized equations are given in F. 5 (for buses without concentrated loads) and F. 6 (for buses with concentrated loads) for analyzing some common bus configurations.

In applying these procedures, the sign conventions for the forces and bending moments are shown in Table F.1.

Table F.1-Sign convention for the applied forces and bending moments

| Force Direction | Sign |
| :--- | :--- |
| Vertical downward force in -Y-axis | Positive |
| Vertical uplift force in +Y-axis | Negative |
| Horizontal force in -X-axis | Positive |
| Horizontal force in + X-axis | Negative |
| Horizontal force in -Z-axis | Positive |
| Horizontal force in + Z-axis | Negative |
| Bending moments due to positive force | Negative |
| Bending moments due to negative force | Positive |

Note that the bus with negative bending moments in the vertical direction has convex upward curvature at the point of a moment. Since the external weight, wind, and short-circuit forces are in X, Y, and Z directions, the analysis should be performed separately in each direction and combined vectorially to obtain resultant values.

## F. 3 Step-by-step method

The structural analysis of bus conductor can be performed in three stages. The first stage is to determine the maximum allowable span length of the bus based on either vertical deflection or fiber stress using the method outlined in this guide. The second and third stages are for analyzing more complex bus configurations involving double-level bus arrangements. The second stage analyzes the upper-level bus and calculates the values of the concentrated forces of the upper bus that are transmitted to the lower-level bus.

## F.3.1 Stage 1 calculations for general bus structure

a) Lay out complete bus arrangement including main buses and feeder buses in each bay.
b) Determine bus-conductor sizes and their characteristics for each bus configuration.
c) Establish design parameters for each bus configuration (span lengths, spacing, A-frame height, A-frame base width, wind velocity, and short-circuit current).
d) Calculate conductor gravitational forces (weights of conductor, damping material, and ice according to Clause 8 of this guide. Sum the gravitation unit forces [Equation (13) of this guide] in vertical direction of each bus configuration.
e) Determine maximum allowable span length of the bus $L_{d}$ for a given vertical deflection using one of the applicable equations given in 11.1 of this guide [Equations (14) through (21)] for each bus configuration.
f) Calculate conductor wind forces [Equation (9)] in horizontal (X-axis or Z-axis) per Clause 9 of this guide for each bus configuration.
g) Calculate conductor short-circuit forces [Equation (12)] in horizontal (X-axis or Z-axis) according to Clause 10 of this guide for each bus configuration.
h) Combine the vertical forces and the horizontal forces, and get the resultant force [Equation (22)] for each bus configuration.
i) Find the allowable span length of the bus $L_{S}$ based on the fiber stress using one of the applicable equations given in 11.2 of this guide [Equations (23) through (29)] for each bus configuration.
j) Check that the longest span utilized in the bus configuration is less than the calculated maximum allowable span length of the bus $\left(L_{A}\right)$. If not, change the bus configuration and repeat the above steps for the new configuration.
k) For single-level bus arrangement without concentrated loads, the analysis ends here. For two-level bus using A-frame supports proceed to the second stage.

## F.3.2 Stage 2 calculations for upper-level bus structure

The following steps are required to determine the force for the upper-level bus:
a) Find the per unit total vertical and horizontal (X-axis) direction. See Figure F.2.
b) Calculate the moment at each support (A-frame or insulator support) by writing the equation for Three-Moment theorem [Equation (F1)] consecutively for each two adjacent spans and by solving the simultaneous equations. Moments should be calculated separately for vertical and horizontal (X-axis) direction. See Figure F. 2


Figure F.2-Stage 2 calculations

$$
\begin{aligned}
& M_{1} L_{1}+2 M_{2}\left(L_{1}+L_{2}\right)+M_{3} L_{2}=\frac{-w\left(L_{1}{ }^{3}+L_{2}{ }^{3}\right)}{4} \\
& M_{2} L_{2}+2 M_{3}\left(L_{2}+L_{3}\right)+M_{4} L_{3}=\frac{-w\left(L_{2}{ }^{3}+L_{3}{ }^{3}\right)}{4} \\
& M_{n-2} L_{n-2}+2 M_{n-1}\left(L_{n-2}+L_{n-1}\right)+M_{n} L_{n-1}=\frac{-w\left(L^{3}{ }_{n-2}+L^{3}{ }_{n-1}\right)}{4} \\
& M_{n-1} L_{n-1}+2 M_{n}\left(L_{n-1}+L_{n}\right)+M_{n+1} L_{n}=\frac{-w\left(L_{n-1}{ }^{3}+L_{n}{ }^{3}\right)}{4}
\end{aligned}
$$

There will be $n-1$ equations for bus with $n$ sections and $n+1$ supports. Normally end moments $M_{1}$ and $M_{n+1}$ are zero for pinned supports or can be easily figured out for continuous cantilever supports.
c) Determine vertical and horizontal reactions at each support by solving moment equilibrium Equation (F2):

Moment at a point $=$ moments due to distributed loads plus moments due to concentrated loads.

$$
\begin{equation*}
=\frac{w L^{2}}{2}+\sum P x \tag{F2}
\end{equation*}
$$

where
$\frac{w L^{2}}{2}=$ moment due to distributed load
$P x=$ moments due to concentrated load
where
$w=$ Generalized distributed load in one direction in lbf per ft
$P=$ Generalized concentrated load(s) in same direction in lbf
$L=$ Span length of the bus conductor on one side of moment point in ft
$x=$ Distance of concentrated load(s) from moment point in ft

For example write moment at second support to solve the reaction $R_{1}$ as follows:

$$
M_{2}=-\left(\frac{w L_{1}{ }^{2}}{2}+R_{1} L_{1}\right) \quad R_{1}=\frac{\left(M_{2}+w L_{1}{ }^{2} / 2\right)}{L_{1}}
$$

d) Calculate moments at other points preferably midpoints of the bus using the same moment equilibrium to determine the location of maximum and minimum moments along the bus conductor. These calculations may be needed to locate the welds at the minimum stress points.
$R m=\sqrt{V m^{2}+H m^{2}}$
where
$R m=$ Resultant moment
Vm $=$ Vertical moment
$H m=$ Horizontal moment
e) Combine vertical and horizontal moments and get bending moments for upper bus [Equation (F3)].
f) Find maximum resultant moment and determine the maximum fiber stress [Equation (F4)] at that point. The maximum stress normally occurs at the second last support.

Maximum fiber stress $=$ Max. moment $\times 12 / S$
where
$S$ is the section modulus $\mathrm{in}^{3}$.
g) Check that the calculated fiber stress is less than the maximum allowable stress (minimum yield strength of bus material).
h) Determine the vertical (Y-axis) and horizontal (X-axis) concentrated forces that will be transmitted to low bus at a particular A-frame support (See Figure F.3).


Figure F.3-Upper-level bus

$$
\begin{align*}
& Y_{1}=\frac{F_{v}}{2}+\left(\frac{F_{h}}{2}\right)+\left(\frac{H_{A}}{L_{A}}\right)  \tag{F5}\\
& Y_{2}=\frac{F_{v}}{2}+\left(\frac{F_{h}}{2}\right)+\left(\frac{H_{A}}{L_{A}}\right) \tag{F6}
\end{align*}
$$

where
$Y_{1}=$ concentrated vertical force at base one of A-frame
$Y_{2}=$ concentrated vertical force at base two of A-frame
$X_{1}=\frac{F_{h}}{2}+\left(\frac{F_{v}}{2}\right)+\left(\frac{L_{A}}{H_{A}}\right)$
$X_{2}=\frac{F_{h}}{2}+\left(\frac{F_{v}}{2}\right)+\left(\frac{L_{A}}{H_{A}}\right)$
$X_{1}=$ concentrated horizontal force at base one of A-frame
$X_{2}=$ concentrated horizontal force at base two of A-frame
where

$$
\begin{aligned}
F_{v} & =\text { vertical force } \\
& =- \text { vertical reaction at the top of A-frame } \\
F_{h} & =\text { horizontal force } \\
& =- \text { horizontal reaction at the top of A-frame } \\
H_{A} & =\text { height of A-frame in } \mathrm{ft} \\
L_{A} & =\text { half of base of A-frame in } \mathrm{ft}
\end{aligned}
$$

i) Calculation for high bus ends here. Proceed to F.3.3 for lower bus structure.

## F.3.3 Stage 3 calculations for lower-level bus structure

The following steps are required to determine the forces for the lower-level bus:
a) Start with unit total vertical and horizontal forces for lower-level bus. The horizontal force should be in the Z-axis direction.
b) Consider the vertical forces transmitted from high bus to each base of the A-frame [Equations (F5) and (F6)] as concentrated load.
c) Calculate the moment at each insulator support by writing equations for Three-Moment Theorem [Equation (F1)] consecutively for each two adjacent spans and by solving the simultaneous equations. Moments should be calculated separately for vertical and horizontal (Z-axis) directions. Moment calculation for the horizontal direction is identical to step b) of F.3.2. However, the moment equation in the vertical direction for a span involving an A-frame would include terms for concentrated vertical loads. For example moment equations for bus structure with A-frame resting on the second and third spans (A-frame over insulator no. 3 see Figure F.4) will be as follows:


Figure F.4-Lower-level bus

$$
\begin{aligned}
& M_{1} L_{1}+2 M_{2}\left(L_{1}+L_{2}\right)+M_{3} L_{2}=-\left[\frac{w\left(L_{1}{ }^{3}+L_{2}{ }^{3}\right)}{4}+\frac{Y_{1} L_{A}\left(L_{2}{ }^{2}-L_{A}{ }^{2}\right)}{L_{2}}\right] \\
& M_{2} L_{2}+2 M_{3}\left(L_{2}+L_{3}\right)+M_{4} L_{3}=-\left[\frac{w\left(L_{2}{ }^{3}+L_{3}{ }^{3}\right)}{4}+\frac{Y_{1}\left(L_{2}-L_{A}\right)\left(L_{2}{ }^{2}-L_{A}{ }^{2}\right)}{L_{2}}\right. \\
& \left.+\frac{Y_{2}\left(L_{3}-L_{A}\right)\left(L_{3}{ }^{2}-\left(L_{3}-L_{A}\right)^{2}\right.}{L_{3}}\right] \\
& M_{3} L_{3}+2 M_{4}\left(L_{3}+L_{4}\right)+M_{5} L_{4}=-\left[\frac{w\left(L_{3}{ }^{3}+L_{4}{ }^{3}\right)}{4}+\frac{Y_{2} L_{A}\left(L_{3}{ }^{2}-L_{A}{ }^{2}\right)}{L_{3}}\right] \\
& M_{n-1} L_{n-1}+2 M_{n}\left(L_{n-1}+L_{n}\right)+M_{n+1} L_{n}=-\frac{w\left(L_{n-2}{ }^{3}+L_{n-1}{ }^{3}\right)}{4}
\end{aligned}
$$

where $Y_{1}$ and $Y_{2}$ are vertical forces transmitted at the base of A-frame and $L_{A}$ is the distance of the base number one and two from support number three (3) (half of A-frame base).

Normally end moments $M_{1}$ and $M_{n}$ are zero for pinned supports or can be easily calculated for continuous cantilever supports.
d) Calculate vertical and horizontal reactions at each insulator support using moment equilibrium equations, similar to step c) of F.3.2.
e) Calculate moments at other points preferably midpoints of the bus spans and base of each A-frame using the same moment equilibrium equations to determine the point of maximum and minimum moments.
f) Combine vertical and horizontal moments and get resultant bending moments for lower bus [Equation (F3)].
g) Find maximum resultant moment among all moments and determine the maximum fiber stress [Equation (F4)]. The maximum moment will usually be developed at one of the bases of the A-frame.
h) Check that the calculated maximum fiber stress is less than the maximum allowable stress (minimum yield strength of the bus material). The maximum fiber stress will occur at the base of A-frame where the frame is welded. Determination must be made whether to consider the effects of welding and to reduce the maximum yield strength at the welding point.
i) This is the end of bus strength calculations. Proceed to calculate the insulator cantilever requirements as outlined in F.4.

## F. 4 Insulator cantilever forces

Clause 12 of this guide provides simplified equations for calculating insulator cantilever forces as a function of effective bus span length as given in Table 5. The effective span length as defined in Table 5 of this guide is applicable only to equal span bus length and concentrated loads. The bus analysis should be made according to the procedures described in F. 3 above to determine the horizontal reactions.

The horizontal reactions calculated are the bus forces on the insulators. The span lengths are already accounted for in the reaction calculations made in step B3 or C4. These bus forces do not account for the overload factors given in the guide. Since the calculation is tedious, this method of adjusting the bus reactions is preferred. The adjustment factor is

$$
\begin{equation*}
K_{a}=\frac{\left(K_{1} F_{w} K_{2} F_{S C}\right)}{\left(F_{S C}+F_{W}\right)} \tag{F9}
\end{equation*}
$$

where

$$
\begin{aligned}
& K_{1}=\text { overload factor for wind force } \\
& K_{2}=\text { overload factor for short circuit force } \\
& F_{W}=\text { unit wind force } \\
& F_{S C}=\text { unit short circuit force }
\end{aligned}
$$

The complete equation [Equation (35)] for the total cantilever load on a vertically mounted insulator supporting a horizontal bus becomes

$$
\begin{equation*}
F_{I S}=\frac{K_{1} F_{W I}}{2}+\frac{K_{a} R_{i}\left(H_{i} H_{F}\right)}{H_{1}} \tag{F10}
\end{equation*}
$$

where
$F_{W I}=$ wind force on insulator
$R_{i}=$ adjusted horizontal bus reaction for a support point (see Figure F.11)
$H_{i}=$ insulator height in inches
$H_{f}=$ bus center height above insulator in inches
$K_{a}=$ adjustment factor for overload

When calculating the insulator cantilever strength for low bus in double bus arrangement, the concentrated horizontal force transmitted from the upper bus (in X-axis) should be included in the horizontal bus reaction $R_{i}$. The resultant $R_{i}$ can be calculated using Equation (F11). The X-axis force is assumed to be divided among the low-bus insulators.

$$
\begin{equation*}
R_{i}=\sqrt{\left[\frac{\left(X_{1}+X_{2}\right)}{N}\right]^{2}+R^{2}} \tag{F11}
\end{equation*}
$$

where
$X_{1}$ and $X_{2}=$ concentrated horizontal forces [Equations (F7) and (F8)] from upper bus
$N=$ number of insulators in low bus
$R=$ calculated horizontal bus reaction for a support point
$R_{i}=$ adjusted horizontal bus reaction for a support point

## F. 5 Examples for common bus configurations without concentrated load

## F.5.1 Single-span bus, two pinned supports



Figure F.5-Single-span bus, two pinned supports

$$
\begin{aligned}
& M_{1}=0 \\
& M_{2}=0 \\
& R_{1}=-\frac{(w L)}{2} \quad R_{1}=-\frac{(w L)}{2} \quad M_{\max }=-\frac{\left(w L^{2}\right)}{8 S} \quad F S_{\max }=-\frac{\left(w L^{2}\right)}{8 S}
\end{aligned}
$$

where
$w=$ generalized distributed unit force in lbf per ft
$L=$ span length in feet
$M_{1}$ and $M_{2}=$ moments at supports 1 and 2 in lbf
$R_{1}$ and $R_{2}=$ reactions at supports 1 and 2 in lbf
$S=$ section modulus of bus conductor
$F S=$ fiber stress in lbf per in ${ }^{3}$
F.5.2 Single-span bus with cantilever on one side, one pinned and one continuous support


Figure F.6-Single-span bus, one pinned and one continuous support

$$
\begin{array}{ll}
M_{1}=-\frac{\left(w L_{c}{ }^{2}\right)}{2} & M_{2}=0 \\
R_{1}=-\left(\frac{w\left(L+L_{c}\right)^{2}}{2 L}\right) & R_{2}=-\left(\frac{w\left(L^{2}+L_{c}^{2}\right)}{2 L}\right) \quad M_{m i d}=\frac{w L^{2}}{8}-\frac{w L_{c}^{2}}{4}
\end{array}
$$

$$
\begin{aligned}
& M_{\max }=\frac{w L^{2}}{8}-\frac{w L_{c}{ }^{2}}{4} \quad \text { at midpoint of the bus span } \\
& M_{\max }=-\frac{w L_{c}{ }^{2}}{2} \quad \text { at support } 1 \\
& F S_{\max }=\frac{\left(12 M_{\max }\right)}{S}
\end{aligned}
$$

where
$w=$ generalized distributed unit force in lbf per ft
$L=$ span length in feet
$L_{c}=$ protruding length of cantilever
$M_{1}$ and $M_{2}=$ moments at supports 1 and 2 in lbf
$R_{1}$ and $R_{2}=$ reactions at supports 1 and 2 in lbf
$S=$ section modulus of bus conductor
$F S=$ fiber stress in lbf per in ${ }^{2}$

## F.5.3 Single-span bus, two fixed supports



Figure F.7-Single-span bus, two fixed supports

$$
\begin{aligned}
& M_{1}=-\left(\frac{w L^{2}}{12}\right) \quad M_{2}=-\left(\frac{w L^{2}}{12}\right) \quad M_{\text {mid }}=\frac{w L^{2}}{24} \\
& R_{1}=-\left(\frac{w L}{2}\right) \quad R_{2}=-\left(\frac{w L}{2}\right) \quad M_{\max }=\frac{w L^{2}}{12} \text { at end support of the bus support } \\
& F S_{\max }=\frac{w L^{2}}{S}
\end{aligned}
$$

where
$w=$ generalized distributed unit force in lbf per ft
$L=$ span length in feet
$M_{1}$ and $M_{2}=$ moments at supports 1 and 2 in lbf
$R_{1}$ and $R_{2}=$ reactions at supports 1 and 2 in lbf
$S=$ section modulus of bus conductor
$F S=$ fiber stress in lbf per in ${ }^{2}$

## F.5.4 Continuous two-span bus, two pinned and one continuous support



Figure F.8-Continuous two-span, two-pinned and one continuous support

$$
\begin{aligned}
& M_{1}=0 \quad M_{2}=-\frac{w\left(L_{1}{ }^{3}+L_{2}{ }^{3}\right) 2\left(L_{1}+L_{2}\right)}{4} \quad M_{3}=0 \\
& R_{1}=-\frac{M_{2}+\frac{w L_{1}{ }^{2}}{2}}{L_{1}} \quad R_{2}=-\left[\frac{w\left(L_{1}+L_{2}\right)^{2}}{2}+\frac{R_{1}\left(L_{1}+L_{2}\right)}{L_{2}}\right] R_{3}=-\frac{M_{2}+\frac{w L_{2}{ }^{2}}{2}}{L_{2}} \\
& M_{\operatorname{mid} 1}=-\left[\frac{w\left(\frac{L_{1}}{2}\right)^{2}}{2}+R_{1}\left(\frac{L_{1}}{2}\right)\right] M_{\operatorname{mid} 2}=-\left[\frac{w\left(\frac{L_{2}}{2}\right)^{2}}{2}+R_{3}\left(\frac{L_{2}}{2}\right)\right] \\
& F S_{\max }=\frac{\left(12 M_{\max }\right)}{S} \quad M_{\max }=-\frac{w\left(L_{1}{ }^{3}+L_{2}{ }^{3}\right) 2\left(L_{1}+L_{2}\right)}{4} \text { at mid-support }
\end{aligned}
$$

where
$w=$ generalized distributed unit force in lbf per ft
$L_{1}$ and $L_{2}=$ span lengths in feet
$M_{1}, M_{2}$ and $M_{3}=$ moments at supports 1,2 , and 3 in lbf
$R_{1}, R_{2}$ and $R_{3}=$ reactions at supports 1,2 , and 3 in lbf
$S=$ section modulus of bus conductor
$F S=$ fiber stress in lbf per in ${ }^{2}$

For $L_{1}=L_{2}$ :

$$
\begin{aligned}
& M_{1}=0 \quad M_{2}=-\left(\frac{w L^{2}}{8}\right) \quad M_{3}=0 \\
& R_{1}=-\left(\frac{3}{8}\right) w L \quad R_{2}=-\left(\frac{5}{4}\right) w L \quad R_{3}=-\left(\frac{3}{8}\right) w L
\end{aligned}
$$

## F.5.5 Continuous three-span bus, two pinned and two continuous supports



Figure F.9-Continuous three-span bus, two pinned and two continuous supports

$$
\begin{aligned}
& M_{1}=0 M_{4}=0 \\
& M_{2}=\left[2\left(L_{1}+L_{2}\right)\right]+M_{3} L_{2}=-\frac{w\left(L_{1}{ }^{3}+L_{2}{ }^{3}\right)}{4} \quad M_{2} L_{2}+M_{3}\left[2\left(L_{2}+L_{3}\right)\right]=-\frac{w\left(L_{2}{ }^{3}+L_{3}{ }^{3}\right)}{4} \\
& R_{1}=-\left(M_{2}+\frac{w L_{1}{ }^{2}}{2}\right) L_{1} \quad R_{2}=-\frac{\left[M_{3}+\frac{w\left(L_{1}+L_{2}\right)^{2}}{2}+R_{1}\left(L_{1}+L_{2}\right)\right]}{L_{2}} \\
& R_{3}=-\frac{\left[M_{4}+\frac{w\left(L_{1}+L_{2}+L_{3}\right)^{2}}{2}+R_{1}\left(L_{1}+L_{2}+L_{3}\right)+R_{2}\left(L_{2}+L_{3}\right)\right]}{L_{3}} \\
& \left.R_{4}=-\left(M_{3}+\frac{w L_{3}{ }^{2}}{2}\right){L_{3}}^{2}\right] \\
& M_{\text {mid }}=-\left[\frac{w\left(\frac{L_{1}}{2}\right)^{2}}{2}+\frac{R_{1} L_{1}}{2}\right] \quad M_{\text {mid } 2}=-\left[\frac{w\left(L_{1}+\frac{L_{2}}{2}\right)^{2}}{2}+R_{1}\left(L_{1}+\frac{L_{2}}{2}\right)+\frac{R_{2} L_{2}}{2}\right] \\
& M_{\text {mid }}=-\left[\frac{w\left(\frac{L_{2}}{2}\right)^{2}}{2}+\frac{R_{3} L_{2}}{2}\right] \quad M_{\text {max }}=M_{2} \text { or } M_{2} \text { at second or third support }
\end{aligned}
$$

where
$w=$ generalized distributed unit force in lbf per ft
$L_{1}, L_{2}$, and $L_{3}=$ span lengths in feet
$M_{1}, M_{2}, M_{3}$, and $M_{4}=$ moments at supports $1,2,3$, and 4 in lbf
$R_{1}, R_{2}, R_{3}$, and $R_{4}=$ reactions at supports $1,2,3$, and 4 in lbf
$S=$ section modulus of bus conductor
$F S=$ fiber stress in lbf per in ${ }^{2}$

For $L_{1}=L_{2}=L_{3}=L$ (for three equal spans)

$$
\begin{aligned}
& M_{1}=0 \quad M_{2}=-\left(\frac{w L^{2}}{10}\right) \quad M_{3}=-\left(\frac{w L^{2}}{10}\right) \quad M_{4}=0 \\
& R_{1}=-\left(\frac{4}{10}\right) w L \quad R_{2}=-\left(\frac{11}{10}\right) w L \quad R_{3}=-\left(\frac{11}{10}\right) w L \quad R_{4}=-\left(\frac{4}{10}\right) w L \\
& M_{\max }=M_{2} \text { or } M_{3}=-w L^{2} / 10 \\
& F S_{\max }=\frac{-12\left(w L^{2}\right)}{10 S}
\end{aligned}
$$

## F.5.6 Continuous four equal spans, two pinned and three fixed supports



Figure F.10-Continuous four spans, two pinned and three fixed supports

$$
\begin{aligned}
& M_{1}=0 \quad M_{5}=0 \\
& M_{2}=-\left[\frac{\frac{w\left(L^{3}+L^{3}\right)}{4}+M_{3} L}{2(L+L)}\right]=-\left(\frac{3}{28}\right) w L^{2} \quad M_{3}=-\left(\frac{2}{28}\right) w L^{2} \quad M_{4}=-\left(\frac{2}{28}\right) w L^{2} \\
& R_{1}=-\left(\frac{11}{28}\right) w L^{2} \quad R_{2}=-\left(\frac{32}{28}\right) w L^{2} \quad R_{3}=-\left(\frac{26}{28}\right) w L^{2} \quad R_{4}=-\left(\frac{32}{28}\right) w L^{2} \\
& R_{5}=-\left(\frac{32}{28}\right) w L^{2} \quad M_{\max }=M_{2} \text { or } M_{4}=\left(\frac{3}{28}\right) w L^{2} \text { at second or fourth support } \quad F S_{\max }=\left(\frac{36}{28 S}\right) w L^{2}
\end{aligned}
$$

## F. 6 Example for common bus configuration with concentrated load

F.6.1 Single-span bus with A-frame on cantilever side, one pinned and one continuous support


Figure F.11-Single-span bus with A-frame one pinned and one continuous support

$$
\begin{aligned}
& M_{1}=-\left[\frac{w L_{C}{ }^{2}}{2}+Y_{1} L_{A}\right] \quad R_{1}=-\frac{\left[\frac{w L_{1}+L_{C}{ }^{2}}{2}+Y_{1}\left(L_{A}+L\right)+Y_{2}\left(L-L_{A}\right)\right]}{L} \\
& M_{2}=0 \quad R_{2}=-\frac{\left[M_{1}+\frac{w L^{2}}{2}+Y_{1} L_{A}\right]}{L} \\
& M_{\operatorname{mid}}=-\left[\frac{w L^{2}}{8}+\frac{R_{2} L}{2}\right] \quad M_{Y 1}=-\frac{w\left(L_{C}+L_{A}\right)^{2}}{2} \quad M_{Y 2}=-\left[\frac{w\left(L_{C}+L_{A}\right)^{2}}{2}+R_{1} L_{A}+2 Y_{1} L_{A}\right] \\
& M_{\max }=M_{y 2} \quad F S_{\max }=\frac{12 M_{\max }}{S}
\end{aligned}
$$

where
$w=$ generalized distributed unit force in lbf per ft
$L=$ span lengths in feet
$L_{A}=$ length of half base of A-frame in ft
$L_{C}=$ protruding length of cantilever
$M_{1}, M_{2}=$ moments at supports 1,2 in lbf ft
$R_{1}, R_{2}=$ reactions at supports 12 , in lbf ft
$S=$ section modulus of bus conductor
$F S=$ fiber stress in lbf per in ${ }^{2}$

## F.6.2 Two spans with A-frame on cantilever side, one pinned and two fixed supports



Figure F.12-Two spans with A-frame, one pinned and two fixed supports

$$
\begin{aligned}
& M_{1} L_{1}+M_{2}\left[2\left(L_{1}+L_{2}\right)\right]=-\left[\frac{w L_{1}+L_{2}}{4}+\frac{Y_{2} L_{A}\left(L_{1}{ }^{2}-L_{A}{ }^{2}\right)}{L_{1}}\right] \quad M_{1}=-\left[\frac{w L_{C}{ }^{2}}{2}+Y_{1} L_{A}\right] \quad M_{3}=0 \\
& R_{1}=-\left[\frac{M_{2}+\frac{w\left(L_{1}+L_{C}\right)^{2}}{2}+Y_{1}\left(L_{A}+L_{1}\right)+Y_{2}\left(L_{1}-L_{A}\right)}{L_{1}}\right] \\
& R_{2}=-\frac{\left[\frac{w\left(L_{1}+L_{2}+L_{C}\right)^{2}}{2}+R_{1}\left(L_{1}+L_{2}\right)+Y_{1}\left(L_{A}+L_{1}+L_{2}\right)+Y_{2}\left(L_{1}+L_{2}-L_{A}\right)\right]}{L_{2}} \\
& R_{3}=-\left(M_{2}+\frac{w L_{2}^{2}}{2}\right) L_{2} \\
& M_{\operatorname{mid1}}=-\left[\frac{w\left(\frac{L_{1}}{2}+L_{C}\right)^{2}}{2}+R_{1}\left(\frac{L_{1}}{2}\right)+Y_{1}\left(L_{A}+\frac{L_{1}}{2}\right)+Y_{2}\left(\frac{L_{1}}{2}-L_{A}\right)\right] \quad M_{\operatorname{mid} 2}=-\left[\frac{w\left(\frac{L_{2}}{2}\right)^{2}}{2}+R_{3} \frac{L_{2}}{2}\right] \\
& M_{Y 1}=-\left[\frac{w\left(L_{C}-L_{A}\right)^{2}}{2}\right] \quad M_{Y_{2}}=-\left[\frac{w\left(L_{C}+L_{A}\right)^{2}}{2}+R_{1} L_{A}+Y_{1}\left(2 L_{A}\right)\right] \quad M_{\max }=M_{Y_{2}} \\
& F S_{\max }=\frac{12 M_{\max }}{S}
\end{aligned}
$$

where
$w=$ generalized distributed unit force in lbf per ft
$L_{1}$ and $L_{2}=$ span lengths of two spans in feet
$L_{A}=$ length of half base of A-frame in ft
$L_{C}=$ protruding length of cantilever
$M_{1}, M_{2}$, and $M_{3}=$ moments at supports 1,2 , and 3 in lbf ft
$R_{1}, R_{2}$ and $R_{3}=$ reactions at supports 1,2 , and 3 in lbf ft
$S=$ section modulus of bus conductor
$F S=$ fiber stress in lbf per in ${ }^{2}$

## F.6.3 Two-span bus with A-frame on mid-support-two pinned and one continuous support



Figure F.13-Two span bus with A-frame, two pinned and one continuous support

$$
\begin{aligned}
& M_{1}=\begin{aligned}
& 0 M_{3}=0 \quad M_{2}\left(2\left(L_{1}+L_{2}\right)\right)=-\left[\frac{w\left(L_{1}{ }^{3}+L_{2}{ }^{3}\right)}{4}+Y_{1}\left(L_{1}-L_{A}\right)\left(L_{1}{ }^{2}-\frac{\left(L_{1}-L_{A}\right)^{2}}{L_{1}}\right)\right. \\
&\left.+Y_{2}\left(L_{2}-L_{A}\right)\left(L_{2}{ }^{2}-\frac{\left(L_{2}-L_{A}\right)^{2}}{L_{2}}\right)\right] \\
& R_{1}=-\frac{\left[M_{2}+w\left(\frac{L_{1}{ }^{2}}{2}\right)+Y_{1} L_{A}\right]}{L_{1}} \quad R_{2}=-\frac{\left[\frac{w\left(L_{1}+L_{2}\right)^{2}}{2}+R_{1}\left(L_{1}+L_{2}\right)+Y_{1}\left(L_{A}+L_{2}\right)+Y_{2}\left(L_{2}-L_{A}\right)\right]}{L_{2}} \\
& R_{3}=-\frac{\left[M_{2}+w\left(\frac{L_{2}{ }^{2}}{2}\right)+Y_{2} L_{A}\right]}{L_{2}} \quad M_{\text {mid1 }}=-\left[\frac{w\left(\frac{L_{1}}{2}\right)^{2}}{2}+R_{1}\left(\frac{L_{1}}{2}\right)\right] M_{\operatorname{mid} 2}=-\left[\frac{w\left(\frac{L_{2}}{2}\right)^{2}}{2}+R_{3}\left(\frac{L_{2}}{2}\right)\right] \\
& M_{Y 1}=-\left[\frac{w\left(L_{1}-L_{A}\right)^{2}}{2}+R_{1}\left(L_{1}+L_{A}\right)\right] \quad M_{Y 2}=-\left[\frac{w\left(L_{2}-L_{A}\right)^{2}}{2}+R_{3}\left(L_{2}-L_{A}\right)\right] \\
& M_{\max }=\max . \operatorname{among} \text { all moments }=M_{Y 1} \text { or } M_{Y 2} \\
& F S_{\max }=\frac{\left(12 M_{\text {max }}\right)}{S}
\end{aligned}, l
\end{aligned}
$$

where
$w=$ generalized distributed unit force in lbf per ft
$L_{1}$ and $L_{2}=$ span lengths of two spans in feet
$L_{A}=$ length of half base of A-frame in ft
$L_{C}=$ protruding length of cantilever
$Y_{1}$ and $Y_{2}=$ concentrated forces transmitted from upper bus
$M_{1}, M_{2}$, and $M_{3}=$ moments at supports 1,2 , and 3 in lbf ft
$R_{1}, R_{2}$, and $R_{3}=$ reactions at supports 1,2 , and 3 in lbf ft
$S=$ section modulus of bus conductor
$F S=$ fiber stress in lbf per in ${ }^{2}$


[^0]:    The Institute of Electrical and Electronics Engineers, Inc.
    345 East 47th Street, New York, NY 10017-2394, USA

[^1]:    ${ }^{1}$ ANSI publications are available from the Sales Department, American National Standards Institute, 11 West 42nd Street, 13th Floor, New York, NY 10036, USA.

[^2]:    ${ }^{2}$ ASCE publications are available from the American Society of Civil Engineers, 1801 Alexander Bell Drive, Reston, VA 20191-4400, USA.
    ${ }^{3}$ ASTM publications are available from the American Society for Testing and Materials, 100 Barr Harbor Drive, West Conshohocken, PA 19428-2959, USA.
    ${ }^{4}$ IEEE publications are available from the Institute of Electrical and Electronics Engineers, 445 Hoes Lane, P.O. Box 1331, Piscataway, NJ 08855-1331, USA.
    ${ }^{5}$ NEMA publications are available from the National Electrical Manufacturers Association, 1300 N. 17th St., Ste. 1847, Rosslyn, VA 22209, USA.
    ${ }^{6}$ NFPA publications are available from Publications Sales, National Fire Protection Association, 1 Batterymarch Park, P.O. Box 9101, Quincy, MA 02269-9101, USA.

[^3]:    ${ }^{7}$ The numbers in brackets preceded by the letter B correspond to those of the bibliography in Clause 14.

[^4]:    ${ }^{8}$ Published by IEEE Transaction on Power Apparatus and Systems, Vol PAS-96, NO 4, July/August 1977.

[^5]:    

[^6]:    ${ }^{9}$ Published by IEEE Transaction on Power Apparatus and Systems, Vol PAS-95, NO 4, July/August 1976. Paper F 76 205-5. Recommended and approved by the IEEE Substations Committee of the IEEE Power Engineering Society for presentation at the IEEE PES Winter Meeting \& Tesla Symposium, New York, N.Y., January 25-30, 1976. Manuscript submitted October 31, 1975; made available for printing November 24, 1975.

[^7]:    ${ }^{10}$ The numbers in brackets correspond to those of the references at the end of this annex.

[^8]:    --......,.,...........,.....................

[^9]:    ${ }^{11}$ Published by Proceedings IEEE, Volume 110, No. 9, Sept. 1963.

[^10]:    ${ }^{12}$ The numbers in brackets correspond to those of the references at the end of this annex.

